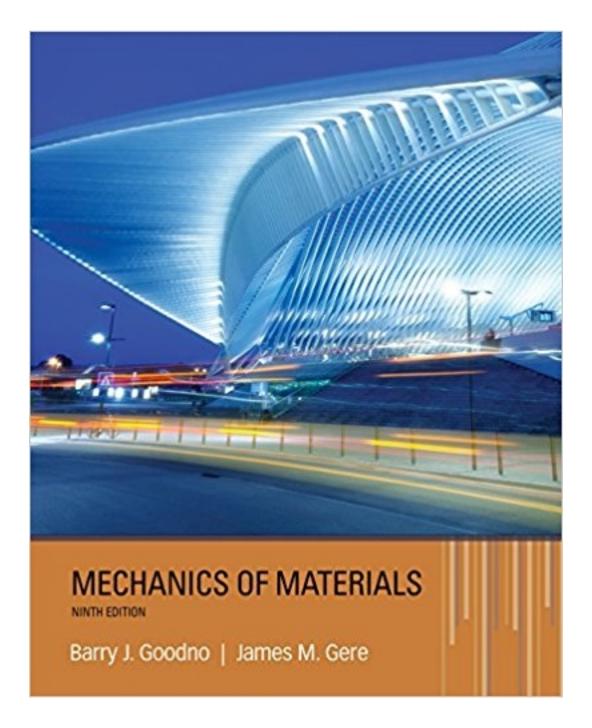
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Solutions

Chapter 2 Solutions

Problem 2.2-1

$$L = 10 \text{ft} \qquad q_0 = 2 \cdot \frac{\text{kip}}{\text{ft}} \qquad k = 4 \cdot \frac{\text{kip}}{\text{in}}$$

$$\Sigma M_{\text{B}} = 0 \qquad \qquad A_{\text{y}} = \frac{1}{L} \cdot \left(\frac{1}{2} \cdot q_0 \cdot L \cdot \frac{L}{3}\right) = 3.333 \cdot \text{kip} \qquad \qquad \delta_{\text{A}} = \frac{A_{\text{y}}}{k} = 0.833 \cdot \text{in} \qquad \qquad \frac{\delta_{\text{A}}}{L} = 6.944 \times 10^{-3}$$

$$\frac{\delta_{A}}{T_{c}} = 6.944 \times 10^{-3}$$

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Problem 2.2-2

$$E = 200 GPa \qquad \qquad d_{_{{\bf f}}} = 25 mm \qquad \qquad q = 5 \, \frac{kN}{m} \qquad \qquad L_{_{{\bf f}}} = 0.75 m \qquad \qquad P = 10 kN \qquad \qquad a = 2.5 m \qquad \qquad b = 0.75 m$$

$$P = 10kN$$
 a

$$= 2.5 \text{m}$$
 $b = 0.75 \text{m}$

$$A_{r} = \frac{\pi}{4} \cdot d_{r}^{2} = 0.761 \cdot in^{2}$$

$$\Sigma M_A = 0$$

$$\Sigma M_{A} = 0$$
 $F_{r} = \frac{1}{a} \left[q \cdot a \cdot \frac{a}{2} + P \cdot (a+b) \right] = 19.25 \cdot kN$

$$\delta_{\text{rod}} = \frac{F_{\text{r}} \cdot L_{\text{r}}}{E \cdot A_{\text{r}}} = 0.1471 \cdot \text{mm}$$

$$\delta_{\text{B}} = \frac{a+b}{a} \cdot \delta_{\text{rod}} = 0.1912 \cdot \text{mm}$$

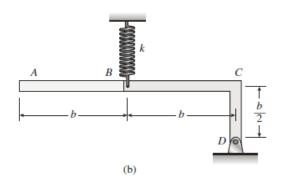
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Problem 2.2-3

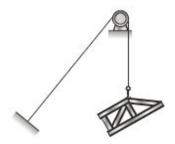
(a) Sum moments about A

$$\Sigma M_A = 0 \qquad \frac{2b}{\frac{5}{2}b}Wb + \frac{\frac{b}{2}}{\frac{5}{2}b}W(2b) = k$$

$$\delta = \frac{\frac{2b}{\frac{5}{2}b}Wb + \frac{\frac{b}{2}}{\frac{5}{2}b}W(2b)}{kb} = \frac{6W}{5k}$$



(b)
$$\Sigma M_D = 0$$
 $kb\delta = \frac{2b}{\frac{5}{2}b}Wb = \frac{4Wb}{5}$ so $\delta = \frac{\frac{2b}{\frac{5}{2}b}Wb}{kb} = \frac{4W}{5k}$



$$A = 304 \text{ mm}^2 \text{ (from Table 2-1)}$$

W = 38 kN

E = 140 GPa

L = 14 m

(b) FACTOR OF SAFETY

$$P_{\text{ULT}} = 406 \text{ kN (from Table 2-1)}$$

$$P_{\text{max}} = 70 \text{ kN}$$

$$n = \frac{P_{ULT}}{P_{\text{max}}} = \frac{406 \text{ kN}}{70 \text{ kN}} = 5.8 \quad \leftarrow$$

(a) STRETCH OF CABLE

$$\delta = \frac{WL}{EA} = \frac{(38 \text{ kN})(14 \text{ m})}{(140 \text{ GPa})(304 \text{ mm}^2)}$$

(a)
$$\frac{\delta_a}{\delta_s} = \frac{\frac{PL}{E_a A}}{\left(\frac{PL}{E_s A}\right)} \rightarrow \frac{E_s}{E_a}$$

 $E_s = 30,000 \text{ ksi}$ $E_a = 11,000 \text{ ksi}$

$$\frac{E_s}{E_a} = 2.727 \qquad \frac{30}{11} = 2.727$$

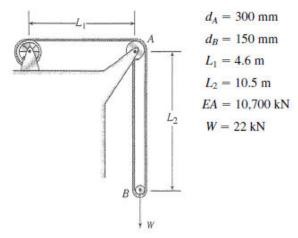
(b)
$$\delta_a = \delta_s$$
 so $\frac{PL}{E_a A_a} = \frac{PL}{E_s A_s}$ so $\frac{A_a}{A_s} = \frac{E_s}{E_a}$ and $\frac{d_a}{d_s} = \sqrt{\frac{E_s}{E_a}} = 1.651$

(c) Same diameter, same load, find ratio of length of aluminum to steel wire if elongation of aluminum is 1.5 times that of steel wire

$$\frac{\delta_a}{\delta_s} = \frac{\frac{PL_a}{E_a A}}{\left(\frac{PL_s}{E_s A}\right)} \qquad \frac{\frac{PL_a}{E_a A}}{\left(\frac{PL_s}{E_s A}\right)} = 1.5 \qquad \frac{L_a}{L_s} = 1.5 \frac{E_a}{E_s} = 0.55$$

(d) Same diameter, same length, same load—but wire 1 elongation 1.7 times the steel wire > what is wire 1 material?

$$\frac{\delta_{1}}{\delta_{s}} = \frac{\frac{PL}{E_{1}A}}{\left(\frac{PL}{E_{r}A}\right)} \qquad \frac{\frac{PL}{E_{1}A}}{\left(\frac{PL}{E_{r}A}\right)} = 1.7 \qquad E_{1} = \frac{E_{s}}{1.7} = 17,647 \text{ ksi} \qquad \boxed{\text{cast iron or copper alloy (see App. I)}}$$



TENSILE FORCE IN CABLE

$$T = \frac{W}{2} = 11 \text{ kN}$$

LENGTH OF CABLE
$$L = L_1 + 2L_2 + \frac{1}{4} (\pi d_A) + \frac{1}{2} (\pi d_B)$$

$$= 4,600 \text{ mm} + 21,000 \text{ mm} + 236 \text{ mm} + 236 \text{ mm}$$

ELONGATION OF CABLE

= 26,072 mm

$$\delta = \frac{TL}{EA} = \frac{(11 \text{ kN})(26,072 \text{ mm})}{(10,700 \text{ kN})} = 26.8 \text{ mm}$$

LOWERING OF THE CAGE

h = distance the cage moves downward

$$h = \frac{1}{2}\delta = 13.4 \text{ mm} \leftarrow$$

$$\begin{aligned} &d_{o} = 15 in & d_{i} = 14.4 in & E = 29000 ksi & P = 5 kip \\ &L_{DC} = \sqrt{\left(3 \, ft\right)^{2} + \left(4 \, ft\right)^{2}} = 5 \cdot ft & A_{DC} = \frac{\pi}{4} \cdot \left(d_{o}^{2} - d_{i}^{2}\right) = 13.854 \cdot in^{2} \end{aligned}$$

Find force in DC - use FBD of ACB

$$\Sigma M_{A} = 0$$
 $\frac{3}{5}F_{DC}$ (4ft) = P·(9ft) so $F_{DC} = \frac{5}{3} \cdot P \cdot \left(\frac{9}{4}\right) = 18.75 \cdot \text{kip}$ compression

Change in length of strut

$$\Delta_{DC} = \frac{F_{DC} \cdot L_{DC}}{E \cdot A_{DC}} = 2.8 \times 10^{-3} \cdot in \qquad \text{shortening}$$

Vertical displacement at C (see Example 2-7) and at B

$$\delta_{C} = \frac{\Delta_{DC}}{\sin(\text{ACD})} \qquad \delta_{C} = \frac{\Delta_{DC}}{\frac{3}{5}} = 4.667 \times 10^{-3} \cdot \text{in} \qquad \delta_{B} = \frac{9}{4} \cdot \delta_{C} = 1.05 \times 10^{-2} \cdot \text{in} \quad \text{downward}$$

$$L_{DD} = 350 m$$

$$L_{BD} = 350 \text{mm}$$
 $L_{CE} = 450 \text{mm}$ $A = 720 \text{mm}^2$ $E = 200 \text{GPa}$ $P = 20 \text{kN}$

$$A = 720 \text{mm}$$

$$E = 200GPa$$

$$P = 20kN$$

Statics - find axial forces in BD and CE - remove pins at B and E, use FBD of beam ABC - assume beam is rigid

$$\Sigma M_B = 0$$
 $CE = \frac{1}{350 \text{mm}} \cdot [P \cdot (600 \text{mm})] = 34.286 \cdot \text{kN}$ CE is in tension; force CE acts downward on ABC

$$\Sigma F_{-} = 0$$

$$\Sigma F_{V} = 0$$
 BD = P + CE = 54.286·kN

BD is in compression; force BD acts upward on ABC

Use force-displacement relation to find change in lengths of CE and BD and vertical displacements at B and C

$$\delta_{BD} = \frac{BD \cdot L_{BD}}{F. \Delta} = 0.13194 \cdot mm$$
 shortening

$$\delta_{CE} = \frac{\text{CE-L}_{CE}}{\text{E-A}} = 0.10714 \cdot \text{mm elongation}$$

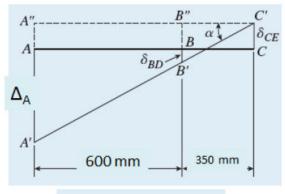
Use geometry to find downward displacement at A

$$\alpha = \operatorname{atan}\left(\frac{\left|\delta_{BD}\right| + \delta_{CE}}{350 \text{mm}}\right) = 0.03914 \cdot \text{deg}$$

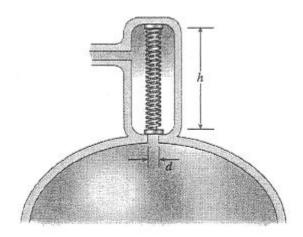
$$\Delta_A = 950 \text{mm} \cdot \tan(\alpha) - \delta_{CE} = 0.542 \cdot \text{mm}$$
 downward

or similar triangles
$$\frac{\Delta_A + \delta_{CE}}{600 + 350} = \frac{\left|\delta_{BD}\right| + \delta_{CE}}{350}$$

$$\Delta_{A} = \left(\left|\delta_{BD}\right| + \delta_{CE}\right) \cdot \left(\frac{950}{350}\right) - \delta_{CE} = 0.542 \cdot mm$$



$$\frac{\delta_A + \delta_{CE}}{600 + 350} = \frac{\delta_{BD} + \delta_{CE}}{350}$$



h = height of valve (compressed length of the spring)

d = diameter of discharge hole

p = pressure in tank

 p_{max} = pressure when valve opens

L =natural length of spring (L > h)

k = stiffness of spring

FORCE IN COMPRESSED SPRING

$$F = k(L - h)$$
 (From Eq. 2-1a)

PRESSURE FORCE ON SPRING

$$P = p_{\text{max}} \left(\frac{\pi d^2}{4} \right)$$

Equate forces and solve for h:

$$F = P \quad k(L - h) = \frac{\pi p_{\text{max}} d^2}{4}$$

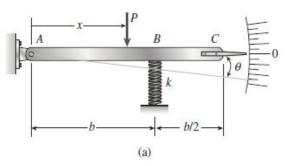
$$h = L - \frac{\pi p_{\text{max}} d^2}{4 k} \quad \leftarrow$$

k = 950 N/m b = 165 mm P = 11 N $\theta = 2.5^{\circ}$ $\theta_{\text{max}} = 2^{\circ}$ NUMERICAL DATA $W_n = 3N \qquad W_s = 2.75N$

(a) If the load P = 11 N, at what distance x should the load be placed so that the pointer will read $\theta = 2.5^{\circ}$ on the scale (see Fig. a)?

Sum moments about A, then solve for x:

$$x = \frac{k\theta b^2}{P} = 102.6 \text{ mm}$$
 $x = 102.6 \text{ mm}$

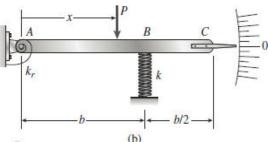


(b) Repeat (a) if a rotational spring $k_r = kb^2$ is added at A (see

$$k_c = k b^2 = 25864 \text{ N} \cdot \text{mm}$$

Sum moments about A, then solve for x:

$$x = \frac{k\theta b^2 + k_r \theta}{P} = 205 \text{ mm}$$
 $\frac{x}{b} = 1.244 \text{ } [x = 205 \text{ mm}]$



(c) Now if x = 7b/8, what is P_{max} (N) if θ cannot exceed 2° ? $x = \frac{7}{8}b = 144.375$ mm

$$P_{\text{max}} = \frac{k\theta_{\text{max}}b^2 + k_r\theta_{\text{max}}}{\frac{7}{8}b} = 12.51 \text{ N}$$
 $P_{\text{max}} = 12.51 \text{ N}$

$$P_{\text{max}} = 12.51\,\text{N}$$

(d) Now, if the weight of the pointer ABC is known to be $W_p = 3$ N and the weight of the spring is $W_s = 2.75$ N, what initial angular position (i.e., θ in degrees) of the pointer will result in a zero reading on the angular scale once the pointer is released from rest? Assume $P = k_r = 0$.

Deflection at spring due to W_p :

Deflection at B due to self weight of spring:

$$\delta_{Bp} = \frac{W_p \left(\frac{3}{4}b\right)}{kb} = 2.368 \text{ mm} \qquad \delta_{Bk} = \frac{W_s}{2k} = 1.447 \text{ mm}$$

$$\delta_B = \delta_{Bp} + \delta_{Bk} = 3.816 \text{ mm} \qquad \theta_{\text{init}} = \frac{\delta_B}{b} = 1.325^{\circ}$$

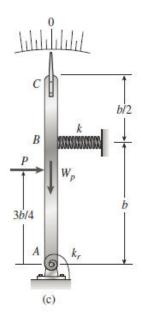
$$\text{OR} \quad \theta_{\text{init}} = \arctan\left(\frac{\delta_B}{b}\right) = 1.325^{\circ} \quad \boxed{\theta_{\text{init}} = 1.325^{\circ}}$$

(e) If the pointer is rotated to a vertical position (figure part c), find the required load P, applied at mid-height of the pointer that will result in a pointer reading of $\theta = 2.5^{\circ}$ on the scale. Consider the weight of the pointer, W_p , in your analysis.

$$k = 950 \text{ N/m}$$
 $b = 165 \text{ mm}$ $W_p = 3 \text{ N}$
 $k_r = kb^2 = 25.864 \text{ N} \cdot \text{m}$ $\theta = 2.5^{\circ}$

Sum moments about A to get P:

$$P = \frac{\theta}{\left(\frac{3b}{4}\right)} \left[k_r + k \left(\frac{5}{4}b^2\right) - W_p \left(\frac{3b}{4}\right) \right] = 20.388 \text{ N} \qquad \boxed{P = 20.4 \text{ N}}$$



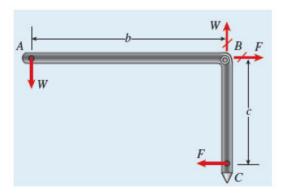
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Problem 2.2-11

$$b = 10in$$
 $c = 7in$ $k = 5\frac{1bf}{in}$ $p = \frac{1}{16}in$ $n = 12$

Use FBD of ABC (pin forces $B_x = F$ and $B_y = W$ at B; see fig.); sum moments about B s.t. Wb = Fc, F = force in spring

$$\Sigma M_B = 0$$
 $W = F \cdot \frac{c}{b}$



$$F = k \cdot (n \cdot p) = 3.75 \cdot 1bf$$

Force in spring is
$$F = k \cdot (n \cdot p) = 3.75 \cdot 1bf \qquad \text{so} \qquad W = F \cdot \frac{c}{b} = 2.625 \cdot 1bf$$

$$b = 30 cm$$
 $c = 20 cm$ $k = 3650 \frac{N}{m}$ $p = 1.5 mm$ $W = 65 N$

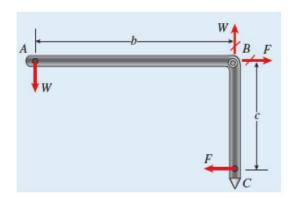
Force in parallel springs is $F = 2 \cdot k \cdot (n \cdot p)$

Sum moments about B (see FBD) to find F in terms of weight W

$$Wb = Fc$$
 so $F = W \cdot \frac{b}{c}$

Substitute expression for F and solve for n

$$n = \frac{W \cdot \frac{b}{c}}{2 \cdot k \cdot p} = 8.904$$



(a) Derive a formula for the displacement δ₄ at point 4 when the load P is applied at joint 3 and moment PL is applied at joint 1, as shown.

Cut horizontally through both springs to create upper and lower FBD's. Sum moments about joint 1 for upper FBD and also sum moments about joint 6 for lower FBD to get two equations of equilibrium; assume both springs are in tension.

Note that

$$\delta_2 = \frac{2}{3} \, \delta_3$$
 and $\delta_5 = \frac{3}{4} \, \delta_4$

Force in left spring: $k\left(\delta_4 - \frac{2}{3}\delta_3\right)$

Force in right spring:
$$2k\left(\frac{3}{4}\delta_4 - \delta_3\right)$$

Summing moments about joint 1 (upper FBD) and about joint 6 (lower FBD) then dividing through by k gives

$$\begin{pmatrix} \frac{-22}{9} & \frac{13}{6} \\ \frac{-26}{9} & \frac{17}{6} \end{pmatrix} \begin{pmatrix} \delta_3 \\ \delta_4 \end{pmatrix} = \begin{pmatrix} \frac{-2P}{k} \\ 0 \end{pmatrix} \quad \begin{pmatrix} \delta_3 \\ \delta_4 \end{pmatrix} = \begin{pmatrix} \frac{-22}{9} & \frac{13}{6} \\ \frac{-26}{9} & \frac{17}{6} \end{pmatrix}^{-1} \begin{pmatrix} \frac{-2P}{k} \\ 0 \end{pmatrix} = \begin{pmatrix} \frac{17P}{2k} \\ \frac{26P}{3k} \end{pmatrix} \quad \frac{17}{2} = 8.5 \\ \frac{26P}{3k} = 8.667 \quad \frac{8}{3} = 8.667 \quad \frac{8$$

^ deltas are positive downward

(b) Repeat part (a) if a rotational spring $k_r = kL^2$ is now added at joint 6. What is the ratio of the deflection $\delta 4$ in part (a) to that in (b)?

Upper FBD—sum moments about joint 1:

$$k\left(\delta_4 - \frac{2}{3}\delta_3\right)\frac{2L}{3} + 2k\left(\frac{3}{4}\delta_4 - \delta_3\right)L = -2PL \quad \text{OR} \quad \left(\frac{22Lk}{9}\right)\delta_3 + \frac{13Lk}{6}\delta_4 = -2PL$$

Lower FBD-sum moments about joint 6:

$$k\left(\delta_4 - \frac{2}{3}\delta_3\right) \frac{4L}{3} + 2k\left(\frac{3}{4}\delta_4 - \delta_3\right)L - k_r\theta_6 = 0$$

$$\left[k \left(\delta_4 - \frac{2}{3} \, \delta_3 \right) \frac{4L}{3} + 2k \left(\frac{3}{4} \, \delta_4 - \delta_3 \right) L \right] + (kL^2) \left(\frac{\delta_4}{\frac{4}{3} \, L} \right) = 0 \quad \text{OR} \quad \left(\frac{26Lk}{9} \right) \delta_3 + \frac{43Lk}{12} \, \delta_4 = 0$$

Divide matrix equilibrium equations through by k to get the following displacement equations:

$$\begin{pmatrix} \frac{-22}{9} & \frac{13}{6} \\ \frac{-26}{9} & \frac{43}{6} \end{pmatrix} \begin{pmatrix} \delta_3 \\ \delta_4 \end{pmatrix} = \begin{pmatrix} \frac{-2P}{k} \\ 0 \end{pmatrix} \quad \begin{pmatrix} \delta_3 \\ \delta_4 \end{pmatrix} = \begin{pmatrix} \frac{-22}{9} & \frac{13}{6} \\ \frac{-26}{9} & \frac{43}{12} \end{pmatrix}^{-1} \begin{pmatrix} \frac{-2P}{k} \\ 0 \end{pmatrix} = \begin{pmatrix} \frac{43P}{15k} \\ \frac{104P}{45k} \end{pmatrix} \quad \frac{43}{15} = 2.867$$

^ deltas are positive downward

Ratio of the deflection
$$\delta_4$$
 in part (a) to that in (b):
$$\frac{\frac{26}{3}}{\frac{104}{45}} = \frac{15}{4} \quad \boxed{\text{Ratio} = \frac{15}{4} = 3.75}$$

NUMERICAL DATA

$$A = 3900 \text{ mm}^2$$
 $E = 200 \text{ GPa}$

$$P = 475 \text{ kN}$$
 $L = 3000 \text{ mm}$

 $\delta_{Bmax} = 1.5 \text{ mm}$

(a) Find horizontal displacement of joint ${\it B}$ Statics
To find support reactions and then member forces:

$$\sum M_A = 0$$
 $B_y = \frac{1}{L} \left(2P \frac{L}{2} \right)$ $B_y = P$

$$\sum F_H = 0$$
 $A_x = -P$

$$\sum F_V = 0$$
 $A_y = P - B_y$ $A_y = 0$

METHOD OF JOINTS: $AC_V = A_Y$ $AC_V = 0$ Force in AC = 0

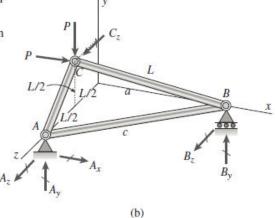
$$AB = A_X$$

Force in AB is P (tension) so elongation of AB is the horizontal displacement of joint B.

$$\delta_B = \frac{F_{AB}L}{EA}$$
 $\delta_B = \frac{PL}{EA}$ $\delta_B = 1.82692 \text{ mm}$ $\delta_B = 1.827 \text{ mm}$

- (b) Find P_{max} if displacement of joint $B = \delta_{B_{\text{max}}} = 1.5 \text{ mm}$ $P_{\text{max}} = \frac{EA}{L} \delta_{B_{\text{max}}}$ $P_{\text{max}} = 390 \text{ kN}$
- (c) Repeat parts (a) and (b) if the plane truss is replaced by a space truss (see figure part b).

Find missing dimensions a and c: P = 475 kN L = 3 m



(a)

$$a = \sqrt{L^2 - 2\left(\frac{L}{2}\right)^2} = 2.12132 \text{ m} \qquad \frac{a}{L} = 0.707 \qquad a = \frac{L}{\sqrt{2}} = 2.12132 \text{ m}$$

$$c = \sqrt{L^2 + a^2} = 3.67423 \text{ m} \qquad c = \sqrt{L^2 + \left(\frac{L}{\sqrt{2}}\right)^2} = 3.67423 \text{ m} \qquad c = L\sqrt{\frac{3}{2}} = 3.67423 \text{ m}$$

(1) SUM MOMENTS ABOUT A LINE THRU A WHICH IS PARALLEL TO THE Y-AXIS

$$B_z = -P\frac{L}{a} = -671.751 \text{ kN}$$

(2) SUM MOMENTS ABOUT THE Z-AXIS

$$B_y = \frac{P\left(\frac{L}{2}\right)}{a} = 335.876 \text{ kN}$$
 SO $A_y = P - B_y = 139.124 \text{ kN}$

(3) SUM MOMENTS ABOUT THE X-AXIS

$$C_z = \frac{A_y L - P\frac{L}{2}}{\frac{L}{2}} = -196.751 \,\text{kN}$$

- (4) Sum forces in the x- and z-directions $A_x = -P = -475 \text{ kN}$ $A_z = -C_z B_z = 868.503 \text{ kN}$
- (5) Use method of joints to find member forces

Sum forces in x-direction at joint A:
$$\frac{a}{c}F_{AB} + A_x = 0$$
 $F_{AB} = \frac{-c}{a}A_x = 823 \text{ kN}$

Sum forces in y-direction at joint A:
$$\frac{\frac{L}{2}}{\sqrt{2}}F_{AC} + A_y = 0 \qquad F_{AC} = \sqrt{2}(-A_y) = -196.8 \text{ kN}$$

Sum forces in y-direction at joint B:
$$\frac{L}{2}F_{BC} + B_y = 0$$
 $F_{BC} = -2B_y = -672 \text{ kN}$

(6) FIND DISPLACEMENT ALONG x-AXIS AT JOINT B Find change in length of member AB then find its projection along x axis:

$$\delta_{AB} = \frac{F_{AB}c}{F_{A}} = 3.875 \text{ mm}$$
 $\beta = \arctan\left(\frac{L}{a}\right) = 54.736^{\circ}$ $\delta_{Bx} = \frac{\delta_{AB}}{\cos(\beta)} = 6.713 \text{ mm}$ $\delta_{Bx} = 6.71 \text{ mm}$

(7) Find P_{max} for space truss if δ_{Bx} must be limited to 1.5 mm

Displacements are linearly related to the loads for this linear elastic small displacement problem, so reduce load variable P from 475 kN to

$$\frac{1.5}{6.71254} \text{ 475} = 106.145 \text{ kN} \qquad P_{\text{max}} = 106.1 \text{ kN}$$

Repeat space truss analysis using vector operations a = 2.121 m L = 3 m P = 475 kN

Position and unit vectors:

$$r_{AB} = \begin{pmatrix} a \\ 0 \\ -L \end{pmatrix}$$
 $e_{AB} = \frac{r_{AB}}{|r_{AB}|} = \begin{pmatrix} 0.577 \\ 0 \\ -0.816 \end{pmatrix}$ $r_{AC} = \begin{pmatrix} 0 \\ \frac{L}{2} \\ \frac{-L}{2} \end{pmatrix}$ $e_{AC} = \frac{r_{AC}}{|r_{AC}|} = \begin{pmatrix} 0 \\ 0.707 \\ -0.707 \end{pmatrix}$

FIND MOMENT AT A:

$$M_A = r_{AB} \times R_B + r_{AC} \times R_C$$

$$M_A = r_{AB} \times \begin{pmatrix} 0 \\ RB_y \\ RB_z \end{pmatrix} + r_{AC} \times \begin{pmatrix} 2.P \\ -P \\ RC_z \end{pmatrix} = \begin{pmatrix} 3.0 \text{ m } RB_y + 1.5 \text{ m } RC_z - 712.5 \text{ kN} \cdot \text{m} \\ -2.1213 \text{ m } RB_z - 1425.0 \text{ kN} \cdot \text{m} \\ 2.1213 \text{ m } RB_y - 1425.0 \text{ kN} \cdot \text{m} \end{pmatrix}$$

FIND MOMENTS ABOUT LINES OR AXES:

$$\begin{split} M_A e_{AB} &= -1.732 \text{ m } RB_y + 1.7321 \text{ m } RB_y + 0.86603 \text{ m } RC_z + 752.15 \text{ kN} \cdot \text{m} \\ RC_z &= \frac{-244.12}{0.72169} = -338.262 \qquad C_z = -196.751 \text{ kN} \\ M_A e_{AC} &= -1.5 \text{ m } RB_y + -1.5 \text{ m } RB_z \quad \text{So} \quad RB_y = -RB_z \\ M_A \begin{pmatrix} 0 \\ 1 \\ 0 \end{pmatrix} &= -2.1213 \text{ m } RB_z + -1425.0 \text{ kN} \cdot \text{m} \quad \text{So} \quad RB_z = \frac{462.5}{-1.7678} = -261.625 \quad B_z = -671.75 \text{ kN} \\ M_A \begin{pmatrix} 0 \\ 1 \\ 0 \end{pmatrix} &= 2.1213 \text{ m } RB_y + -1425.0 \text{ kN} \cdot \text{m} \quad \text{So} \quad RB_y = -RB_z = 261.625 \quad B_y = -335.876 \text{ kN} \\ \sum F_y &= 0 \qquad A_y = P - B_y = 139.124 \text{ kN} \end{split}$$

Reactions obtained using vector operations agree with those based on scalar operations.

$$d = \frac{1}{10}$$
 in. $L = 12(12)$ in. $E = 10,600 \times (10^3)$ psi $\delta_a = \frac{1}{8}$ in. $\sigma_a = 10 \times (10^3)$ psi $A = \frac{\pi d^2}{4}$ $A = 7.854 \times 10^{-3}$ in.² $EA = 8.325 \times 10^4$ lb



Maximum load based on elongation:

$$P_{\text{max}1} = \frac{EA}{L} \delta_a \quad P_{\text{max}1} = 72.3 \text{ lb} \quad \leftarrow \text{ controls}$$

Maximum load based on stress:

$$P_{\text{max}2} = \sigma_a A$$
 $P_{\text{max}2} = 78.5 \text{ lb}$

NUMERICAL DATA

W = 25 N $k_1 = 0.300 \text{ N/mm}$ $L_1 = 250 \text{ mm}$

 $k_2 = 0.400 \text{ N/mm}$ $L_2 = 200 \text{ mm}$

L = 350 mm h = 80 mm P = 18 N

(a) Location of load P to bring bar to horizontal position

Use statics to get forces in both springs:

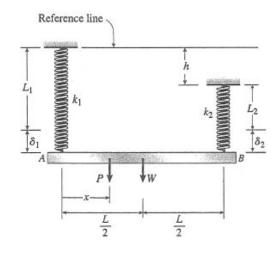
$$\sum M_A = 0 \qquad F_2 = \frac{1}{L} \left(W \frac{L}{2} + Px \right)$$
$$F_2 = \frac{W}{2} + P \frac{x}{L}$$

$$\sum F_V = 0 \qquad F_1 = W + P - F_2$$

$$F_1 = \frac{W}{2} + P\left(1 - \frac{x}{L}\right)$$

Use constraint equation to define horizontal position, then solve for location x:

$$L_1 + \frac{F_1}{k_1} = L_2 + h + \frac{F_2}{k_2}$$



Substitute expressions for F_1 and F_2 above into constraint equilibrium and solve for x:

$$x = \frac{-2L_1 L k_1 k_2 - k_2 WL - 2k_2 P L + 2L_2 L k_1 k_2 + 2 h L k_1 k_2 + k_1 WL}{-2P(k_1 + k_2)}$$

$$x = 134.7 \text{ mm} \quad \leftarrow$$

(b) Next remove P and find new value of spring constant k_1 so that bar is horizontal under weight W

Now,
$$F_1 = \frac{W}{2}$$
 $F_2 = \frac{W}{2}$ since $P = 0$

Same constraint equation as above but now P = 0:

$$L_1 + \frac{\frac{W}{2}}{k_1} - (L_2 + h) - \frac{\left(\frac{W}{2}\right)}{k_2} = 0$$

Solve for k

$$k_1 = \frac{-Wk_2}{[2k_2[L_1 - (L_2 + h)]] - W}$$

$$k_1 = 0.204 \text{ N/mm} \leftarrow$$

PART (c)—CONTINUED (from page below)

STATICS

$$\sum M_{k_1} = 0 \qquad F_2 = \frac{w\left(\frac{L}{2} - b\right)}{L - b}$$

$$\sum F_V = 0$$

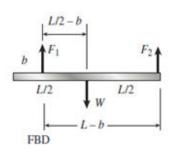
$$F_1 = W - F_2$$

$$F_1 = W - \frac{w\left(\frac{L}{2} - b\right)}{L - b}$$

$$F_1 = \frac{WL}{2(L - b)}$$

Part (c) continued in right column below

(c) Use $k_1 = 0.300 \text{ N/mm}$ but relocate SPRING k_1 (x = b) so that bar ends up IN HORIZONTAL POSITION UNDER WEIGHT W



$$b = \frac{2L_1k_1k_2L + WLk_2 - 2L_2k_1k_2L - 2hk_1k_2L - Wk_1L}{(2L_1k_1k_2) - 2L_2k_1k_2 - 2hk_1k_2 - 2Wk_1} \qquad b = 74.1 \text{ mm} \quad \leftarrow$$

Part (c) continued on page above

(d) Replace spring k_1 with springs in series: $k_1 = 0.3 \text{ N/mm}, L_1/2, \text{ AND } k_3, L_1/2. \text{ FIND } k_3$ SO THAT BAR HANGS IN HORIZONTAL POSITION

Statics
$$F_1 = \frac{W}{2}$$
 $F_2 = \frac{W}{2}$

$$k_3 = \frac{Wk_1k_2}{-2L_1k_1k_2 - Wk_2 + 2L_2k_1k_2 + 2hk_1k_2 + Wk_1}$$

NOTE-equivalent spring constant for series springs:

$$k_e = \frac{k_1 k_3}{k_1 + k_3}$$

Constraint equation-substitute above expressions for F_1 and F_2 and solve for b:

$$L_1 + \frac{F_1}{k_1} - (L_2 + h) - \frac{F_2}{k_2} = 0$$

Use the following data:

$$k_1 = 0.300 \text{ N/mm}$$
 $k_2 = 0.4 \text{ N/mm}$ $L_1 = 250 \text{ mm}$ $L_2 = 200 \text{ mm}$ $L = 350 \text{ mm}$

Part (d) continued from left column

New constraint equation; solve for k_3 :

$$\begin{split} L_1 + \frac{F_1}{k_1} + \frac{F_1}{k_3} - (L_2 + h) - \frac{F_2}{k_2} &= 0 \\ L_1 + \frac{W/2}{k_1} + \frac{W/2}{k_3} - (L_2 + h) - \frac{W/2}{k_2} &= 0 \end{split}$$

$$k_3 = 0.638 \text{ N/mm} \leftarrow$$

$$k_e = 0.204 \text{ N/mm}$$
 \leftarrow checks—same as (b) above

The figure shows a section cut through the pipe, cap, and rod.

NUMERICAL DATA

$$E_c = 12000 \text{ ksi}$$
 $E_b = 14,000 \text{ ksi}$

$$W = 2 \text{ k}$$
 $d_c = 6 \text{ in.}$ $d_r = \frac{1}{2} \text{ in.}$

$$\sigma_a = 5 \text{ ksi } \delta_a = 0.02 \text{ in.}$$

Unit weights (see Table I-1): $\gamma_s = 2.836 \times 10^{-4} \text{ k/in.}^3$

$$\gamma_b = 3.009 \times 10^{-4} \text{ k/in.}^3$$

$$L_c = 48 \text{ in.}$$
 $L_r = 42 \text{ in.}$

$$t_{\rm v}=1$$
 in.

(a) MINIMUM REQUIRED WALL THICKNESS OF CAST IRON PIPE, t_{cmin}

First check allowable stress then allowable shortening:

$$W_{\rm cap} = \gamma_s \left(\frac{\pi}{4} d_c^2 t_s \right)$$

$$W_{\rm cap} = 8.018 \times 10^{-3} \,\mathrm{k}$$

$$W_{\rm rod} = \gamma_b \left(\frac{\pi}{4} d_r^2 L_r \right)$$

$$W_{\rm rod} = 2.482 \times 10^{-3} \,\mathrm{k}$$

$$W_t = W + W_{\text{cap}} + W_{\text{rod}} \qquad W_t = 2.01 \text{ k}$$

$$A_{\min} = \frac{W_t}{\sigma_c}$$
 $A_{\min} = 0.402 \text{ in.}^2$

$$A_{\text{pipe}} = \frac{\pi}{4} [d_c^2 - (d_c - 2t_c)^2]$$

$$A_{\text{pipe}} = \pi t_c (d_c - t_c)$$

$$t_c(d_c - t_c) = \frac{W_t}{\pi \sigma_c}$$

Let
$$\alpha = \frac{W_t}{\pi \sigma_a}$$
 $\alpha = 0.128$:

$$t_c^2 - d_c t_c + \alpha = 0$$

$$t_c = \frac{d_c - \sqrt{d_c^2 - 4\alpha}}{2}$$
 $t_c = 0.021 \text{ in.}$

^ minimum based on σ_a

Now check allowable shortening requirement:

$$\delta_{\text{pipe}} = \frac{W_t L_c}{E_c A_{\min}} \quad A_{\min} = \frac{W_t L_c}{E_c \delta_a}$$

 $A_{\min} = 0.447 \text{ in.}^2 < \text{larger than value based on}$

$$\sigma_a$$
 above

$$\pi t_c (d_c - t_c) = \frac{W_t L_c}{E_c \delta_a}$$

$$t_c^2 - d_c t_c + \beta = 0$$
 $\beta = \frac{W_t L_c}{\pi E_c \delta_c}$

$$\beta = 0.142$$

$$t_c = \frac{d_c - \sqrt{d_c^2 - 4\beta}}{2}$$

 $t_c = 0.021$ in. \leftarrow minimum based on δ_a and σ_a

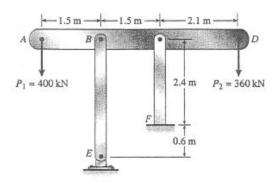
controls

(b) Elongation of rod due to self weight and also weight W

$$\delta_r = \frac{\left(W + \frac{W_{\rm rod}}{2}\right)L_r}{E_b\left(\frac{\pi}{4}d_r^2\right)} \quad \delta_r = 0.031 \text{ in.} \quad \leftarrow$$

(c) MINIMUM CLEARANCE h

$$h_{\min} = \delta_a + \delta_r$$
 $h_{\min} = 0.051$ in. \leftarrow



$$A_{BE} = 11,100 \text{ mm}^2$$

 $A_{CF} = 9,280 \text{ mm}^2$

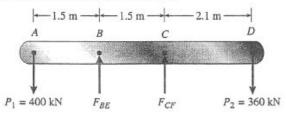
E = 200 GPa

 $L_{BE} = 3.0 \text{ m}$

 $L_{CF} = 2.4 \text{ m}$

 $P_1 = 400 \text{ kN}; P_2 = 360 \text{ kN}$

FREE-BODY DIAGRAM OF BAR ABCD



$$\Sigma M_B = 0$$

$$(400 \text{ kN})(1.5 \text{ m}) + F_{CF}(1.5 \text{ m}) - (360 \text{ kN})(3.6 \text{ m}) = 0$$

$$F_{CF} = 464 \text{ kN}$$

$$\sum M_C = 0$$

$$(400 \text{ kN})(3.0 \text{ m}) - F_{BE}(1.5 \text{ m}) - (360 \text{ kN})(2.1 \text{ m}) = 0$$

$$F_{BE} = 296 \text{ kN}$$

SHORTENING OF BAR BE

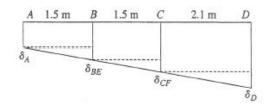
$$\delta_{BE} = \frac{F_{BE}L_{BE}}{EA_{BE}} = \frac{(296 \text{ kN})(3.0 \text{ m})}{(200 \text{ GPa})(11,100 \text{ mm}^2)}$$

$$= 0.400 \text{ mm}$$

SHORTENING OF BAR CF

$$\delta_{CF} = \frac{F_{CF}L_{CF}}{EA_{CF}} = \frac{(464 \text{ kN})(2.4 \text{ m})}{(200 \text{ GPa})(9,280 \text{ mm}^2)}$$
$$= 0.600 \text{ mm}$$

DISPLACEMENT DIAGRAM



$$\delta_{BE} - \delta_A = \delta_{CF} - \delta_{BE} \text{ or } \delta_A = 2\delta_{BE} - \delta_{CF}$$

$$\delta_A = 2(0.400 \text{ mm}) - 0.600 \text{ m}$$

$$= 0.200 \text{ mm} \leftarrow (\text{Downward})$$

$$\begin{split} \delta_D - \delta_{CF} &= \frac{2.1}{1.5} (\delta_{CF} - \delta_{BE}) \\ \text{or} \quad \delta_D &= \frac{12}{5} \delta_{CF} - \frac{7}{5} \delta_{BE} \\ &= \frac{12}{5} (0.600 \text{ mm}) - \frac{7}{5} (0.400 \text{ mm}) \\ &= 0.880 \text{ mm} \quad \longleftarrow \end{split}$$

(Downward)

(a) DISPLACEMENT δ_D

Use
$$FBD$$
 of beam BCD $\sum M_B = 0$ $R_C = \frac{1}{L} \left[\left(2 \frac{P}{L} \right) \left(\frac{3}{4} L \right) \left(\frac{3}{8} L \right) + \frac{P}{4} \left(L + \frac{3}{4} L \right) \right] = P$ < compression force in column CF

$$\sum F_V = 0$$
 $R_B = \left(2\frac{P}{L}\right)\left(\frac{3}{4}L\right) + \frac{P}{4} - R_C = \frac{3P}{4}$ < compression force in column BA

Downward displacements at B and C:
$$\delta_B = R_B f_1 = \frac{3Pf_1}{4}$$
 $\delta_C = R_C f_2 = Pf_2$

Geometry:
$$\delta_D = \delta_B + (\delta_C - \delta_B) \left(\frac{L + \frac{3}{4}L}{L} \right) = \frac{7Pf_2}{4} - \frac{9Pf_1}{16}$$
 $\delta_D = \frac{7Pf_2}{4} - \frac{9Pf_1}{16} = \frac{P}{16}(28f_2 - 9f_1)$

(b) Displacement to horizontal position, so
$$\delta_C = \delta_B$$
 and $\frac{3Pf_1}{4} = Pf_2$ or $\frac{f_1}{f_2} = \frac{4}{3}$

$$\frac{\frac{L_1}{EA_1}}{\frac{L_2}{EA_2}} = \frac{4}{3} \quad \text{or} \quad \frac{L_1}{L_2} = \frac{4}{3} \left(\frac{A_1}{A_2}\right) \quad \frac{L_1}{L_2} = \frac{4}{3} \left(\frac{\frac{\pi}{4} d_1^2}{\frac{\pi}{4} d_2^2}\right) = \frac{4d_1^2}{3d_2^2} \quad \frac{L_1}{L_2} = \frac{4}{3} \left(\frac{d_1}{d_2}\right)^2 \quad \text{with} \quad \frac{d_1}{d_2} = \frac{9}{8}$$

$$\frac{L_1}{L_2} = \frac{4}{3} \left(\frac{9}{8}\right)^2 = \frac{27}{16} \qquad \frac{L_1}{L_2} = \frac{27}{16}$$

(c) If $L_1=2$ L_2 , find the d_1/d_2 ratio so that beam BCD displaces downward to a horizontal position

$$\frac{L_1}{L_2} = 2$$
 and $\delta_C = \delta_B$ from part (b). $\left(\frac{d_1}{d_2}\right)^2 = \frac{3}{4} \left(\frac{L_1}{L_2}\right)$ so $\frac{d_1}{d_2} = \sqrt{\frac{3}{4}(2)} = 1.225$

(d) If $d_1 = (9/8)$ d_2 and $L_1/L_2 = 1.5$, at what horizontal distance x from B should load P/4 at D be placed?

Given
$$\frac{d_1}{d_2} = \frac{9}{8}$$
 and $\frac{L_1}{L_2} = 1.5$ or $\frac{f_1}{f_2} = \frac{L_1}{L_2} \left(\frac{A_2}{A_1}\right)$ $\frac{f_1}{f_2} = \frac{L_1}{L_2} \left(\frac{d_2}{d_1}\right)^2 = \frac{3}{2} \left(\frac{8}{9}\right)^2 = \frac{32}{27}$

Recompute column forces R_R and R_C but now with load P/4 positioned at distance x from B.

Use FBD of beam BCD:
$$\sum M_B = 0$$
 $R_C = \frac{1}{L} \left[\left(2 \frac{P}{L} \right) \left(\frac{3}{4} L \right) \left(\frac{3}{8} L \right) + \frac{P}{4} (x) \right] = \frac{9LP}{16} + \frac{Px}{4}$

$$\Sigma F_V = 0$$
 $R_B = \left(2\frac{P}{L}\right)\left(\frac{3}{4}L\right) + \frac{P}{4} - R_C = \frac{7P}{4} - \frac{\frac{9LP}{16} + \frac{Px}{4}}{L}$

Horizontal displaced position under load q and load P/4 so $\delta_C = \delta_B$ or $R_C f_2 = R_B f_1$.

$$\left(\frac{9LP}{16} + \frac{Px}{4}\right) f_2 = \left(\frac{7P}{4} - \frac{9LP}{16} + \frac{Px}{4}\right) f_1 \text{ solve, } x = -\frac{9Lf_2 - 19Lf_1}{4f_1 + 4f_2} = -\frac{L(9f_2 - 19f_1)}{4(f_1 + f_2)}$$

$$x = -\frac{L(9f_2 - 19f_1)}{4(f_1 + f_2)} \text{ or } x = L \left[\frac{19\frac{f_1}{f_2} - 9}{4\left(\frac{f_1}{f_2} + 1\right)}\right]$$

Now substitute f_1/f_2 ratio from above: $x = L \left| \frac{19 \frac{32}{27} - 9}{4 \left(\frac{32}{27} + 1 \right)} \right| = \frac{365 L}{236} = 1.547$

Apply the laws of statics to the structure in its displaced position; also use FBD's of the left and right bars alone (referred to as LHFB and RHFB below).

OVERALL FBD:
$$\sum F_{H} = 0$$
 $H_{A} - k_{1}\delta = 0$ so $H_{A} = k_{1}\delta$ $\sum F_{V} = 0$ $R_{A} + R_{C} = P$ $\sum M_{A} = 0$ $k_{r}(\alpha - \theta) - P\frac{L_{2}}{2} + R_{C}L_{2} = 0$ $R_{C} = \frac{1}{L_{2}} \left[P\frac{L_{2}}{2} - k_{r}(\alpha - \theta) \right]$ LHFB: $\sum M_{B} = 0$ $H_{A}h + k\frac{\delta}{2}\left(\frac{h}{2}\right) - R_{A}\left(\frac{L_{2}}{2}\right) + k_{r}(\alpha - \theta) = 0$ $R_{A} = \frac{2}{L_{2}} \left[k_{1}\delta h + k\frac{\delta}{2}\left(\frac{h}{2}\right) + k_{r}(\alpha - \theta) \right]$ RHFB: $\sum M_{B} = 0$ $-k\frac{\delta}{2}\left(\frac{h}{2}\right) - k_{1}\delta h + R_{C}\frac{L_{2}}{2} = 0$ $R_{C} = \frac{2}{L_{2}} \left[k\frac{\delta}{2}\left(\frac{h}{2}\right) + k_{1}\delta h \right]$

Equate the two expressions for R_C then substitute expressions for L_2 , k_r , k_1 , h and δ

$$\begin{split} &\frac{1}{L_{2}} \left[P \frac{L_{2}}{2} - k_{r}(\alpha - \theta) \right] = \frac{2}{L_{2}} \left[k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_{1} \delta h \right] \quad \text{OR} \\ &\frac{1}{L_{2}} \left[P \frac{L_{2}}{2} - k_{r}(\alpha - \theta) \right] - \left[\frac{2}{L_{2}} \left[k \frac{2b \left(\cos(\theta) - \cos(\alpha) \right)}{2} \frac{b \sin(\theta)}{2} + k_{1} [2b \left(\cos(\theta) - \cos(\alpha) \right)] \left(b \sin(\theta) \right) \right] \right] = 0 \end{split}$$

(a) Substitute numerical values, then solve numerically for angle θ and distance increase δ

$$\begin{split} b &= 200 \text{ mm} \quad k = 3.2 \text{ kN/m} \quad \alpha = 45^{\circ} \quad P = 50 \text{ N} \quad k_1 = 0 \quad k_r = 0 \\ L_2 &= 2b \cos(\theta) \quad L_1 = 2b \cos(\alpha) \quad \delta = L_2 - L_1 \quad \delta = 2b (\cos(\theta) - \cos(\alpha)) \quad h = b \sin(\theta) \\ \frac{1}{L_2} \bigg[P \frac{L_2}{2} - k_r (\alpha - \theta) \bigg] - \bigg[\frac{1}{L_2} \bigg[k \frac{2b (\cos(\theta) - \cos(\alpha))}{2} \frac{b \sin(\theta)}{2} + k_1 [2b (\cos(\theta) - \cos(\alpha))] (b \sin(\theta)) \bigg] \bigg] = 0 \end{split}$$

Solving above equation numerically gives $\theta = 35.1^{\circ}$ $\delta = 44.6 \text{ mm}$

COMPUTE REACTIONS

$$R_{C} = \frac{2}{L_{2}} \left[k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_{1} \delta h \right] = 25 \text{ N} \qquad R_{C} = \frac{1}{L_{C}} \left[P \frac{L_{2}}{2} - k_{r} (\alpha - \theta) \right] = 25 \text{ N}$$

$$R_{A} = \frac{2}{L_{2}} \left[k_{1} \delta h + k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_{r} (\alpha - \theta) \right] = 25 \text{ N} \qquad M_{A} = k_{r} (\alpha - \theta) = 0$$

$$R_{A} + R_{C} = 50 \text{ N} \qquad < \text{check} \qquad \qquad \boxed{R_{A} = 25 \text{ N}} \qquad \boxed{R_{C} = 25 \text{ N}}$$

(b) Substitute numerical values, then solve numerically for angle heta and distance increase δ

$$b = 200 \text{ mm} \qquad k = 3.2 \text{ kN/m} \qquad \alpha = 45^{\circ} \qquad P = 50 \text{ N} \qquad k_{1} = \frac{k}{2} \qquad k_{r} = \frac{k}{2} \ b^{2}$$

$$L_{2} = 2b \cos(\theta) \qquad L_{1} = 2b \cos(\alpha) \qquad \delta = L_{2} - L_{1} \qquad \delta = 2b (\cos(\theta) - \cos(\alpha)) \qquad h = b \sin(\theta)$$

$$\frac{1}{L_{2}} \left[P \frac{L_{2}}{2} - k_{r}(\alpha - \theta) \right] - \left[\frac{2}{L_{2}} \left[k \frac{2b (\cos(\theta) - \cos(\alpha)) b \sin(\theta)}{2} + k_{1} [2b (\cos(\theta) - \cos(\alpha))] (b \sin(\theta)) \right] \right] = 0$$
Solving above equation numerically gives $\theta = 43.3^{\circ}$ $\delta = 8.19 \text{ mm}$

$$Compute \text{ REACTIONS}$$

$$R_{C} = \frac{2}{L_{2}} \left[k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_{1} \delta h \right] = 18.5 \text{ N} \qquad R_{2} = \frac{1}{L_{2}} \left[P \frac{L_{2}}{2} - k_{r}(\alpha - \theta) \right] = 18.5 \text{ N}$$

$$R_{A} = \frac{2}{L_{2}} \left[k_{1} \delta h + k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_{r}(\alpha - \theta) \right] = 31.5 \text{ N} \qquad M_{A} = k_{r}(\alpha - \theta) = 1.882 \text{ N·m}$$

$$R_{A} + R_{C} = 50 \text{ N} \qquad < \text{check} \qquad \overline{R_{A}} = 31.5 \text{ N} \qquad \overline{R_{C}} = 18.5 \text{ N} \qquad \overline{M_{A}} = 1.882 \text{ N·m}$$

Apply the laws of statics to the structure in its displaced position; also use FBD's of the left and right bars alone (referred to as LHFB and RHFB below)

OVERALL FBD
$$\sum F_H = 0$$
 $H_A - k_1 \delta = 0$ so $H_A = k_1 \delta$
 $\sum F_V = 0$ $R_A + R_C = P$
 $\sum M_A = 0$ $k_r(\alpha - \theta) - P \frac{L_2}{2} + R_C L_2 = 0$ $R_C = \frac{1}{L_2} \left[P \frac{L_2}{2} - k_r(\alpha - \theta) \right]$
LHFB $\sum M_B = 0$ $H_A h + k \frac{\delta}{2} \left(\frac{h}{2} \right) - R_A \frac{L_2}{2} + k_r(\alpha - \theta) = 0$
 $R_A = \frac{2}{L_2} \left[k_1 \delta h + k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_r(\alpha - \theta) \right]$
RHFB $\sum M_B = 0$ $-k \frac{\delta}{2} \left(\frac{h}{2} \right) - k_1 \delta h + R_C \frac{L_2}{2} = 0$ $R_C = \frac{2}{L_2} \left[k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_1 \delta h \right]$

Equate the two expressions above for R_C then substitute expressions for L_2 , k_r , k_1 , h, and δ

$$\frac{1}{L_2} \left[P \frac{L_2}{2} - k_r(\alpha - \theta) \right] = \frac{2}{L_2} \left[k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_1 \delta h \right] \quad \text{OR}$$

$$\frac{1}{L_2} \left[P \frac{L_2}{2} - k_r(\alpha - \theta) \right] - \left[\frac{2}{L_2} \left[k \frac{2b(\cos(\theta) - \cos(\alpha))}{2} \frac{b\sin(\theta)}{2} + k_1 [2b(\cos(\theta) - \cos(\alpha))](b\sin(\theta)) \right] \right] = 0$$

(a) Substitute numerical values, then solve numerically for angle heta and distance increase δ

$$b = 8 \text{ in.} \quad k = 16 \text{ lb/in.} \quad \alpha = 45^{\circ} \quad P = 101b \quad k_{1} = 0 \quad k_{r} = 0$$

$$L_{2} = 2b\cos(\theta) \quad L_{1} = 2b\cos(\alpha) \quad \delta = L_{2} - L_{1} \quad \delta = 2b(\cos(\theta) - \cos(\alpha)) \quad h = b\sin(\theta)$$

$$\frac{1}{L_{2}} \left[P\frac{L_{2}}{2} - k_{r}(\alpha - \theta) \right] - \left[\frac{2}{L_{2}} \left[k \frac{2b(\cos(\theta) - \cos(\alpha))b\sin(\theta)}{2} + k_{1}[2b(\cos(\theta) - \cos(\alpha))](b\sin(\theta)) \right] \right] = 0$$

Solving above equation numerically gives $\theta = 35.1^{\circ}$ $\delta = 1.782$ in.

COMPUTE REACTIONS

$$R_C = \frac{2}{L_2} \left[k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_1 \delta h \right] = 5 \text{ lb} \qquad R_C = \frac{1}{L_C} \left[P \frac{L_2}{2} - k_r (\alpha - \theta) \right] = 5 \text{ lb}$$

$$R_A = \frac{2}{L_2} \left[k_1 \delta h + k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_1 (\alpha - \theta) \right] = 5 \text{ lb} \qquad M_A = k_r (\alpha - \theta) = 0$$

$$R_A + R_C = 10 \text{ lb} \qquad < \text{check} \qquad \overline{R_A = 5 \text{ lb}} \qquad \overline{R_C = 5 \text{ lb}}$$

(b) Substitute numerical values, then solve numerically for angle θ and distance increase δ

$$b=8 \text{ in.} \quad k=16 \text{ lb/in.} \quad \alpha=45^{\circ} \quad P=101b \quad k_1=\frac{k}{2} \quad k_r=\frac{k}{2} \ b^2$$

$$L_2=2b\cos(\theta) \quad L_1=2b\cos(\alpha) \quad \delta=L_2-L_1 \quad \delta=2b(\cos(\theta)-\cos(\alpha)) \quad h=b\sin(\theta)$$

$$\frac{1}{L_2}\bigg[P\frac{L_2}{2}-k_r(\alpha-\theta)\bigg]-\bigg[\frac{2}{L_2}\bigg[k\frac{2b(\cos(\theta)-\cos(\alpha))}{2}\frac{b\sin(\theta)}{2}+k_1[2b(\cos(\theta)-\cos(\alpha))](b\sin(\theta))\bigg]\bigg]=0$$
 Solving above equation numerically gives
$$\theta=43.3^{\circ}\bigg[\delta=0.327 \text{ in.}\bigg]$$

COMPUTE REACTIONS

$$R_C = \frac{2}{L_2} \left[k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_1 \delta h \right] = 3.71 \,\text{lb} \qquad R_C = \frac{1}{L_2} \left[P \frac{L_2}{2} - k_r (\alpha - \theta) \right] = 3.71 \,\text{lb}$$

$$R_A = \frac{2}{L_2} \left[k_1 \delta h + k \frac{\delta}{2} \left(\frac{h}{2} \right) + k_r (\alpha - \theta) \right] = 6.3 \,\text{lb} \qquad M_A = k_r (\alpha - \theta) = 1.252 \,\text{ft·lb}$$

$$R_A + R_C = 10.01 \,\text{lb} \qquad < \text{check} \qquad \boxed{R_A = 6.3 \,\text{lb}} \qquad \boxed{R_C = 3.71 \,\text{lb}} \qquad \boxed{M_A = 1.252 \,\text{lb·ft}}$$

NUMERICAL DATA

$$P = 3 \text{ k}$$
 $L_1 = 20 \text{ in.}$ $L_2 = 50 \text{ in.}$ $d_A = 0.5 \text{ in.}$ $d_B = 1 \text{ in.}$ $E = 18000 \text{ ksi}$

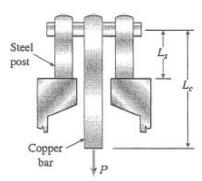
(a) Total elongation

$$\delta_1 = \frac{4PL_1}{\pi E d_A d_B} = 0.00849 \text{ in.} \qquad \delta_2 = \frac{PL_2}{E\frac{\pi}{4} d_B^2} = 0.01061 \text{ in.}$$

$$\delta = 2\delta_1 + \delta_2 = 0.0276 \text{ in.}$$
 $\delta = 0.0276 \text{ in.}$

(b) Find New Diameters at \emph{B} and \emph{C} if total elongation cannot exceed 0.025 in.

$$2\left(\frac{4PL_1}{\pi E d_A d_B}\right) + \frac{PL_2}{E\frac{\pi}{4}d_B^2} = 0.025 \text{ in.}$$
 Solving for d_B : $d_B = 1.074 \text{ in.}$



$$L_c = 2.0 \text{ m}$$

 $A_c = 4800 \text{ mm}^2$

 $E_c = 120 \text{ GPa}$

 $L_{\rm s} = 0.5 \; {\rm m}$

 $A_s = 4500 \text{ mm}^2$

 $E_s = 200 \text{ GPa}$

(a) Downward displacement δ (P = 180 kN)

$$\delta_c = \frac{PL_c}{E_c A_c} = \frac{(180 \text{ kN})(2.0 \text{ m})}{(120 \text{ GPa})(4800 \text{ mm}^2)}$$

$$= 0.625 \text{ mm}$$

$$\delta_s = \frac{(P/2)L_s}{E_s A_s} = \frac{(90 \text{ kN})(0.5 \text{ m})}{(200 \text{ GPa})(4500 \text{ mm}^2)}$$

$$= 0.050 \text{ mm}$$

$$\delta = \delta_c + \delta_s = 0.625 \text{ mm} + 0.050 \text{ mm}$$

$$= 0.675 \text{ mm} \leftarrow$$

(b) Maximum load P_{max} ($\delta_{\text{max}} = 1.0 \text{ mm}$)

$$\frac{P_{\text{max}}}{P} = \frac{\delta_{\text{max}}}{\delta} \quad P_{\text{max}} = P\left(\frac{\delta_{\text{max}}}{\delta}\right)$$

$$P_{\text{max}} = (180 \text{ kN}) \left(\frac{1.0 \text{ mm}}{0.675 \text{ mm}}\right) = 267 \text{ kN} \quad \leftarrow$$

NUMERICAL DATA

$$A = 0.40 \text{ in.}^2$$
 $P_1 = 1700 \text{ lb}$

$$P_2 = 1200 \text{ lb}$$
 $P_3 = 1300 \text{ lb}$

$$E = 10.4 (10^6) \text{ psi}$$

$$a = 60 \text{ in.}$$
 $b = 24 \text{ in.}$ $c = 36 \text{ in.}$

(a) Total elongation

$$\delta = \frac{1}{EA} \left[(P_1 + P_2 - P_3)a + (P_2 - P_3)b + (-P_3)c \right] = 0.01125 \text{ in. } \boxed{\delta = 0.01125 \text{ in.}} \text{ (elongation)}$$

(b) Increase P_3 so that bar does not change length

$$\frac{1}{EA} [(P_1 + P_2 - P_3)a + (P_2 - P_3)b + (-P_3)c] = 0 \text{ solve, } P_3 = 1690 \text{ lb}$$
So new value of P_3 is 1690 lb, an increase of 390 lb.

(c) Now change cross-sectional area of AB so that bar does not change length $P_3 = 1300 \text{ lb}$

$$\frac{1}{E} \left[(P_1 + P_2 - P_3) \frac{a}{A_{AB}} + (P_2 - P_3) \frac{b}{A} + (-P_3) \frac{c}{A} \right] = 0$$

Solving for
$$A_{AB}$$
: $A_{AB} = 0.78 \text{ in.}^2$ $A_{AB} = 1.951$

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Problem 2.3-4

E = 200GPa

$$A_1 = 6000 \text{mm}^2$$
 $A_2 = 5000 \text{mm}^2$ $A_3 = 4000 \text{mm}^2$ $L_1 = 500 \text{mm}$ $L_2 = L_1$ $L_3 = L_1$

$$P_{B} = 50N$$
 $P_{C} = 250N$ $P_{E} = 350N$

Internal forces in each segment (tension +) - cut bar and use lower FBD

$$N_{AB} = -P_B + P_C + P_E = 550 \,N$$
 $N_{BC} = P_C + P_E = 600 \,N$ $N_{CD} = P_E = 350 \,N$ $N_{DE} = P_E = 350 \,N$

Use force-displacement relation to find segment elongations then sum elongations to find downward displacements

$$\delta_{B} = \frac{N_{AB} \cdot L_{1}}{E \cdot A_{1}} = 2.292 \times 10^{-4} \cdot \text{mm} \quad \text{downward}$$

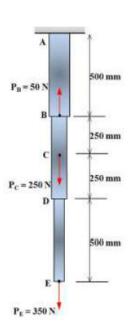
$$\delta_{C} = \delta_{B} + \frac{N_{BC} \cdot \frac{L_{2}}{2}}{E \cdot A_{2}} = 3.792 \times 10^{-4} \cdot \text{mm}$$

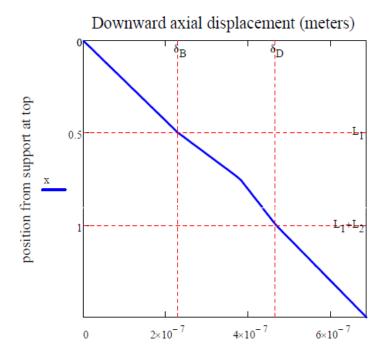
$$\delta_{D} = \delta_{C} + \frac{N_{CD} \cdot \frac{L_{2}}{2}}{E \cdot A_{2}} = 4.667 \times 10^{-4} \cdot \text{mm}$$

$$\delta_{E} = \delta_{D} + \frac{N_{DE} \cdot L_{3}}{E \cdot A_{3}} = 6.854 \times 10^{-4} \cdot \text{mm}$$

Axial displacement diagram - x origin at A, positive downward

$$\begin{split} \delta(x) &= \left| \begin{array}{l} \delta_B \cdot \frac{x}{L_1} & \text{if } x \leq L_1 \\ \\ \delta_B + \left(\delta_C - \delta_B \right) \cdot \left(\frac{x - L_1}{\frac{L_2}{2}} \right) & \text{if } L_1 \leq x \leq L_1 + \frac{L_2}{2} \\ \\ \delta_C + \left(\delta_D - \delta_C \right) \cdot \left[\frac{x - \left(L_1 + \frac{L_2}{2} \right)}{\frac{L_2}{2}} \right] & \text{if } L_1 + \frac{L_2}{2} \leq x \leq L_1 + L_2 \\ \\ \delta_D + \left(\delta_E - \delta_D \right) \cdot \left[\frac{x - \left(L_1 + L_2 \right)}{L_3} \right] & \text{otherwise} \end{split}$$





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Problem 2.3-5

$$E = 29000ksi$$
 $A = 8.24in^2$ $L_1 = 20in$ $L_2 = 20in$ $L_3 = 40in$

$$P_{\rm B} = 501 {\rm bf}$$
 $P_{\rm C} = 1001 {\rm bf}$ $P_{\rm D} = 2001 {\rm bf}$

Internal forces in each segment (tension +) - cut bar and use lower FBD

$${\rm N_{AB}} \, = \, - P_{\rm B} \, + \, P_{\rm C} \, - \, P_{\rm D} \, = \, -150 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{BC}} \, = \, P_{\rm C} \, - \, P_{\rm D} \, = \, -100 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, - P_{\rm D} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \qquad {\rm N_{CD}} \, = \, -200 \cdot 1 \\ {\rm lbf} \qquad \sim \, -200 \cdot 1 \\ {\rm lbf} \qquad \sim \, -200 \cdot 1 \\ {\rm lbf} \qquad \sim \, -200 \cdot 1 \\ {\rm lbf} \qquad \sim \, -200 \cdot 1 \\ {\rm lbf} \qquad \sim \, -200 \cdot 1 \\ {\rm lbf} \qquad \sim \, -200 \cdot 1 \\ {\rm lbf} \qquad \sim \, -200 \cdot 1 \\ {\rm lbf} \qquad \sim \, -200 \cdot 1 \\$$

Use force-displacement relation to find segment elongations then sum elongations to find downward displacements

$$\delta_{B} = \frac{N_{AB} \cdot L_{1}}{E \cdot A} = -1.255 \times 10^{-5} \cdot in \qquad \delta_{C} = \delta_{B} + \frac{N_{BC} \cdot L_{2}}{E \cdot A} = -2.092 \times 10^{-5} \cdot in \qquad \delta_{D} = \delta_{C} + \frac{N_{CD} \cdot L_{3}}{E \cdot A} = -5.44 \times 10^{-5} \cdot in$$
 upward

$$\gamma = 77.0 \frac{kN}{m^3}$$
 from Table I-1

E = 200GPa

$$A_1 = 6000 \text{mm}^2$$
 $A_2 = 5000 \text{mm}^2$ $A_3 = 4000 \text{mm}^2$ $L_1 = 500 \text{mm}$ $L_2 = L_1$ $L_3 = L_1$ $P_B = 50 \text{N}$ $P_C = 250 \text{N}$ $P_F = 350 \text{N}$

Internal forces in each segment (tension +) - cut bar and use lower FBD - weight per unit length = γA,

$$P_{AB} = -P_B + P_C + P_E = 550 \,\mathrm{N}$$
 $P_{BC} = P_C + P_E = 600 \,\mathrm{N}$ $P_{CD} = P_E = 350 \,\mathrm{N}$ $P_{DE} = P_E = 350 \,\mathrm{N}$

Now add weight per unit length - x origin at A, positive downward

$$\begin{aligned} \mathbf{N}_{AB}(\mathbf{x}) &= \mathbf{P}_{AB} + \gamma \cdot \mathbf{A}_{1} \cdot \left(\mathbf{L}_{1} - \mathbf{x}\right) + \gamma \cdot \mathbf{A}_{2} \cdot \mathbf{L}_{2} + \gamma \cdot \mathbf{A}_{3} \cdot \mathbf{L}_{3} & \mathbf{N}_{BC}(\mathbf{x}) &= \mathbf{P}_{BC} + \gamma \cdot \mathbf{A}_{2} \cdot \left(\mathbf{L}_{1} + \frac{\mathbf{L}_{2}}{2} - \mathbf{x}\right) + \gamma \cdot \mathbf{A}_{2} \cdot \frac{\mathbf{L}_{2}}{2} + \gamma \cdot \mathbf{A}_{3} \cdot \mathbf{L}_{3} \\ \mathbf{N}_{CD}(\mathbf{x}) &= \mathbf{P}_{CD} + \gamma \cdot \mathbf{A}_{2} \cdot \left(\mathbf{L}_{1} + \mathbf{L}_{2} - \mathbf{x}\right) + \gamma \cdot \mathbf{A}_{3} \cdot \mathbf{L}_{3} & \mathbf{N}_{DE}(\mathbf{x}) &= \mathbf{P}_{DE} + \gamma \cdot \mathbf{A}_{3} \cdot \left(\mathbf{L}_{1} + \mathbf{L}_{2} + \mathbf{L}_{3} - \mathbf{x}\right) \end{aligned}$$

Note that total bar weight is not small compared to applied loads $W = \gamma \cdot (A_1 \cdot L_1 + A_2 \cdot L_2 + A_3 \cdot L_3) = 577.5 \text{ N}$

Use force-displacement relation to find segment elongations then sum elongations to find displacements

$$\Delta_{B} = \int_{0}^{L_{1}} \frac{N_{AB}(x)}{E \cdot A_{1}} dx = 4.217 \times 10^{-4} \cdot mm$$

$$\Delta_{C} = \Delta_{B} + \int_{L_{1}}^{L_{1} + \frac{L_{2}}{2}} \frac{N_{BC}(x)}{E \cdot A_{2}} dx = 6.463 \times 10^{-4} \cdot mm$$

$$\Delta_{D} = \Delta_{C} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2}} \frac{N_{CD}(x)}{E \cdot A_{2}} dx = 7.843 \times 10^{-4} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{1} + L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

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$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

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$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

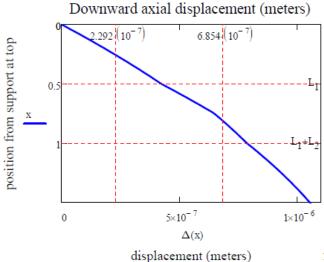
$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

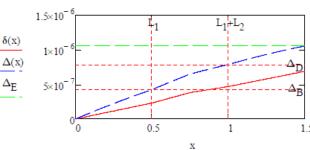
$$\Delta_{E} = \Delta_{D} + \int_{L_{1} + L_{2}}^{L_{2} + L_{3}} \frac{N_{DE}(x)}{E \cdot A_{3}} dx = 1.051 \times 10^{-3} \cdot mm$$

Axial displacement diagram including weight of bar - x origin at A, positive downward

$$\begin{split} \Delta(x) &= \left| \int_0^x \frac{N_{AB}(x)}{E \cdot A_1} \, \mathrm{d}x \ \text{if} \ x \leq L_1 \right. \\ \Delta_B &+ \left. \int_{L_1}^x \frac{N_{BC}(x)}{E \cdot A_2} \, \mathrm{d}x \ \text{if} \ L_1 \leq x \leq L_1 + \frac{L_2}{2} \right. \\ \Delta_C &+ \left. \int_{L_1 + \frac{L_2}{2}}^x \frac{N_{CD}(x)}{E \cdot A_2} \, \mathrm{d}x \ \text{if} \ L_1 + \frac{L_2}{2} \leq x \leq L_1 + L_2 \right. \\ \Delta_D &+ \left. \int_{L_1 + L_2}^x \frac{N_{DE}(x)}{E \cdot A_3} \, \mathrm{d}x \ \text{if} \ x \geq L_1 + L_2 \right. \end{split}$$



Compare ADD's without ($\delta)$ & with (Δ) weight of bar (plotted horizontally)



$$E = 29000 \text{ksi}$$
 $A = 8.24 \text{in}^2$ $\gamma = 490 \frac{\text{lbf}}{\text{ft}^3}$

$$L_1 = 20in$$
 $L_2 = 20in$ $L_3 = 40in$

$$P_B = 50lbf$$
 $P_C = 100lbf$ $P_D = 200lbf$

Internal forces in each segment (tension +) - cut bar and use lower FBD x origin at A, positive downward

$$P_{AB} = -P_B + P_C - P_D = -150 \cdot lbf$$
 $P_{BC} = P_C - P_D = -100 \cdot lbf$ $P_{CD} = -P_D = -200 \cdot lbf$

$$P_{BC} = P_{C} - P_{D} = -100 \cdot 1bf$$

$$P_{CD} = -P_{D} = -200 \cdot 10f$$

$$\mathrm{N}_{AB}(x) = \mathrm{P}_{AB} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 - x \right) + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_2 + \mathsf{L}_3 \right) \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC} + \gamma \cdot \mathrm{A} \cdot \left(\mathsf{L}_1 + \mathsf{L}_2 - x \right) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC}(x) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC}(x) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC}(x) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC}(x) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC}(x) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC}(x) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3 \\ \mathrm{N}_{BC}(x) = \mathrm{P}_{BC}(x) + \gamma \cdot \mathrm{A} \cdot \mathsf{L}_3$$

$$N_{BC}(x) = P_{BC} + \gamma \cdot A \cdot (L_1 + L_2 - x) + \gamma \cdot A \cdot L_3$$

$$N_{CD}(x) = P_{CD} + \gamma \cdot A \cdot (L_1 + L_2 + L_3 - x)$$

Note that total bar weight is not small compared to applied loads $W = \gamma \cdot A \cdot (L_1 + L_2 + L_3) = 186.926 \cdot lbf$

$$W = \gamma \cdot A \cdot (L_1 + L_2 + L_3) = 186.926 \cdot lbt$$

Use force-displacement relation to find segment elongations then sum elongations to find downward displacements

$$\delta_{\mathbf{B}} = \int_{0}^{L_{1}} \frac{N_{\mathbf{A}\mathbf{B}}(\mathbf{x})}{\mathbf{E} \cdot \mathbf{A}} d\mathbf{x} = 1.135 \times 10^{-6} \cdot \mathbf{i} \cdot \mathbf{a}$$

$$\delta_{B} = \int_{0}^{L_{1}} \frac{N_{AB}(x)}{E \cdot A} dx = 1.135 \times 10^{-6} \cdot in \qquad \delta_{C} = \delta_{B} + \int_{L_{1}}^{L_{1} + L_{2}} \frac{N_{BC}(x)}{E \cdot A} dx = 2.543 \times 10^{-6} \cdot in$$
downward

$$\delta_D = \delta_C + \int_{L_1 + L_2}^{L_1 + L_2 + L_3} \frac{N_{CD}(x)}{\text{E-A}} \, dx = -2.311 \times 10^{-5} \cdot \text{in}$$
 upward

(a)
$$\delta = \frac{P}{E} \left(\frac{2\frac{L}{4}}{bt} + \frac{\frac{L}{2}}{\frac{3}{4}bt} \right) = \frac{7LP}{6Ebt}$$
 $\delta = \frac{7PL}{6Ebt}$

(b) Numerical data
$$E=210~\mathrm{GPa}$$
 $L=750~\mathrm{mm}$ $\sigma_{\mathrm{mid}}=160~\mathrm{MPa}$

so
$$\sigma_{\text{mid}} = \frac{P}{\frac{3}{4}bt}$$
 and $\frac{P}{bt} = \frac{3}{4}\sigma_{\text{mid}}$

$$\delta = \frac{7LP}{6Ebt}$$
 or $\delta = \frac{7L}{6E} \left(\frac{3}{4} \sigma_{\text{mid}} \right) = 0.5 \text{ mm}$ $\delta = 0.5 \text{ mm}$

(c)
$$\delta_{\text{max}} = \frac{P}{E} \left(\frac{L - L_{\text{slot}}}{bt} + \frac{L_{\text{slot}}}{\frac{3}{4}bt} \right)$$
 or $\delta_{\text{max}} = \left(\frac{P}{bt} \right) \left(\frac{1}{E} \right) \left(L - L_{\text{slot}} + \frac{4}{3} L_{\text{slot}} \right)$

or
$$\delta_{\rm max} = \left(\frac{3}{4}\,\sigma_{\rm mid}\right)\left(\frac{1}{E}\right)\left(L + \frac{L_{\rm slot}}{3}\right)$$
 Solving for $L_{\rm slot}$ with $\delta_{\rm max} = 0.475\,{\rm mm}$

$$L_{\rm slot} = \frac{4E\delta_{\rm max} - 3L\sigma_{\rm mid}}{\sigma_{\rm mid}} \ = 244 \ {\rm mm} \quad \boxed{L_{\rm slot} = 244 \ {\rm mm}} \quad \frac{L_{\rm slot}}{L} = 0.325$$

(a)
$$\delta = \frac{P}{E} \left(\frac{2\frac{L}{4}}{bt} + \frac{\frac{L}{2}}{\frac{3}{4}bt} \right) = \frac{7LP}{6Ebt}$$

(b)
$$E = 30,000 \text{ ksi}$$
 $L = 30 \text{ in.}$ $\sigma_{\text{mid}} = 24 \text{ ksi}$

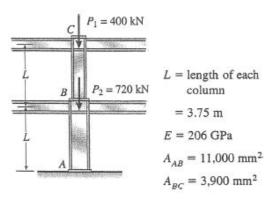
So $\sigma_{\text{mid}} = \frac{P}{\frac{3}{4}bt}$ and $\frac{P}{bt} = \frac{3}{4}\sigma_{\text{mid}}$

$$\delta = \frac{7LP}{6Ebt}$$
 or $\delta = \frac{7L}{6E} \left(\frac{3}{4} \sigma_{\text{mid}} \right) = 0.021 \text{ in.}$ $\delta = 0.021 \text{ in.}$

(c)
$$\delta_{\text{max}} = \frac{P}{E} \left(\frac{L - L_{\text{slot}}}{bt} + \frac{L_{\text{slot}}}{\frac{3}{4}bt} \right)$$
 or $\delta_{\text{max}} = \left(\frac{P}{bt} \right) \left(\frac{1}{E} \right) \left(L - L_{\text{slot}} + \frac{4}{3}L_{\text{slot}} \right)$

or
$$\delta_{\max} = \left(\frac{3}{4}\sigma_{\min}\right)\left(\frac{1}{E}\right)\left(L + \frac{L_{\text{slot}}}{3}\right)$$
 Solving for L_{slot} with $\delta_{\max} = 0.02$ in.:

$$L_{\rm slot} = \frac{4E\delta_{\rm max} - 3L\sigma_{\rm mid}}{\sigma_{\rm mid}} = 10 \text{ in.}$$
 $L_{\rm slot} = 10 \text{ in.}$ $L_{\rm slot} = 0.333$



(a) Shortening δ_{AC} of the two columns

$$\begin{split} \delta_{AC} &= \sum \frac{N_i L_i}{E_i A_i} = \frac{N_{AB} L}{E A_{AB}} + \frac{N_{BC} L}{E A_{BC}} \\ &= \frac{(1120 \text{ kN})(3.75 \text{ m})}{(206 \text{ GPa})(11,000 \text{ mm}^2)} \\ &+ \frac{(400 \text{ kN})(3.75 \text{ m})}{(206 \text{ GPa})(3,900 \text{ mm}^2)} \\ &= 1.8535 \text{ mm} + 1.8671 \text{ mm} = 3.7206 \text{ mm} \\ \delta_{AC} &= 3.72 \text{ mm} & \longleftarrow \end{split}$$

(b) Additional load P_0 at point C

$$(\delta_{AC})_{\text{max}} = 4.0 \text{ mm}$$

 δ_0 = additional shortening of the two columns due to the load P_0

$$\delta_0 = (\delta_{AC})_{\text{max}} - \delta_{AC} = 4.0 \text{ mm} - 3.7206 \text{ mm}$$

= 0.2794 mm

Also,
$$\delta_0 = \frac{P_0 L}{E A_{AB}} + \frac{P_0 L}{E A_{BC}} = \frac{P_0 L}{E} \left(\frac{1}{A_{AB}} + \frac{1}{A_{BC}} \right)$$

Solve for P_0 :

$$P_0 = \frac{E\delta_0}{L} \left(\frac{A_{AB} A_{BC}}{A_{AB} + A_{BC}} \right)$$

SUBSTITUTE NUMERICAL VALUES:

$$E = 206 \times 10^9 \text{ N/m}^2$$
 $\delta_0 = 0.2794 \times 10^{-3} \text{ m}$
 $L = 3.75 \text{ m}$ $A_{AB} = 11,000 \times 10^{-6} \text{ m}^2$
 $A_{BC} = 3,900 \times 10^{-6} \text{ m}^2$
 $P_0 = 44,200 \text{ N} = 44.2 \text{ kN}$ \leftarrow

$$\underline{\text{Numerical data}} \qquad \text{E = } 30 \cdot \left(10^6\right) \text{psi} \qquad \text{P = } 50001 \text{b} \qquad \text{L = 4ft} \qquad d_1 = 0.75 \text{in} \qquad d_2 = 0.5 \text{in}$$

$$\delta_{a} = \frac{P \cdot L}{E} \cdot \left(\frac{1}{\frac{\pi}{4} \cdot d_{1}^{2}} + \frac{1}{\frac{\pi}{4} \cdot d_{2}^{2}} \right) = 0.0589 \cdot in$$

$$\boxed{\delta_{a} = 0.0589 \cdot in}$$

$$\delta_b = \frac{P \cdot (2 \cdot L)}{E \cdot A} = 0.0501 \cdot in \qquad \qquad \boxed{\delta_b = 0.0501 \cdot in}$$

$$\frac{\text{Part (c)}}{\delta_c} = \frac{q \cdot L^2}{2 \cdot E \cdot \left(\frac{\pi}{4} \cdot d_1^2\right)} + \frac{p \cdot L}{E \cdot \left(\frac{\pi}{4} \cdot d_2^2\right)} = 0.0498 \cdot \text{in} \qquad \boxed{\frac{\delta_c}{\delta_a} = 0.846} \qquad \boxed{\frac{\delta_c}{\delta_b} = 0.993}$$

NUMERICAL DATA

$$d_1 = 100 \text{ mm}$$
 $d_2 = 60 \text{ mm}$

$$L = 1200 \text{ mm}$$
 $E = 4.0 \text{ GPa}$ $P = 110 \text{ kN}$

$$\delta_a = 8.0 \text{ mm}$$

(a) Find d_{\max} if shortening is limited to δ_a

$$A_1 = \frac{\pi}{4} d_1^2$$
 $A_2 = \frac{\pi}{4} d_2^2$

$$\delta = \frac{P}{E} \left[\frac{\frac{L}{4}}{\frac{\pi}{4} (d_1^2 - d_{\text{max}}^2)} + \frac{\frac{L}{4}}{A_1} + \frac{\frac{L}{2}}{A_2} \right]$$

Set δ to δ_a , and solve for d_{max} :

$$d_{\text{max}} = d_1 \sqrt{\frac{E\delta_a \pi d_1^2 d_2^2 - 2PLd_2^2 - 2PLd_1^2}{E\delta_a \pi d_1^2 d_2^2 - PLd_2^2 - 2PLd_1^2}}$$

$$d_{\text{max}} = 23.9 \text{ mm} \leftarrow$$

(b) Now, if $d_{\rm max}$ is instead set at $d_2/2$, at what distance b from end C should load P be applied to limit the bar shortening to $\delta_a=8.0$ mm?

$$A_0 = \frac{\pi}{4} \left[d_1^2 - \left(\frac{d_2}{2} \right)^2 \right]$$

$$A_1 = \frac{\pi}{4}{d_1}^2$$
 $A_2 = \frac{\pi}{4}{d_2}^2$

$$\delta = \frac{P}{E} \left[\frac{L}{4A_0} + \frac{L}{4A_1} + \frac{\left(\frac{L}{2} - b\right)}{A_2} \right]$$

No axial force in segment at end of length b; set $\delta = \delta_a$ and solve for b:

$$b = \left[\frac{L}{2} - A_2 \left[\frac{E\delta_a}{P} - \left(\frac{L}{4A_0} + \frac{L}{4A_1}\right)\right]\right]$$

(c) Finally if loads P are applied at the ends and $d_{\max}=d_2/2$, what is the permissible length x of the hole if shortening is to be limited to $\delta_a=8.0~\mathrm{mm}$?

$$\delta = \frac{P}{E} \left[\frac{x}{A_0} + \frac{\left(\frac{L}{2} - x\right)}{A_1} + \frac{\left(\frac{L}{2}\right)}{A_2} \right]$$

Set $\delta = \delta_a$ and solve for x:

$$x = \frac{\left[A_0 A_1 \left(\frac{E\delta_a}{P} - \frac{L}{2A_2}\right)\right] - \frac{1}{2}A_0 L}{A_1 - A_0}$$

AFD LINEAR

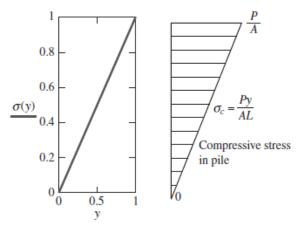
(a)
$$N(y) = fy$$

$$\delta = \int_0^L \frac{(fy)}{EA} dy = \frac{L^2 f}{2AE} \qquad \delta = \frac{PL}{2EA}$$

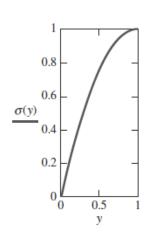
(b)
$$\sigma(y) = \frac{N(y)}{A}$$
 $\sigma(y) = \frac{fy}{A}$ $\sigma(L) = \frac{fL}{A} = \frac{P}{A}$

 $\sigma(0) = 0$ So linear variation, zero at bottom, P/A at top (i.e., at ground surface)

$$N(L) = f$$
 $\sigma(y) = \frac{P}{A} \left(\frac{y}{L}\right)$



f(y) is constant and AFD is linear



f(y) is linear and AFD quadratic

(c)
$$N(y) = f(y)y$$

$$N(y) = \int_0^y f_0 \left(1 - \frac{\zeta}{L} \right) d\zeta = \frac{f_0 y (y - 2)}{2} \qquad \qquad N(L) = \frac{f_0}{2} \qquad \qquad N(0) = 0$$

$$\delta = \frac{\left(\frac{f_0L}{2}\right)}{\frac{3}{2}EA} \qquad P = \frac{1}{2}f_0L \qquad \left[\delta = \frac{PL}{EA}\left(\frac{2}{3}\right)\right] \qquad \sigma(y) = \frac{P}{A}\left[\frac{y}{L}\left(2 - \frac{y}{L}\right)\right] \qquad \sigma(0) = 0 \qquad \sigma(L) = \frac{f_0}{2} = P/A$$

NUMERICAL DATA

$$P = 5 \text{ kN}$$
 $E_c = 120 \text{ GPa}$

$$L_2 = 18 \text{ mm}$$
 $L_4 = L_2$

$$L_3 = 40 \text{ mm}$$

$$d_{o3} = 22.2 \text{ mm}$$
 $t_3 = 1.65 \text{ mm}$

$$d_{o5} = 18.9 \text{ mm}$$
 $t_5 = 1.25 \text{ mm}$

$$\tau_Y = 30 \text{ MPa}$$
 $\sigma_Y = 200 \text{ MPa}$

$$FS_{\tau} = 2$$
 $FS_{\sigma} = 1.7$

$$\tau_a = \frac{\tau_Y}{FS_{\tau}}$$
 $\tau_a = 15 \text{ MPa}$

$$\sigma_a = \frac{\sigma_Y}{FS_{\sigma}}$$
 $\sigma_a = 117.6 \text{ MPa}$

(a) Elongation of segment 2-3-4

$$A_2 = \frac{\pi}{4} [d_{o3}^2 - (d_{o5} - 2t_5)^2]$$

$$A_3 = \frac{\pi}{4} [d_{o3}^2 - (d_{o3} - 2t_3)^2]$$

$$A_2 = 175.835 \text{ mm}^2$$
 $A_3 = 106.524 \text{ mm}^2$

$$\delta_{24} = \frac{P}{E_c} \left(\frac{L_2 + L_4}{A_2} + \frac{L_3}{A_3} \right)$$

$$\delta_{24} = 0.024 \, \text{mm} \quad \leftarrow$$

(b) Maximum load P_{\max} that can be applied to the joint

First check normal stress:

$$A_1 = \frac{\pi}{4} [d_{o5}^2 - (d_{o5} - 2t_5)^2]$$

A₁ = 69.311 mm² < smallest cross-sectional area controls normal stress

 $P_{\max\sigma} = \sigma_a A_1$ $P_{\max\sigma} = 8.15 \text{ kN} \leftarrow \text{smaller than}$ P_{\max} based on shear below so normal stress controls

Next check shear stress in solder joint:

$$A_{\rm sh} = \pi d_{o5} L_2$$
 $A_{\rm sh} = 1.069 \times 10^3 \, \rm mm^2$

$$P_{\text{max}\tau} = \tau_a A_{\text{sh}}$$
 $P_{\text{max}\tau} = 16.03 \text{ kN}$

(c) Find the value of L₂ at which tube and solder CAPACITIES ARE EQUAL

Set P_{max} based on shear strength equal to P_{max} based on tensile strength and solve for L_2 :

$$L_2 = \frac{\sigma_a A_1}{\tau_a (\pi d_{o5})} \qquad L_2 = 9.16 \text{ mm} \quad \leftarrow$$

(a) STATICS
$$\sum F_H = 0$$
 $R_1 = -P - \frac{P}{2}$ $R_1 = \frac{-3}{2}P$ \leftarrow

(b) DRAW FBD's CUTTING THROUGH SEGMENT 1 AND AGAIN THROUGH SEGMENT 2

$$N_1 = \frac{3P}{2}$$
 < tension $N_2 = \frac{P}{2}$ < tension

(c) Find x required to obtain axial displacement at joint 3 of $\delta_3 = PL/EA$

Add axial deformations of segments 1 and 2, then set to δ_3 ; solve for x:

$$\frac{N_1 x}{E_A^3 A} + \frac{N_2 (L - x)}{EA} = \frac{PL}{EA}$$

$$\frac{\frac{3P}{2}x}{E\frac{3}{4}A} + \frac{\frac{P}{2}(L-x)}{EA} = \frac{PL}{EA}$$

$$\frac{3}{2}x = \frac{L}{2} \quad x = \frac{L}{3} \quad \leftarrow$$

(d) What is the displacement at joint 2, δ_2 ?

$$\delta_2 = \frac{N_1 x}{E_{\frac{3}{4}A}^3} \quad \delta_2 = \frac{\left(\frac{3P}{2}\right) \frac{L}{3}}{E_{\frac{3}{4}A}^3}$$

$$\delta_2 = \frac{2}{3} \frac{PL}{EA}$$

(e) If x = 2L/3 and P/2 at joint 3 is replaced by βP , find β so that $\delta_3 = PL/EA$

$$N_1 = (1 + \beta)P$$
 $N_2 = \beta P$ $x = \frac{2L}{3}$

substitute in axial deformation expression above and solve for β

$$\frac{[(1+\beta)P]\frac{2L}{3}}{E\frac{3}{4}A} + \frac{\beta P\left(L - \frac{2L}{3}\right)}{EA} = \frac{PL}{EA}$$

$$\frac{1}{9}PL\frac{8+11\beta}{EA} = \frac{PL}{EA}$$

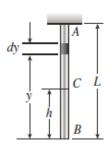
$$(8+11\beta)=9$$

$$\beta = \frac{1}{11} \leftarrow$$

$$\beta = 0.091$$

(f) Draw AFD, ADD—see plots for $x = \frac{L}{3}$

No plots provided here



W =Weight of bar (a) Downward displacement δ_C Consider an element at distance y from the lower end.

$$N(y) = \frac{Wy}{L} \quad d\delta = \frac{N(y)dy}{EA} = \frac{Wydy}{EAL}$$
$$\delta_C = \int_h^L d\delta = \int_h^L \frac{Wydy}{EAL} = \frac{W}{2EAL} (L^2 - h^2)$$
$$\delta_C = \frac{W}{2EAL} (L^2 - h^2) \quad \leftarrow$$

(d) Numerical data

$$v_n = 77 \text{ kN/m}^2$$

$$\gamma_w = 10 \text{ kN/m}^3$$

$$W = (\gamma_c - \gamma_{cc})AL = 1577.85 \text{ kN}$$

In air:

$$W = (\gamma_s)AL = 1813.35 \,\mathrm{kN}$$

(b) Elongation of Bar (h = 0)

$$\delta_B = \frac{WL}{2FA} \leftarrow$$

(c) RATIO OF ELONGATIONS

Elongation of upper half of bar $\left(h = \frac{L}{2}\right)$:

$$\delta_{\text{upper}} = \frac{3WL}{8EA}$$

Elongation of lower half of bar:

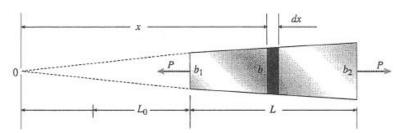
$$\delta_{\text{lower}} = \delta_B - \delta_{\text{upper}} = \frac{WL}{2EA} - \frac{3WL}{8EA} = \frac{WL}{8EA}$$

$$\beta = \frac{\delta_{\text{upper}}}{\delta_{\text{lower}}} = \frac{3/8}{1/8} = 3 \quad \leftarrow$$

$$\gamma_s = 77 \text{ kN/m}^3$$
 $\gamma_w = 10 \text{ kN/m}^3$ $L = 1500 \text{ m}$ $A = 0.0157 \text{ m}^2$ $E = 210 \text{ GPa}$

$$W = (\gamma_s - \gamma_w)AL = 1577.85 \text{ kN}$$
 $\delta = \frac{WL}{2EA} = 359 \text{ mm}$ $\frac{\delta}{L} = 2.393 \times 10^{-4}$

$$\delta = \frac{WL}{2EA} = 412 \text{ mm} \qquad \frac{\delta}{L} = 2.75 \times 10^{-4}$$



t =thickness (constant)

$$b = b_1 \left(\frac{x}{L_0}\right) \quad b_2 = b_1 \left(\frac{L_0 + L}{L_0}\right)$$

(Eq. 1)

$$A(x) = bt = b_1 t \left(\frac{x}{L_0}\right)$$

(a) ELONGATION OF THE BAR

$$d\delta = \frac{Pdx}{EA(x)} = \frac{PL_0 dx}{Eb_1 tx}$$

$$\delta = \int_{L_0}^{L_0 + L} d\delta = \frac{PL_0}{Eb_1 t} \int_{L_0}^{L_0 + L} \frac{dx}{x}$$

$$= \frac{PL_0}{Eb_1 t} \ln x \bigg|_{L_0}^{L_0+L} = \frac{PL_0}{Eb_1 t} \ln \frac{L_0 + L}{L_0}$$
 (Eq. 2)

From Eq. (1):
$$\frac{L_0 + L}{L_0} = \frac{b_2}{b_1}$$
 (Eq. 3)

Solve Eq. (3) for
$$L_0$$
: $L_0 = L\left(\frac{b_1}{b_2 - b_1}\right)$ (Eq. 4)

Substitute Eqs. (3) and (4) into Eq. (2):

$$\delta = \frac{PL}{Et(b_2 - b_1)} \ln \frac{b_2}{b_1}$$
 (Eq. 5)

(b) SUBSTITUTE NUMERICAL VALUES:

$$L = 5 \text{ ft} = 60 \text{ in.}$$
 $t = 10 \text{ in.}$

$$P = 25 \text{ k}$$
 $b_1 = 4.0 \text{ in.}$

$$b_2 = 6.0 \text{ in.}$$
 $E = 30 \times 10^6 \text{ psi}$

From Eq. (5): $\delta = 0.010$ in. \leftarrow

$$P = 200kN$$
 $L = 2m$ $t = 20mm$ $b_1 = 100mm$ $b_2 = 115mm$ $E = 96GPa$

Bar width at B at L/2
$$b_B = \frac{b_1 + b_2}{2} = 107.5 \cdot mm$$

Axial forces in bar segments (use RHFB)
$$N_{AB} = 2 \cdot P - P = 200 \cdot kN \qquad N_{BC} = 2 \cdot P = 400 \cdot kN$$

Axial displacement at B
$$\delta_{B} = \frac{N_{AB} \cdot \frac{L}{2}}{E \cdot t \cdot \left(b_{2} - b_{B}\right)} \cdot ln \left(\frac{b_{2}}{b_{B}}\right) = 0.937 \cdot mm$$

$$P = 50 \text{kip}$$
 $L = 5 \text{ft}$ $t = \frac{3}{8} \cdot \text{in}$ $b_1 = 3 \text{in}$ $b_2 = 2.75 \text{in}$ $E = 16000 \text{ksi}$

$$t = \frac{3}{8} \cdot in$$

$$b_1 = 3in$$

$$b_2 = 2.75 in$$

$$E = 16000 ksi$$

$$N_{AB} = 2 \cdot P - P = 50 \cdot kip$$
 $N_{BC} = 2 \cdot P = 100 \cdot kip$

$$N_{BC} = 2 \cdot P = 100 \cdot kip$$

$$\delta_{\mathbf{B}} = \frac{N_{\mathbf{AB}} \cdot \frac{\mathbf{L}}{2}}{\mathbf{E} \cdot \mathbf{t} \cdot (\mathbf{b}_2 - \mathbf{b}_1)} \cdot \ln \left(\frac{\mathbf{b}_2}{\mathbf{b}_1} \right) = 0.087 \cdot \text{in}$$

$$\delta_{\text{C}} = \delta_{\text{B}} + \frac{N_{\text{BC}} \cdot \frac{L}{2}}{\text{E} \cdot \text{t} \cdot \left(b_2 - b_1\right)} \cdot \ln \left(\frac{b_2}{b_1}\right) = 0.261 \cdot \text{in}$$

Problem 2.3-20

$$E = 72GPa$$
 $P_2 = 200kN$ $L = 2m$ $t = 20mm$ $b_1 = 100mm$ $b_2 = 115mm$

$$A_{BC} = b_1 \cdot t = 2 \times 10^3 \cdot mm^2$$

$$\frac{\text{If only load P}_2 \text{ is applied at C}}{\text{If only load P}_2 \text{ is applied at C}} \qquad \delta_B = \frac{P_2 \cdot \frac{L}{2}}{E \cdot t \cdot (b_2 - b_1)} \cdot \ln \left(\frac{b_2}{b_1}\right) = 1.294 \, \text{mm} \qquad \delta_C = \delta_B + \frac{P_2 \cdot \frac{L}{2}}{E \cdot A_{BC}} = 2.683 \cdot \text{mm}$$

Now apply both P_1 (to the left) and P_2 at C and solve for P_1 s.t. axial displacement at C = 0

Given

$$\frac{\left(P_{2} - P_{1}\right) \cdot \frac{L}{2}}{E \cdot t \cdot \left(b_{2} - b_{1}\right)} \cdot \ln\left(\frac{b_{2}}{b_{1}}\right) + \frac{P_{2} \cdot \frac{L}{2}}{E \cdot A_{BC}} = 0 \qquad \text{Find}(P_{1}) = 414.651 \cdot \text{kN}$$

Axial displacement at B with both loads applied as shown

Let
$$P_1 = 414.651 \text{kN}$$
 Check

$$\delta_{\mathbf{B}} = \frac{\left(P_2 - P_1\right) \cdot \frac{L}{2}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot \ln \left(\frac{b_2}{b_1}\right) = -1.389 \cdot mm$$

$$\frac{\left(P_2 - P_1\right) \cdot \frac{L}{2}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot \ln \left(\frac{b_2}{b_1}\right) + \frac{P_2 \cdot \frac{L}{2}}{E \cdot A_{\mathbf{BC}}} = -0 \, m$$

$$\begin{aligned} d_A &= 4 in & d_B &= 8 in & P &= 45 kip & \delta_A &= 0.02 in & d(x) &= d_A + \left(\frac{d_B - d_A}{L}\right) \cdot x \\ A(x) &= \frac{\pi}{4} \cdot d(x)^2 & E &= 10400 ksi \end{aligned}$$

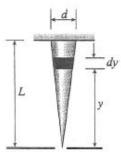
$$\int_{0}^{L} \frac{P}{E \cdot A(x)} \, dx = \delta_{A} \qquad \text{expand integral to obtain following expression} \qquad \frac{4 \cdot P \cdot L}{\pi \cdot E \cdot d_{A} \cdot d_{B}} = \delta_{A}$$
 Solving for L
$$L = \frac{\pi \cdot E \cdot d_{A} \cdot d_{B}}{4 \cdot P} \cdot \delta_{A} = 9.681 \cdot \text{ft}$$

$$A_1 = \pi \cdot r^2 = 4071.504 \cdot mm^2$$
 Use formulas in **Appendix F, Case 15** for area of slotted segment

$$\alpha = a\cos\left(\frac{a}{r}\right) = 1.445 \qquad b = \sqrt{r^2 - a^2} = 35.718 \cdot mm \qquad A_2 = 2 \cdot r^2 \cdot \left(\alpha - \frac{a \cdot b}{r^2}\right) = 3425.196 \cdot mm^2 \qquad \frac{A_2}{A_1} = 0.841$$

Stress in middle half is known so use to find force P
$$P = \sigma_2 \cdot A_2 = 616.535 \cdot kN$$

Compute bar elongation now that P is known
$$\delta = 2 \cdot \frac{P \cdot \frac{L}{4}}{E \cdot A_1} + \frac{P \cdot \frac{L}{2}}{E \cdot A_2} = 4.143 \cdot mm$$



TERMINOLOGY

 $N_{\rm y}$ = axial force acting on element $d{\rm y}$

 A_y = cross-sectional area at element dy

 A_B = cross-sectional area at base of cone

$$= \frac{\pi d^2}{4} \quad V = \text{volume of cone}$$

$$= \frac{1}{3} A_B L \quad V_y = \text{volume of cone below element } dy$$

$$=\frac{1}{3}A_y y$$
 W_y = weight of cone below element dy

$$= \frac{V_y}{V}(W) = \frac{A_y yW}{A_B L} \quad N_y = W_y$$

ELEMENT OF BAR



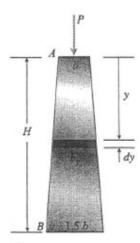
W =weight of cone

ELONGATION OF ELEMENT dy

$$d\delta = \frac{N_y \, dy}{E \, A_y} = \frac{Wy \, dy}{E \, A_B L} = \frac{4W}{\pi d^2 \, EL} \, y \, dy$$

ELONGATION OF CONICAL BAR

$$\delta = \int d\delta = \frac{4W}{\pi d^2 EL} \int_0^L y \, dy = \frac{2WL}{\pi d^2 E} \quad \leftarrow$$



Square cross sections:

$$b = \text{width at } A$$

 $1.5b = \text{width at } B$
 $b_y = \text{width at distance } y$

$$= b + (1.5b - b)\frac{y}{H}$$
$$= \frac{b}{H}(H + 0.5y)$$

 A_y = cross-sectional area at distance y

$$= (b_y)^2 = \frac{b^2}{H^2}(H + 0.5y)^2$$

SHORTENING OF ELEMENT dy

$$d\delta = \frac{Pdy}{EA_y} = \frac{Pdy}{E\left(\frac{b^2}{H^2}\right)(H + 0.5y)^2}$$

SHORTENING OF ENTIRE POST

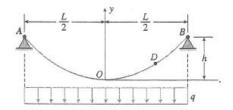
$$\delta = \int d\delta = \frac{PH^2}{Eb^2} \int_0^H \frac{dy}{(H + 0.5y)^2}$$

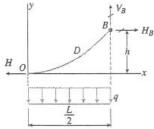
From Appendix C:
$$\int \frac{dx}{(a+bx)^2} = -\frac{1}{b(a+bx)}$$

$$\delta = \frac{PH^2}{Eb^2} \left[-\frac{1}{(0.5)(H + 0.5y)} \right]_0^H$$

$$= \frac{PH^2}{Eb^2} \left[-\frac{1}{(0.5)(1.5H)} + \frac{1}{0.5H} \right]$$

$$= \frac{2PH}{3Eb^2} \leftarrow$$





Equation of parabolic curve:

$$y = \frac{4hx^2}{L^2}$$

$$\frac{dy}{dx} = \frac{8hx}{L^2}$$

FREE-BODY DIAGRAM OF HALF OF CABLE

$$\Sigma M_B = 0 \Leftrightarrow \Delta M_B = 0$$
$$- Hh + \frac{qL}{2} \left(\frac{L}{4}\right) = 0$$

$$H = \frac{qL^2}{8h}$$

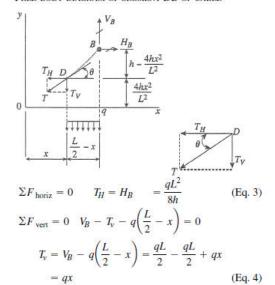
$$\Sigma F_{\text{horizontal}} = 0$$

$$H_B = H = \frac{qL^2}{8h}$$
 (Eq. 1)

$$\Sigma F_{\text{vortical}} = 0$$

$$V_B = \frac{qL}{2} \tag{Eq. 2}$$

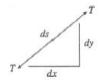
FREE-BODY DIAGRAM OF SEGMENT DB OF CABLE



TENSILE FORCE T IN CABLE

$$T = \sqrt{T_H^2 + T_v^2} = \sqrt{\left(\frac{qL^2}{8h}\right)^2 + (qx)^2}$$
$$= \frac{qL^2}{8h}\sqrt{1 + \frac{64h^2x^2}{L^4}}$$
(Eq. 5)

Elongation $d\delta$ of an element of length ds



$$d\delta = \frac{Tds}{EA}$$

$$ds = \sqrt{(dx)^2 + (dy)^2} = dx\sqrt{1 + \left(\frac{dy}{dx}\right)^2}$$

$$= dx\sqrt{1 + \left(\frac{8hx}{L^2}\right)^2}$$

$$= dx\sqrt{1 + \frac{64h^2x^2}{L^4}}$$
(Eq. 6)

(a) Elongation δ of Cable AOB

$$\delta = \int d\delta = \int \frac{T \, ds}{EA}$$

Substitute for T from Eq. (5) and for ds from Eq. (6):

$$\delta = \frac{1}{EA} \int \frac{qL^2}{8h} \left(1 + \frac{64h^2x^2}{L^4} \right) dx$$

For both halves of cable:

$$\delta = \frac{2}{EA} \int_0^{L/2} \frac{qL^2}{8h} \left(1 + \frac{64h^2x^2}{L^4} \right) dx$$

$$\delta = \frac{qL^3}{8hEA} \left(1 + \frac{16h^2}{3L^4} \right) \quad \leftarrow \quad (Eq. 7)$$

(b) GOLDEN GATE BRIDGE CABLE

$$L = 4200 \text{ ft}$$
 $h = 470 \text{ ft}$
 $q = 12,700 \text{ lb/ft}$ $E = 28,800,000 \text{ psi}$
 $27,572 \text{ wires of diameter } d = 0.196 \text{ in.}$
 $A = (27,572) \left(\frac{\pi}{4}\right) (0.196 \text{ in.})^2 = 831.90 \text{ in.}^2$

Substitute into Eq. (7):

$$\delta = 133.7 \text{ in} = 11.14 \text{ ft} \quad \leftarrow$$

(a) ELONGATION δ FOR CASE OF CONSTANT DIAMETER HOLE

$$d(\zeta) = d_A \left(1 + \frac{\zeta}{L}\right) \qquad A(\zeta) = \frac{\pi}{4} d(\zeta)^2 \qquad < \text{solid portion of length } L - x$$

$$A(\zeta) = \frac{\pi}{4} (d(\zeta)^2 - d_A^2) \qquad < \text{hollow portion of length } x$$

$$\delta = \frac{P}{E} \left(\int \frac{1}{A(\zeta)} d\zeta \right) \qquad \delta = \frac{P}{E} \left[\int_0^{L-x} \frac{4}{\pi d(\zeta)^2} d\zeta + \int_{L-x}^L \frac{4}{\pi (d(\zeta)^2 - d_A^2)} d\zeta \right]$$

$$\delta = \frac{P}{E} \left[\int_0^{L-x} \frac{1}{\left[\frac{\pi}{4} \left[d_A \left(1 + \frac{\zeta}{L} \right) \right]^2 \right]} d\zeta + \int_{L-x}^L \frac{1}{\left[\frac{\pi}{4} \left[\left[d_A \left(1 + \frac{\zeta}{L} \right) \right]^2 - d_A^2 \right] \right]} d\zeta \right]$$

$$\delta = \frac{P}{E} \left[4 \frac{L^2}{(-2 + x)\pi d_A^2} + \left[\left[4 \frac{L}{\pi d_A^2} + \int_{L-x}^L \frac{1}{\left[\frac{\pi}{4} \left[\left[d_A \left(1 + \frac{\zeta}{L} \right) \right]^2 - d_A^2 \right] \right]} d\zeta \right] \right]$$

$$\delta = \frac{P}{E} \left[4 \frac{L^2}{(-2 + x)\pi d_A^2} + \left(4 \frac{L}{\pi d_A^2} - 2L \frac{\ln(3)}{\pi d_A^2} + 2L \frac{-\ln(L-x) + \ln(3L-x)}{\pi d_A^2} \right) \right]$$
if $x = L/2$
$$\delta = \frac{P}{E} \left[\frac{4}{3} \frac{L}{\pi d_A^2} - 2L \frac{\ln(3)}{\pi d_A^2} + 2L \frac{-\ln\left(\frac{1}{2}L\right) + \ln\left(\frac{5}{2}L\right)}{\pi d_A^2} \right)$$

Substitute numerical data:

$$\delta = 2.18 \, \text{mm} \quad \leftarrow$$

(b) elongation δ for case of variable diameter hole but constant wall thickness $t=d_A/20$ over segment x

$$d(\zeta) = d_A \left(1 + \frac{\zeta}{L}\right)$$
 $A(\zeta) = \frac{\pi}{4} d(\zeta)^2$ < solid portion of length $L - x$

$$A(\zeta) = \frac{\pi}{4} \left[d(\zeta)^2 - \left(d(\zeta) - 2 \frac{d_A}{20} \right)^2 \right]$$
 < hollow portion of length x

$$\delta = \frac{P}{E} \left(\int \frac{1}{A(\zeta)} d\zeta \right) \qquad \qquad \delta = \frac{P}{E} \left[\int_0^{L-x} \frac{4}{\pi d(\zeta)^2} d\zeta + \int_{L-x}^L \frac{4}{\pi \left[d(\zeta)^2 - \left(d(\zeta) - 2\frac{d_A}{20} \right)^2 \right]} d\zeta \right]$$

$$\delta = \frac{P}{E} \left[\int_{0}^{L-x} \frac{4}{\pi \left[d_{A} \left(1 + \frac{\zeta}{L} \right) \right]} d\zeta + \int_{L-x}^{L} \frac{4}{\pi \left[\left[d_{A} \left(1 + \frac{\zeta}{L} \right) \right]^{2} - \left[d_{A} \left(1 + \frac{\zeta}{L} \right) - 2 \frac{d_{A}}{20} \right]^{2} \right]} d\zeta \right]$$

$$\delta = \frac{P}{E} \left[4 \frac{L^{2}}{(-2L + x)\pi d_{A}^{2}} + 4 \frac{L}{\pi d_{A}^{2}} + 20L \frac{\ln(3) + \ln(13) + 2\ln(d_{A}) + \ln(L)}{\pi d_{A}^{2}} - 20L \frac{2\ln(d_{A}) + \ln(39L - 20x)}{\pi d_{A}^{2}} \right]$$

if x = L/2

$$\delta = \frac{P}{E} \left(\frac{4}{3} \frac{L}{\pi d_A^2} + 20L \frac{\ln(3) + \ln(13) + 2\ln(d_A) + \ln(L)}{\pi d_A^2} - 20L \frac{2\ln(d_A) + \ln(29L)}{\pi d_A^2} \right)$$

Substitute numerical data:

$$\delta = 6.74 \text{ mm} \leftarrow$$

$$P_1 = 2.5 \text{kip}$$
 $P_2 = 1 \text{kip}$ $M = 25 \text{kip} \cdot \text{in}$ $E = 29000 \text{ksi}$ $A_1 = 0.25 \text{in}^2$ $A_2 = 0.15 \text{in}^2$

Find pin force at B - use FBD of bar BDE
$$\Sigma M_D = 0 \qquad B_y = \frac{1}{25 in} \left[P_2 \cdot (25 in) - M \right] = 0 \cdot kip$$

No pin force at B so bar ABC is subjected force P
$$_1$$
 at C only
$$\delta_C = \frac{P_1}{E} \cdot \left(\frac{20 in}{A_1} + \frac{35 in}{A_2} \right) = 0.027 \cdot in$$
 downward

Problem 2.3-28

Find pin force at B - use FBD of bar ABC
$$\Sigma M_A = 0 \qquad B_y = \frac{1}{\frac{L}{3}} \cdot (3 \cdot P \cdot L) \qquad B_y \rightarrow 9 \ P \qquad \text{acts upward on ABC so downward on DBF}$$

$$B_y \rightarrow 9 P$$

Vertical displacements at B and F

$$N_{BD} = P - 9 \cdot P \rightarrow -8 P$$

$$\delta_{\rm B} = \frac{N_{\rm BD} \cdot L}{2 \cdot {\rm EA}}$$

$${\rm N_{BD}} = {\rm P} - 9 \cdot {\rm P} \rightarrow -8 \ {\rm P} \qquad \quad \delta_{\rm B} = \frac{{\rm N_{BD}} \cdot {\rm L}}{2 \cdot {\rm EA}} \qquad \qquad \delta_{\rm B} \rightarrow -4 \ \frac{{\rm PL}}{{\rm EA}} \qquad {\rm downward}$$

$$\delta_{C} = \delta_{B} + \frac{N_{BF} \cdot \frac{L}{3}}{EA} \qquad \delta_{C} \rightarrow \frac{-11}{3} \frac{PL}{EA} \quad \text{downward}$$

$$C \rightarrow -\frac{11}{3} \frac{PL}{F\Delta}$$
 downward

Axial force (N(y)) and displacement (δ(y)) diagrams - origin of y at D, positive upward (rotated CW to horiz, position below)

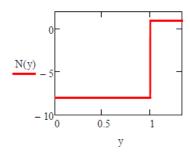
$$N(y) = \begin{bmatrix} N_{BD} & \text{if } y \leq L \\ N_{BF} & \text{otherwise} \end{bmatrix}$$

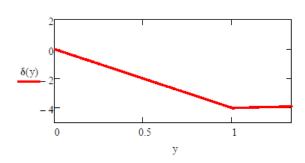
$$\begin{split} \delta(y) &= \left[\begin{bmatrix} N_{\mbox{\footnotesize{BD}}} \cdot y \cdot \left(\frac{L}{2 \cdot EA} \right) \end{bmatrix} \mbox{ if } y \leq L & \delta(0) \to 0 & \delta(L) \to -4 \\ \left[\delta_{\mbox{\footnotesize{B}}} + N_{\mbox{\footnotesize{BF}}} \cdot (y - L) \cdot \left(\frac{L}{3 \cdot EA} \right) \right] \mbox{ otherwise } & \delta\left(\frac{4 \cdot L}{3} \right) \to \frac{35}{9} = -3.889 \end{split}$$

$$\delta(0) \to 0 \qquad \delta(L) \to -4$$

$$\delta\left(\frac{4 \cdot L}{1}\right) \to -\frac{35}{1} = -3.889$$

times PL/EA





Find pin force at B - use FBD of bar ABC $\Sigma F_y = 0$ $B_y = 2P$ upward at B on ABC so downward on DBF

Axial forces in column segments (tension is positive)

$${\rm N}_{DB} = -{\rm P} \qquad \qquad {\rm N}_{BF} = -{\rm P} - {\rm B}_y \rightarrow -3 \cdot {\rm P} \qquad \text{ so AFD is constant and compressive over each column segment}$$

Vertical displacements at B and D (positive upward)

$$\delta_B = \frac{N_{BF} \cdot \frac{L}{2}}{2 \cdot EA} \rightarrow -\frac{3 \cdot L \cdot P}{4 \cdot EA} \qquad \qquad \delta_D = \delta_B + \frac{N_{DB} \cdot \frac{L}{2}}{EA} \rightarrow -\frac{5 \cdot L \cdot P}{4 \cdot EA} \qquad \text{so ADD is linear and downward over each column segment}$$

Use FBD of beam ABC - find pin force at B $\Sigma F_y = 0$ $B_y = 2 \cdot P$ upward on ABC so downward on DBF

$$\Sigma F_{*,*} = 0$$

$$B_v = 2 \cdot P$$

Axial forces in column segments (tension is positive)

$$N_{DB} = 0$$

$$N_{BF} = -B_v \rightarrow -2 \cdot P$$

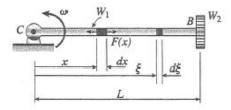
 N_{DB} = 0 N_{BF} = $-B_y \rightarrow -2 \cdot P$ so AFD is 0 over DB and constant and compressive over column segment BF

Vertical displacements at B and D (positive upward)

$$\delta_{\mathbf{B}} = \frac{N_{\mathbf{B}\mathbf{F}} \cdot \frac{\mathbf{L}}{2}}{2 \cdot \mathbf{E}\mathbf{A}} \rightarrow -\frac{\mathbf{L} \cdot \mathbf{P}}{2 \cdot \mathbf{E}\mathbf{A}}$$

$$\delta_{D} = \delta_{B} \rightarrow -\frac{L \cdot P}{2 \cdot EA}$$

 $\delta_B = \frac{N_{BF} \cdot \frac{L}{2}}{2 \cdot EA} \rightarrow -\frac{L \cdot P}{2 \cdot EA} \qquad \qquad \delta_D = \delta_B \rightarrow -\frac{L \cdot P}{2 \cdot EA} \qquad \qquad \text{so ADD is linear over BF and constant over column segment DB, both}$ downward



 ω = angular speed

A = cross-sectional area

E =modulus of elasticity

g = acceleration of gravity

F(x) =axial force in bar at distance x from point C

Consider an element of length dx at distance x from point C.

To find the force F(x) acting on this element, we must find the inertia force of the part of the bar from distance x to distance L, plus the inertia force of the weight W_2 .

Since the inertia force varies with distance from point C, we now must consider an element of length $d\xi$ at distance ξ , where ξ varies from x to L.

Mass of element
$$d = \frac{d}{L} \left(\frac{W_1}{g} \right)$$

Acceleration of element = $\xi \omega^2$

Centrifugal force produced by element

= (mass)(acceleration) =
$$\frac{W_1\omega^2}{gL}d$$

Centrifugal force produced by weight W2

$$=\left(\frac{W_2}{g}\right)(L\omega^2)$$

AXIAL FORCE F(x)

$$F(x) = \int_{-x}^{-L} \frac{W_1 \omega^2}{gL} d + \frac{W_2 L \omega^2}{g}$$
$$= \frac{W_1 \omega^2}{2gL} (L^2 - x^2) + \frac{W_2 L \omega^2}{g}$$

ELONGATION OF BAR BC

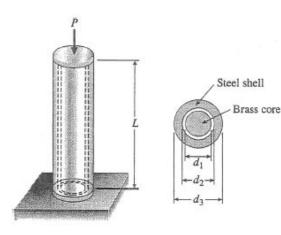
$$\delta = \int_{0}^{L} \frac{E(x) dx}{EA}$$

$$= \int_{0}^{L} \frac{W_{1}\omega^{2}}{2gL} (L^{2} - x^{2}) dx + \int_{0}^{L} \frac{W_{2}L\omega^{2}dx}{gEA}$$

$$= \frac{W_{1}L\omega^{2}}{2gLEA} \left[\int_{0}^{L} L^{2} dx - \int_{0}^{L} x^{2} dx \right] + \frac{W_{2}L\omega^{2}dx}{gEA} \int_{0}^{L} dx$$

$$= \frac{W_{1}L^{2}\omega^{2}}{3gEA} + \frac{W_{2}L^{2}\omega^{2}}{gEA}$$

$$= \frac{L^{2}\omega^{2}}{3gEA} + (W_{1} + 3W_{2}) \leftarrow$$



$$d_1 = 0.25$$
 in. $E_b = 15 \times 10^6$ psi $d_2 = 0.28$ in. $E_s = 30 \times 10^6$ psi $d_3 = 0.35$ in. $A_s = \frac{\pi}{4}(d_3^2 - d_2^2) = 0.03464$ in.²

$$L = 4.0 \text{ in.}$$
 $A_b = \frac{\pi}{4}d_1^2 = 0.04909 \text{ in.}^2$

(a) Decrease in length ($\delta = 0.003$ in.) Use Eq. (2-18) of Example 2-6.

$$\delta = \frac{PL}{E_s A_s + E_b A_b} \quad \text{or}$$

$$P = (E_s A_s + E_s A_b) \left(\frac{\delta}{L}\right)$$

Substitute numerical values:

$$E_s A_s + E_b A_b = (30 \times 10^6 \text{ psi})(0.03464 \text{ in.}^2)$$

+ $(15 \times 10^6 \text{ psi})(0.04909 \text{ in.}^2)$
= $1.776 \times 10^6 \text{ lb}$

$$P = (1.776 \times 10^6 \text{ lb}) \left(\frac{0.003 \text{ in.}}{4.0 \text{ in.}} \right)$$
$$= 1330 \text{ lb} \qquad \leftarrow$$

(b) ALLOWABLE LOAD

$$\sigma_s = 22 \text{ ksi}$$
 $\sigma_b = 16 \text{ ksi}$

Use Eqs. (2-17a and b) of Example 2-6. For steel:

$$\sigma_s = \frac{PE_s}{E_s A_s + E_b A_b} \quad P_s = (E_s A_s + E_b A_b) \frac{\sigma_s}{E_s}$$

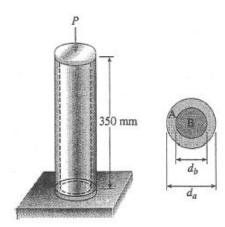
$$P_s = (1.776 \times 10^6 \text{ lb}) \left(\frac{22 \text{ ksi}}{30 \times 10^6 \text{ psi}} \right) = 1300 \text{ lb}$$

For brass:

$$\sigma_b = \frac{PE_b}{E_s A_s + E_b A_b} \quad P_s = (E_s A_s + E_b A_b) \frac{\sigma_b}{E_b}$$

$$P_s = (1.776 \times 10^6 \text{ lb}) \left(\frac{16 \text{ ksi}}{15 \times 10^6 \text{ psi}} \right) = 1890 \text{ lb}$$

Steel governs.
$$P_{\text{allow}} = 1300 \text{ lb}$$



A = aluminum

B = brass

L = 350 mm

 $d_a = 40 \text{ mm}$

 $d_b = 25 \text{ mm}$

$$A_a = \frac{\pi}{4} \left(d_a^2 - d_b^2 \right)$$

 $= 765.8 \text{ mm}^2$

$$E_a = 72 \text{ GPa}$$
 $E_b = 100 \text{ GPa}$ $A_b = \frac{\pi}{4} d_b^2$

(a) DECREASE IN LENGTH

$$(\delta = 0.1\% \text{ of } L = 0.350 \text{ mm})$$

Use Eq. (2-18) of Example 2-6.

$$\delta = \frac{PL}{E_a A_a + E_b A_b} \text{ or}$$

$$P = (E_a A_a + E_b A_b) \left(\frac{\delta}{L}\right)$$

Substitute numerical values:

$$E_a A_a + E_b A_b = (72 \text{ GPa})(765.8 \text{ mm}^2) + (100 \text{ GPa})(490.9 \text{ mm}^2)$$

$$= 55.135 \text{ MN} + 49.090 \text{ MN}$$

$$= 104.23 \text{ MN}$$

$$P = (104.23 \text{ MN}) \left(\frac{0.350 \text{ mm}}{350 \text{ mm}}\right)$$

(b) Allowable load

 $\sigma_a = 80 \text{ MPa}$ $\sigma_b = 120 \text{ MPa}$

Use Eqs. (2-17a and b) of Example 2-6.

For aluminum:

$$\sigma_a = \frac{PE_a}{E_a A_a + E_b A_b}$$
 $P_a = (E_a A_a + E_b A_b) \left(\frac{\sigma_a}{E_a}\right)$
 $P_a = (104.23 \text{ MN}) \left(\frac{80 \text{ MPa}}{72 \text{ GPa}}\right) = 115.8 \text{ kN}$

For brace

$$\sigma_b = \frac{PE_b}{E_a A_a + E_b A_b}$$
 $P_b = (E_a A_a + E_b A_b) \left(\frac{\sigma_b}{E_b}\right)$
 $P_b = (104.23 \text{ MN}) \left(\frac{120 \text{ MPa}}{100 \text{ GPa}}\right) = 125.1 \text{ kN}$

Aluminum governs. $P_{\text{max}} = 116 \text{ kN} \leftarrow$

Problem 2.4-3

$$E = 29000 \text{ksi}$$
 $A = 8 \text{in}^2$

Use superposition - select A_v as the redundant

Released structure with actual load P at C
$$\delta_{A1} = \frac{2.5 kip \cdot (6ft)}{E \cdot A} = 7.759 \times 10^{-4} \cdot in \quad \text{upward}$$

$$\text{Released structure with redundant A}_{y} \text{ applied at A} \qquad \delta_{A2} = A_{y} \cdot \left(\frac{3 \, \text{ft} + 6 \, \text{ft}}{E \cdot A} \right) \qquad \frac{3 \, \text{ft} + 6 \, \text{ft}}{E \cdot A} = 4.655 \times \ 10^{-4} \cdot \frac{\text{in}}{\text{kip}}$$

Compatibility equation
$$\delta_{A1} + \delta_{A2} = 0$$
 solve for redundant A_y $A_y = \frac{-\delta_{A1}}{\frac{3ft + 6ft}{F_{c}\Delta}} = -1.667 \cdot kip$

Statics
$$B_y = -\left(A_y + 2.5 \text{kip}\right) = -0.833 \cdot \text{kip}$$
 Axial displacement at C
$$\frac{-B_y \cdot (6 \text{ft})}{E \cdot A} = 2.586 \times 10^{-4} \cdot \text{in} \quad \text{upward ... or} \qquad \frac{-A_y \cdot (3 \, \text{ft})}{E \cdot A} = 2.586 \times 10^{-4} \cdot \text{in}$$

use either extension of segment BC or compression of AC to find upward displ. δ_{C}

$$P = 10kN$$

$$E = 200GPa$$

$$\sigma_V = 400 MPa$$

$$FS_Y = 2$$

E = 200GPa
$$\sigma_{\mathbf{Y}}$$
 = 400MPa $FS_{\mathbf{Y}}$ = 2 $\sigma_{\mathbf{a}}$ = $\frac{\sigma_{\mathbf{Y}}}{FS_{\mathbf{Y}}}$ = 200·MPa

Static equilibrium - cut through cables, use lower FBD (see fig.)

$$a = 1.5m$$

$$b = 1.5m$$

$$a = 1.5m$$
 $b = 1.5m$ $\alpha_B = atan\left(\frac{a}{b}\right) = 45 \cdot deg$

$$\alpha_{\rm C} = \operatorname{atan}\left(\frac{a}{2 \cdot b}\right) = 26.565 \cdot \deg$$

 $\Sigma M_A = 0$

$$T_1 \cdot \sin(\alpha_B) + 2 \cdot T_2 \cdot \sin(\alpha_C) = P \cdot (2)$$

Compatibility - from figure, see that $\Delta_C = 2 \cdot \Delta_B$

Cable elongations

$$\delta_1 = \Delta_B \cdot \sin(\alpha_B)$$

$$\delta_1 = \Delta_B \cdot \sin(\alpha_B)$$

$$\delta_2 = \Delta_C \cdot \sin(\alpha_C)$$

so
$$\delta_2 = 2 \cdot \left(\frac{\sin(\alpha_C)}{\sin(\alpha_B)} \right) \cdot \delta_1$$
 $2 \cdot \left(\frac{\sin(\alpha_C)}{\sin(\alpha_B)} \right) = 1.26491$

$$2 \cdot \left(\frac{\sin(\alpha_{\rm C})}{\sin(\alpha_{\rm B})} \right) = 1.26491$$



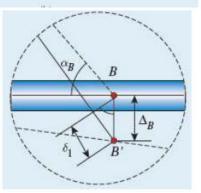
$$L_1 = \sqrt{a^2 + b^2} = 2.121 \text{ m}$$

$$L_1 = \sqrt{a^2 + b^2} = 2.121 \text{ m}$$
 $L_2 = \sqrt{a^2 + (2 \cdot b)^2} = 3.354 \text{ m}$

$$\delta_1 = \mathbf{T_1} \cdot \mathbf{f_1}$$
 $f_1 = \frac{\mathbf{L_1}}{\mathbf{F_1 A_2}}$
 $\delta_2 = \mathbf{T_2} \cdot \mathbf{f_2}$
 $f_2 = \frac{\mathbf{L_2}}{\mathbf{F_1 A_2}}$

$$f_1 = \frac{L_1}{E \cdot A_1}$$

$$\delta_2 = \mathbf{T}_2 \cdot \mathbf{f}_2$$



$$T_2 \cdot f_2 = 2 \cdot \left(\frac{\sin(\alpha_C)}{\sin(\alpha_B)} \right) \cdot T_1 \cdot f_1$$

where
$$\mathbf{T_2 \cdot f_2} = 2 \cdot \left(\frac{\sin(\alpha_C)}{\sin(\alpha_B)}\right) \cdot \mathbf{T_1 \cdot f_1}$$
 or $\mathbf{T_2} = 2 \cdot \left(\frac{\sin(\alpha_C)}{\sin(\alpha_B)}\right) \cdot \left(\frac{\mathbf{f_1}}{\mathbf{f_2}}\right) \cdot \mathbf{T_1}$ and $\mathbf{A_1} = \mathbf{A_2}$ so $\frac{\mathbf{f_1}}{\mathbf{f_2}} = \frac{\mathbf{L_1}}{\mathbf{L_2}}$

$$\frac{\mathbf{f_1}}{\mathbf{f_2}} = \frac{\mathbf{L_1}}{\mathbf{L_2}}$$

Substitute T2 expression into equilibrium equation and solve for T1 then solve for T2

$$T_{1} = \left(\frac{2 \cdot f_{2} \cdot \sin(\alpha_{B})}{f_{2} \cdot \sin(\alpha_{B})^{2} + 4 \cdot f_{1} \cdot \sin(\alpha_{C})^{2}}\right) \cdot P \quad \text{or}$$

$$T_1 = \left(\frac{2 \cdot f_2 \cdot \sin(\alpha_B)}{f_2 \cdot \sin(\alpha_B)^2 + 4 \cdot f_1 \cdot \sin(\alpha_C)^2}\right) \cdot P \qquad \text{or} \qquad T_1 = \left[\frac{2 \cdot \sin(\alpha_B)}{\sin(\alpha_B)^2 + 4 \cdot \left(\frac{L_1}{L_2}\right) \cdot \sin(\alpha_C)^2}\right] \cdot P = 14.058 \cdot kN$$

$$T_2 = 2 \cdot \left(\frac{\sin(\alpha_C)}{\sin(\alpha_B)} \right) \cdot \left(\frac{L_1}{L_2} \right) \cdot T_1 = 11.247 \cdot \text{kN}$$

Use allowable stress σ_a to find minimum required cross sectional area of each cable

$$A_1 = \frac{T_1}{\sigma_0} = 70.291 \cdot mm^2$$
 $A_2 = \frac{T_2}{\sigma_0} = 56.233 \cdot mm^2$ so $A_{reqd} = 70.3 mm^2$

$$A_2 = \frac{T_2}{\sigma_2} = 56.233 \cdot \text{mm}^2$$

$$A_{reqd} = 70.3 \text{mm}^2$$

$$\sigma_{ys} = 50 \text{ksi}$$
 $\sigma_{yA} = 60 \text{ksi}$ $A_s = 12 \text{in}^2$ $A_A = 6 \text{in}^2$ $L = 20 \text{in}$ $E_s = 29000 \text{ksi}$ $E_A = 10600 \text{ksi}$

Axial stiffnesses of cylinder and tube - treat as springs in parallel

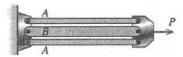
$$k_{S} = \frac{E_{S} \cdot A_{S}}{L} = 1.74 \times 10^{4} \cdot \frac{kip}{in} \qquad k_{A} = \frac{E_{A} \cdot A_{A}}{L} = 3.18 \times 10^{3} \cdot \frac{kip}{in} \qquad k_{T} = k_{S} + k_{A} = 2.058 \times 10^{4} \cdot \frac{kip}{in}$$

Each "spring" carries a force in proportion to its stiffness

$$\mathbf{P_{S}(P)} = \frac{k_{S}}{k_{T}} \cdot \mathbf{P} \; \mathbf{float}, \\ 5 \; \rightarrow 0.84548 \cdot \mathbf{P} \\ \qquad \qquad \mathbf{P_{A}(P)} = \frac{k_{A}}{k_{T}} \cdot \mathbf{P} \; \mathbf{float}, \\ 5 \; \rightarrow 0.15452 \cdot \mathbf{P} \\ \qquad \qquad \mathbf{P_{A}(P)} = \frac{k_{A}}{k_{T}} \cdot \mathbf{P} \; \mathbf{float}, \\ 6 \; \rightarrow 0.15452 \cdot \mathbf{P} \\ \qquad \qquad \mathbf{P_{A}(P)} = \frac{k_{A}}{k_{T}} \cdot \mathbf{P} \; \mathbf{float}, \\ 6 \; \rightarrow 0.15452 \cdot \mathbf{P} \\ \qquad \qquad \mathbf{P_{A}(P)} = \frac{k_{A}}{k_{T}} \cdot \mathbf{P} \; \mathbf{float}, \\ 6 \; \rightarrow 0.15452 \cdot \mathbf{P} \\ \qquad \qquad \mathbf{P_{A}(P)} = \frac{k_{A}}{k_{T}} \cdot \mathbf{P} \; \mathbf{float}, \\ 6 \; \rightarrow 0.15452 \cdot \mathbf{P} \\ \qquad \qquad \mathbf{P_{A}(P)} = \frac{k_{A}}{k_{T}} \cdot \mathbf{P} \; \mathbf{float}, \\ 6 \; \rightarrow 0.15452 \cdot \mathbf{P} \\ \qquad \qquad \mathbf{P_{A}(P)} = \frac{k_{A}}{k_{T}} \cdot \mathbf{P} \; \mathbf{P$$

Maximum force in each component is governed by its yield stress

So the allowable load P is limited by yield stress in steel cylinder $P_{a11} = 709.655 \text{kip}$



(1)

(4)

(5)

FREE-BODY DIAGRAM OF END PLATE



EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{horiz}} = 0$$
 $P_A + P_B - P = 0$

EQUATION OF COMPATIBILITY

$$\delta_A = \delta_B$$
 (2)

FORCE-DISPLACEMENT RELATIONS

 A_A = total area of both outer bars

$$\delta_A = \frac{P_A L}{E_A A_k} \quad \delta_B = \frac{P_B L}{E_R A_R}$$
 (3)

Substitute into Eq. (2):

$$\frac{P_A L}{E_A A_A} = \frac{P_B L}{E_B A_B}$$

SOLUTION OF THE EQUATIONS

Solve simultaneously Eqs. (1) and (4):

$$P_A = \frac{E_A A_A P}{E_A A_A + E_B A_B} \quad P_B = \frac{E_B A_B P}{E_A A_A + E_B A_B}$$

Substitute into Eq. (3):

$$\delta = \delta_A = \delta_B = \frac{PL}{E_A A_A + E_B A_B} \tag{6}$$

STRESSES:

$$\sigma_A = \frac{P_A}{A_A} = \frac{E_A P}{E_A A_A + E_B A_B}$$

$$\sigma_B = \frac{P_B}{A_B} = \frac{E_B P}{E_A A_A + E_B A_B}$$
(7)

(a) LOAD IN MIDDLE BAR

$$\frac{P_B}{P} = \frac{E_B A_B}{E_A A_A + E_B A_B} = \frac{1}{\frac{E_A A_A}{E_B A_B} + 1}$$

Given:
$$\frac{E_A}{E_B} = 2$$
 $\frac{A_A}{A_B} = \frac{1+1}{1.5} = \frac{4}{3}$

$$\therefore \frac{P_B}{P} = \frac{1}{\left(\frac{E_A}{E_B}\right)\left(\frac{A_A}{A_B}\right) + 1} = \frac{1}{\frac{8}{3} + 1} = \frac{3}{11} \quad \longleftarrow$$

(b) RATIO OF STRESSES

$$\frac{\sigma_B}{\sigma_A} = \frac{E_B}{E_A} = \frac{1}{2}$$

(c) RATIO OF STRAINS

All bars have the same strain

(a) Reactions at A and B due to load P at L/2

$$A_{AC} = \frac{\pi}{4} \left[d^2 - \left(\frac{d}{2} \right)^2 \right] \qquad A_{AC} = \frac{3}{16} \pi d^2$$

$$A_{CB} = \frac{\pi}{4} d^2$$

Select R_B as the redundant; use superposition and a compatibility equation at B:

if
$$x \le L/2$$
 $\delta_{B1a} = \frac{Px}{EA_{AC}} + \frac{P\left(\frac{L}{2} - x\right)}{EA_{CB}}$ $\delta_{B1a} = \frac{P}{E} \left(\frac{x}{\frac{3}{16}\pi d^2} + \frac{\frac{L}{2} - x}{\frac{\pi}{4}d^2}\right)$

$$\delta_{B1a} = \frac{2}{3}P \frac{2x + 3L}{E\pi d^2}$$

if
$$x \ge L/2$$
 $\delta_{B1b} = \frac{P\frac{L}{2}}{EA_{AC}}$ $\delta_{B1b} = \frac{P\frac{L}{2}}{E\left(\frac{3}{16}\pi d^2\right)}$ $\delta_{B1b} = \frac{8}{3}\frac{PL}{E\pi d^2}$

The following expression for δ_{B2} is good for all x:

$$\delta_{B2} = \frac{R_B}{E} \left(\frac{x}{A_{AC}} + \frac{L - x}{A_{CB}} \right)$$

$$\delta_{B2} = \frac{R_B}{E} \left(\frac{x}{\frac{3}{16} \pi d^2} + \frac{L - x}{\frac{\pi}{4} d^2} \right)$$

$$\delta_{B2} = \frac{R_B}{E} \left(\frac{16}{3} \frac{x}{\pi d^2} + 4 \frac{L - x}{\pi d^2} \right)$$

Solve for R_B and R_A assuming that $x \le L/2$:

Compatibility:
$$\delta_{B1a} + \delta_{B2} = 0$$
 $R_{Ba} = \frac{-\left(\frac{2}{3}P\frac{2x+3L}{\pi d^2}\right)}{\left(\frac{16}{3}\frac{x}{\pi d^2} + 4\frac{L-x}{\pi d^2}\right)}$ $R_{Ba} = \frac{-1}{2}P\frac{2x+3L}{x+3L} \leftarrow$

^ check—if
$$x = 0$$
, $R_B = -P/2$

Statics:
$$R_{Aa} = -P - R_{Ba}$$
 $R_{Aa} = -P - \frac{-1}{2}P\frac{2x + 3L}{x + 3L}$ $R_{Aa} = \frac{-3}{2}P\frac{L}{x + 3L}$ \leftarrow $^{\circ}$ check—if $x = 0$, $R_{Aa} = -P/2$

Solve for R_B and R_A assuming that $x \ge L/2$:

Compatibility:
$$\delta_{B1b} + \delta_{B2} = 0 \qquad R_{Bb} = \frac{\frac{-8}{3} \frac{PL}{\pi d^2}}{\left(\frac{16}{3} \frac{x}{\pi d^2} + 4 \frac{L - x}{\pi d^2}\right)} \qquad R_{Bb} = \frac{-2PL}{x + 3L} \quad \longleftarrow$$

^ check—if x = L, $R_R = -P/2$

Statics:
$$R_{Ab} = -P - R_{Bb}$$
 $R_{Ab} = -P - \left(\frac{-2PL}{x+3L}\right)$ $R_{Ab} = -P \frac{x+L}{x+3L}$ \leftarrow

(b) Find δ at point of load application; axial force for segment 0 to $L/2 = -R_A$ and $\delta =$ elongation of this segment Assume that $x \le L/2$:

$$\delta_{a} = \frac{-R_{Aa}}{E} \left(\frac{x}{A_{AC}} + \frac{\frac{L}{2} - x}{A_{CB}} \right) \qquad \delta_{a} = \frac{-\left(\frac{-3}{2} P \frac{L}{x + 3L} \right)}{E} \left(\frac{x}{\frac{3}{16} \pi d^{2}} + \frac{\frac{L}{2} - x}{\frac{\pi}{4} d^{2}} \right)$$

$$\delta_a = PL \frac{2x + 3L}{(x + 3L)E\pi d^2}$$

For
$$x = L/2$$
, $\delta_a = \frac{8}{7} L \frac{P}{E\pi d^2} \leftarrow$

Assume that $x \ge L/2$:

$$\delta_b = \frac{(-R_{Ab})\frac{L}{2}}{EA_{AC}} \qquad \delta_b = \frac{\left(P\frac{x+L}{x+3L}\right)\frac{L}{2}}{E\left(\frac{3}{16}\pi d^2\right)} \qquad \delta_b = \frac{8}{3}P\left(\frac{x+L}{x+3L}\right)\frac{L}{E\pi d^2} \quad \longleftarrow$$

for
$$x = L/2$$
 $\delta_b = \frac{8}{7} P \frac{L}{E\pi d^2}$ < same as δ_a above (OK)

(c) For what value of x is $R_B = (6/5) R_A$? Guess that x < L/2 here and use R_{Ba} expression above to find x:

$$\frac{-1}{2}P\frac{2x+3L}{x+3L} - \frac{6}{5}\left(\frac{-3}{2}P\frac{L}{x+3L}\right) = 0 \qquad \frac{-1}{10}P\frac{10x-3L}{x+3L} = 0 \qquad x = \frac{3L}{10} \quad \longleftarrow$$

Now try $R_{Bb} = (6/5)R_{Ab}$, assuming that x > L/2

$$\frac{-2PL}{x+3L} - \frac{6}{5} \left(-P \frac{x+L}{x+3L} \right) = 0 \qquad \frac{2}{5} P \frac{-2L+3x}{x+3L} = 0 \qquad x = \frac{2}{3} L \quad \longleftarrow$$

So, there are two solutions for x.

(d) Find reactions if the bar is now rotated to a vertical position, load P is removed, and the bar is hanging under its own weight (assume mass density = ρ). Assume that x = L/2.

$$A_{AC} = \frac{3}{16} \pi d^2$$
 $A_{CB} = \frac{\pi}{4} d^2$

Select R_B as the redundant; use superposition and a compatibility equation at B

from (a) above. compatibility: $\delta_{B1} + \delta_{B2} = 0$

$$\delta_{B2} = \frac{R_B}{E} \left(\frac{x}{A_{AC}} + \frac{L - x}{A_{CR}} \right)$$
 For $x = L/2$, $\delta_{B2} = \frac{R_B}{E} \left(\frac{14}{3} \frac{L}{\pi d^2} \right)$

$$\delta_{B1} = \int_0^{\frac{L}{2}} \frac{N_{AC}}{EA_{AC}} d\zeta + \int_{\frac{L}{2}}^{L} \frac{N_{CB}}{EA_{CB}} d\zeta$$

Where axial forces in bar due to self weight are $W_{AC} = \rho g A_{AC} \frac{L}{2}$ $W_{CB} = \rho g A_{CB} \frac{L}{2}$ (assume ζ is measured upward from A):

$$N_{AC} = -\left[\rho g A_{CB} \frac{L}{2} + \rho g A_{AC} \left(\frac{L}{2} - \zeta\right)\right] \qquad A_{AC} = \frac{3}{16} \pi d^2 \qquad A_{CB} = \frac{\pi}{4} d^2$$

$$N_{CB} = -[\rho g A_{CB}(L - \zeta)]$$

$$N_{AC} = \frac{-1}{8} \rho g \pi \, d^2 \, L \, - \frac{3}{16} \rho g \pi \, d^2 \left(\frac{1}{2} L \, - \, \zeta \right) \qquad N_{CB} = \, - \left[\frac{1}{4} \rho g \pi \, d^2 (\, L \, - \, \zeta) \, \right]$$

$$\delta_{B1} = \int_{0}^{\frac{L}{2}} \frac{-1}{8} \rho g \pi d^{2}L - \frac{3}{16} \rho g \pi d^{2} \left(\frac{1}{2}L - \zeta\right)}{E\left(\frac{3}{16} \pi d^{2}\right)} d\zeta + \int_{\frac{L}{2}}^{L} \frac{-\left[\frac{1}{4} \rho g \pi d^{2} (L - \zeta)\right]}{E\left(\frac{\pi}{4} d^{2}\right)} d\zeta$$

$$\delta_{B1} = \left(\frac{-11}{24}\rho g \frac{L^2}{E} + \frac{-1}{8}\rho g \frac{L^2}{E}\right) \qquad \delta_{B1} = \frac{-7}{12}\rho g \frac{L^2}{E} \qquad \frac{7}{12} = 0.583$$

Compatibility: $\delta_{B1} + \delta_{B2} = 0$

$$R_B = \frac{-\left(\frac{-7}{12}\rho g \frac{L^2}{E}\right)}{\left(\frac{14}{3} \frac{L}{E\pi d^2}\right)} \qquad R_B = \frac{1}{8}\rho g \pi d^2 L \quad \longleftarrow$$

Statics:
$$R_A = (W_{AC} + W_{CB}) - R_B$$

$$R_{A} = \left[\left[\rho g \left(\frac{3}{16} \pi d^{2} \right) \frac{L}{2} + \rho g \left(\frac{\pi}{4} d^{2} \right) \frac{L}{2} \right] - \frac{1}{8} \rho g \pi d^{2} L \right]$$

$$R_A = \frac{3}{32} \rho g \pi d^2 L \qquad \longleftarrow$$

Problem 2.4-8

$$P = 200kN$$
 $L = 2m$ $t = 20mm$ $b_1 = 100mm$ $b_2 = 115mm$ $E = 96GPa$

$$\delta_{C1} = \frac{P \cdot \left(\frac{3 \cdot L}{5}\right)}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot \ln \left(\frac{b_2}{b_1}\right) = 1.165 \cdot mm$$

axial displacement at C due to redundant
$$R_{\text{C}}$$

$$\delta_{C2} = \mathbf{R_{C}} \cdot \left[\frac{\frac{3 \cdot \mathbf{L}}{5} + \frac{2 \cdot \mathbf{L}}{5}}{\mathbf{E} \cdot \mathbf{t} \cdot \left(\mathbf{b}_{2} - \mathbf{b}_{1}\right)} \cdot \ln \left(\frac{\mathbf{b}_{2}}{\mathbf{b}_{1}}\right) \right]$$

$$\frac{\frac{3 \cdot L}{5} + \frac{2 \cdot L}{5}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot \ln \left(\frac{b_2}{b_1}\right) = 9.706 \times 10^{-3} \cdot \frac{mm}{kN}$$

$$\delta_{C1} + \delta_{C2} = 0$$
 solve for R₀

$$\begin{split} & \delta_{C1} + \delta_{C2} = 0 \qquad \text{solve for R}_{C} \qquad R_{C} = \frac{-\delta_{C1}}{\left[\frac{3 \cdot L}{5} + \frac{2 \cdot L}{5} \cdot \ln\left(\frac{b_{2}}{b_{1}}\right)\right]} = -120 \cdot kN \end{split}$$

Statics
$$\Sigma F = 0$$
 $R_A = -(P + R_C) = -80 \cdot kN$

Negative reactions so both act to left

Compute extension of AB or compression of BC to find displ. δ_{B} (to the right)

$$-R_{A} \cdot \left[\frac{\left(\frac{3 \cdot L}{5}\right)}{E \cdot t \cdot \left(b_{2} - b_{1}\right)} \cdot \ln \left(\frac{b_{2}}{b_{1}}\right) \right] = 0.466 \cdot mm \qquad \text{or} \qquad -R_{C} \cdot \left[\frac{\left(\frac{2 \cdot L}{5}\right)}{E \cdot t \cdot \left(b_{2} - b_{1}\right)} \cdot \ln \left(\frac{b_{2}}{b_{1}}\right) \right] = 0.466 \cdot mm$$

CLICK HERE TO ACCESS THE COMPLETE Solutions

Problem 2.4-9

P = 20kip L = 3ft t =
$$\frac{1}{4}$$
in b₁ = 2in b₂ = 2.5in E = 10400ksi
A_{BC} = b₁·t = 0.5·in² b_{ave} = $\frac{b_1 + b_2}{2}$ = 2.25·in

Select reaction R_{C} as the redundant; use superposition

axial displacement at C due to actual load P at middle of AB

$$\delta_{C1} = \frac{P \cdot \left(\frac{L}{4}\right)}{E \cdot t \cdot \left(b_2 - b_{ave}\right)} \cdot \ln \left(\frac{b_2}{b_{ave}}\right) = 0.029 \cdot in$$

axial displacement at C due to redundant R_C

$$\delta_{C2} = R_{C} \cdot \left[\frac{\frac{L}{2}}{E \cdot t \cdot (b_2 - b_1)} \cdot ln \left(\frac{b_2}{b_1} \right) + \frac{\frac{L}{2}}{E \cdot A_{BC}} \right]$$

flexibility constant for bar

$$\frac{\frac{L}{2}}{\text{E.t.}(b_2 - b_1)} \cdot \ln\left(\frac{b_2}{b_1}\right) + \frac{\frac{L}{2}}{\text{E.A}_{BC}} = 6.551 \times 10^{-3} \cdot \frac{\text{in}}{\text{kip}}$$

Compatibility equation

$$\delta_{C1} + \delta_{C2} = 0$$
 solve for R₀

$$\begin{split} & \delta_{\text{C1}} + \delta_{\text{C2}} = 0 \qquad \text{solve for R}_{\text{C}} \quad \text{R}_{\text{C}} = \frac{-\delta_{\text{C1}}}{\left[\frac{\underline{L}}{2}}{\frac{\underline{L}}{E \cdot t \cdot (b_2 - b_1)} \cdot \ln \left(\frac{b_2}{b_1}\right) + \frac{\underline{L}}{2}}\right]} = -4.454 \cdot \text{kip} \end{split}$$

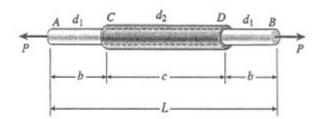
Statics
$$\Sigma F = 0$$
 $R_A = -(P + R_C) = -15.546 \cdot kip$

Negative reactions so both act to left

Compute deformations of AB (two terms, more difficult) or deformation of BC (easier) to find displ. δ_B (to the right)

$$-R_{A} \cdot \left[\frac{\left(\frac{L}{4}\right)}{E \cdot t \cdot \left(b_{2} - b_{ave}\right)} \cdot ln \left(\frac{b_{2}}{b_{ave}}\right) \right] + \frac{\left(-R_{A} - P\right) \cdot \frac{L}{4}}{E \cdot t \cdot \left(b_{ave} - b_{1}\right)} \cdot ln \left(\frac{b_{ave}}{b_{1}}\right) = 1.542 \times 10^{-2} \cdot in$$

or
$$-R_{C} \cdot \left(\frac{\frac{L}{2}}{E \cdot A_{BC}} \right) = 1.542 \times 10^{-2} \cdot in$$



$$P = 12 \text{ kN}$$

$$d_1 = 30 \text{ mm}$$

$$P = 12 \text{ kN}$$
 $d_1 = 30 \text{ mm}$ $b = 100 \text{ mm}$

$$L = 500 \text{ mm}$$
 $d_2 = 45 \text{ mm}$ $c = 300 \text{ mm}$

$$= 45 \, \text{mm}$$

$$c = 300 \, \text{mn}$$

Rod: $E_1 = 3.1 \text{ GPa}$

Sleeve: $E_2 = 2.5 \text{ GPa}$

Rod:
$$A_1 = \frac{\pi d_1^2}{4} = 706.86 \text{ mm}^2$$

Sleeve:
$$A_2 = \frac{\pi}{4}(d_2^2 - d_1^2) = 883.57 \text{ mm}^2$$

$$E_1A_1 + E_2A_2 = 4.400 \text{ MN}$$

(a) ELONGATION OF ROD

Part
$$AC$$
: $\delta_{AC} = \frac{Pb}{E_1A_1} = 0.5476 \text{ mm}$

Part *CD*:
$$\delta_{CD} = \frac{Pc}{E_1 A_1 + E_2 A_2}$$

= 0.81815 mm

(From Eq. 2-16 of Example 2-8)

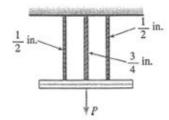
$$\delta = 2\delta_{AC} + \delta_{CD} = 1.91 \text{ mm}$$

(b) SLEEVE AT FULL LENGTH

$$\delta = \delta_{CD} \left(\frac{L}{c} \right) = (0.81815 \text{ mm}) \left(\frac{500 \text{ mm}}{300 \text{ mm}} \right)$$
$$= 1.36 \text{ mm} \qquad \leftarrow$$

(c) SLEEVE REMOVED

$$\delta = \frac{PL}{E_1 A_1} = 2.74 \text{ mm} \quad \leftarrow$$



AREAS OF CABLES (from Table 2-1)

Middle cable: $A_M = 0.268 \text{ in.}^2$

Outer cables: $A_O = 0.119 \text{ in.}^2$

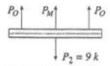
(for each cable)

FIRST LOADING

$$P_1 = 12 \text{ k} \left(\text{Each cable carries } \frac{P_1}{3} \text{ or } 4 \text{ k.} \right)$$

SECOND LOADING

 $P_2 = 9 \text{ k (additional load)}$



EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{vert}} = 0$$
 $2P_O + P_M - P_2 = 0$ (1)

EQUATION OF COMPATIBILITY

$$\delta_M = \delta_O$$
 (2)

FORCE-DISPLACEMENT RELATIONS

$$\delta_M = \frac{P_M L}{E A_M} \quad \delta_O = \frac{P_o L}{E A_o} \tag{3,4}$$

SUBSTITUTE INTO COMPATIBILITY EQUATION:

$$\frac{P_M L}{E A_M} = \frac{P_O L}{E A_O} \frac{P_M}{A_M} = \frac{P_O}{A_O}$$
 (5)

SOLVE SIMULTANEOUSLY EQS. (1) AND (5):

$$P_M = P_2 \left(\frac{A_M}{A_M + 2A_O} \right) = (9 \text{ k}) \left(\frac{0.268 \text{ in.}^2}{0.506 \text{ in.}^2} \right)$$

= 4.767 k

$$P_o = P_2 \left(\frac{A_o}{A_M + 2A_O} \right) = (9 \text{ k}) \left(\frac{0.119 \text{ in.}^2}{0.506 \text{ in.}^2} \right)$$

= 2.117 k

FORCES IN CABLES

Middle cable: Force = 4 k + 4.767 k = 8.767 kOuter cables: Force = 4 k + 2.117 k = 6.117 k(for each cable)

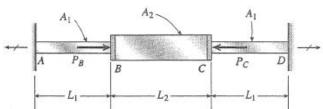
(a) PERCENT OF TOTAL LOAD CARRIED BY MIDDLE CABLE

Percent =
$$\frac{8.767 \text{ k}}{21 \text{ k}} (100\%) = 41.7\%$$

(b) Stresses in Cables ($\sigma = P/A$)

Middle cable:
$$\sigma_M = \frac{8.767 \text{ k}}{0.268 \text{ in.}^2} = 32.7 \text{ ksi} \leftarrow$$

Outer cables:
$$\sigma_O = \frac{6.117 \text{ k}}{0.119 \text{ in.}^2} = 51.4 \text{ ksi} \quad \leftarrow$$



FREE-BODY DIAGRAM



EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{horiz}} = 0 \rightarrow \leftarrow$$

$$P_B + R_D - P_C - R_A = 0 \text{ or}$$

$$R_A - R_D = P_B - P_C = 8.5 \text{ kN}$$
 (Eq. 1)

EQUATION OF COMPATIBILITY

 δ_{AD} = elongation of entire bar

$$\delta_{AD} = \delta_{AB} + \delta_{BC} + \delta_{CD} = 0$$
 (Eq. 2)

FORCE-DISPLACEMENT RELATIONS

$$\delta_{AB} = \frac{R_A L_1}{E A_1} = \frac{R_A}{E} \left(238.05 \frac{1}{m} \right)$$
 (Eq. 3)

$$\delta_{BC} = \frac{(R_A - P_B)L_2}{EA_2}$$

$$= \frac{R_A}{E} \left(198.413 \frac{1}{m} \right) - \frac{P_B}{E} \left(198.413 \frac{1}{m} \right)$$
 (Eq. 4)

$$\delta_{CD} = \frac{R_D L_1}{E A_1} = \frac{R_D}{E} \left(238.095 \frac{1}{m} \right)$$
 (Eq. 5)

$$P_B = 25.5 \text{ kN}$$
 $P_C = 17.0 \text{ kN}$
 $L_1 = 200 \text{ mm}$ $L_2 = 250 \text{ mm}$
 $A_1 = 840 \text{ mm}^2$ $A_2 = 1260 \text{ mm}^2$
 $m = \text{meter}$

SOLUTION OF EQUATIONS

Substitute Eqs. (3), (4), and (5) into Eq. (2):

$$\frac{R_A}{E} \left(238.095 \frac{1}{\text{m}} \right) + \frac{R_A}{E} \left(198.413 \frac{1}{\text{m}} \right)$$
$$-\frac{P_B}{E} \left(198.413 \frac{1}{\text{m}} \right) + \frac{R_D}{E} \left(238.095 \frac{1}{\text{m}} \right) = 0$$

Simplify and substitute $P_B = 25.5 \text{ kN}$:

$$R_A \left(436.508 \frac{1}{\text{m}}\right) + R_D \left(238.095 \frac{1}{\text{m}}\right)$$

= 5,059.53 kN/m (Eq. 6)

(a) REACTIONS R_A AND R_D

Solve simultaneously Eqs. (1) and (6).

From (1):
$$R_D = R_A - 8.5 \text{ kN}$$

Substitute into (6) and solve for R_A :

$$R_A \left(674.603 \frac{1}{\text{m}} \right) = 7083.34 \text{ kN/m}$$

$$R_A = 10.5 \text{ kN} \leftarrow$$

 $R_D = R_A - 8.5 \text{ kN} = 2.0 \text{ kN} \leftarrow$

(b) Compressive axial force F_{BC}

$$F_{BC} = P_B - R_A = P_C - R_D = 15.0 \text{ kN}$$

NUMERICAL DATA

$$n = 6$$
 $d_b = 0.5 \text{ in.}$ $\sigma_a = 14 \text{ ksi}$ $A_b = \frac{\pi}{4} d_b^2 = 0.196 \text{ in.}^2$

(a) Formulas for reactions F

Segment ABC flexibility:
$$f_1 = \frac{2\left(\frac{L}{4}\right)}{EA} = \frac{L}{2EA}$$

Segment *CDE* flexibility:
$$f_2 = \frac{2\left(\frac{L}{4}\right)}{\frac{1}{2}EA} = \frac{L}{EA}$$

Loads at points B and D:

$$P_B = -2P$$
 $P_D = 3P$

(1) Select R_E as the redundant; find axial displacement δ_1 = displacement at E due to loads P_B and P_D :

$$\delta_1 = \frac{(P_B + P_D)\frac{L}{4}}{EA} + \frac{P_D\frac{L}{4}}{EA} + \frac{P_D\frac{L}{4}}{\frac{1}{2}EA} = \frac{5LP}{2EA}$$

(2) Next apply redundant R_E and find axial displacement δ_2 = displacement at E due to redundant R_E :

$$\delta_2 = R_E(f_1 + f_2) = \frac{3LR_E}{2EA}$$

(3) Use compatibility equation to find redundant R_E then use statics to find R_A:

$$\delta_1 + \delta_2 = 0$$
 solve, $R_E = -\frac{5P}{3}$ $R_E = \frac{-5}{3}P$

$$R_A = -R_E - P_B - P_D = \frac{2P}{3}$$
 $R_A = \frac{2P}{3}$ $R_A = \frac{2P}{3}$ $R_A = \frac{2P}{3}$

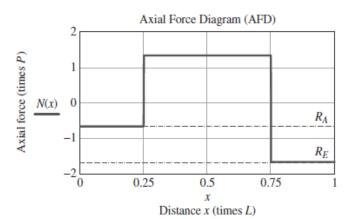
(b) Determine the axial displacements δ_B , δ_C , and δ_D at points B, C, and D, respectively.

$$\delta_B \frac{\left(\frac{-2P}{3}\right)\left(\frac{L}{4}\right)}{EA} = \frac{LP}{6EA} \qquad \delta_c = \delta_B + \frac{\left(2P - \frac{2P}{3}\right)\left(\frac{L}{4}\right)}{EA} = \frac{LP}{6EA} \qquad \delta_D = \frac{\left(\frac{5P}{3}\right)\left(\frac{L}{4}\right)}{\frac{EA}{2}} = \frac{5LP}{6EA}$$
to the right
to the right

(c) Draw an axial-displacement diagram (ADD) in which the abscissa is the distance x from support A to any point on the bar and the ordinate is the horizontal displacement δ at that point.

AFD for use below in Part (d)

AFD is composed of 4 constant segments, so ADD is linear with zero displacements at supports A and E.

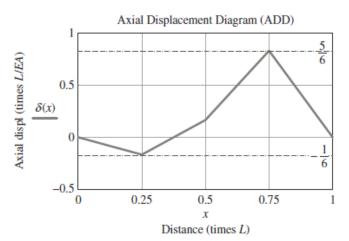


Plot displacements δ_B , δ_C , and δ_D from part (b) above, then connect points using straight lines showing linear variation of axial displacement Between points

$$\delta_{\text{max}} = \delta_D$$
 $\delta_{\text{max}} = \frac{5LP}{6EA}$ to the right

Boundary conditions at supports:

$$\delta_A = \delta_E = 0$$



(d) Maximum permissible value of load variable P based on allowable normal stress in flange bolts From AFD, force at L/2:

$$F_{\text{max}} = \frac{4}{3}P$$
 and $F_{\text{max}} = n\sigma_a A_b = 16.493 \text{ k}$

$$P_{\text{max}} = \frac{3}{4} F_{\text{max}} = 12.37 \text{ k}$$
 $P_{\text{max}} = 12.37 \text{ k}$

CLICK HERE TO ACCESS THE COMPLETE Solutions

Problem 2.4-14

(a) Stresses and reactions: Select R_1 as redundant and do superposition analysis (here q=0; deflection positive upward)

$$d_1 = 50 \text{ mm}$$
 $d_2 = 60 \text{ mm}$ $d_3 = 57 \text{ mm}$ $d_4 = 64 \text{ mm}$ $A_1 = \frac{\pi}{4} (d_2^2 - d_1^2) = 863.938 \text{ mm}^2$
 $E = 110 \text{ MPa}$ $A_2 = \frac{\pi}{4} (d_4^2 - d_3^2) = 665.232 \text{ mm}^2$

Segment flexibilities $L_1 = 2 \text{ m}$ $L_2 = 3 \text{ m}$

$$f_1 = \frac{L_1}{EA_1} = 0.02105 \text{ mm/N}$$
 $f_2 = \frac{L_2}{EA_2} = 0.041 \text{ mm/N}$ $\frac{f_1}{f_2} = 0.513$

Tensile stress (σ_1) is known in upper segment so $R_1 = \sigma_1 \times A_1$ $\sigma_1 = 10.5$ MPa $R_1 = \sigma_1 A_1 = 9.07$ kN $\delta_{1a} = -Pf_2$ $\delta_{1b} = R_1(f_1 + f_2)$ Compatibility: $\delta_{1a} + \delta_{1b} = 0$

Solve for *P*:
$$P = R_1 \left(\frac{f_1 + f_2}{f_2} \right) = 13.73 \text{ kN}$$

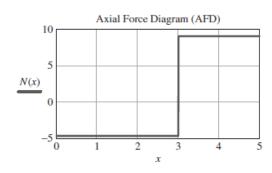
Finally, use statics to find R_2 : $R_2 = P - R_1 = 4.66 \text{ kN}$ $\sigma_2 = \frac{R_2}{A_2} = 7 \text{ MPa}$ < compressive since R_2 is positive (upward)

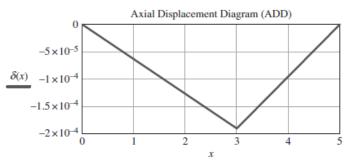
$$P = 13.73 \text{ kN}$$
 $R_1 = 9.07 \text{ kN}$ $R_2 = 4.66 \text{ kN}$ $\sigma_2 = 7 \text{ MPa}$

(b) DISPLACEMENT AT CAP PLATE

$$\delta_c = R_1 f_1 = 190.909 \text{ mm}$$
 < downward OR $\delta_c = (R_2) f_2 = 190.909 \text{ mm}$ < downward (neg. x-direction)
 $\delta_{\text{cap}} = \delta_c = 0.191 \text{ m}$ $\delta_{\text{cap}} = 190.9 \text{ mm}$ AFD and ADD: $R_1 = 9.071$ $R_2 = 4.657$ $L_1 = 2$ $A_1 = 863.938$ $A_2 = 665.232$ $E = 110$ $L_2 = 3$

NOTE: x is measured up from lower support.



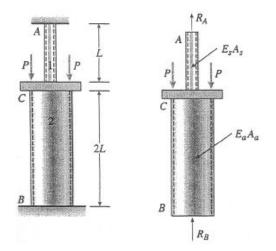


(c) Uniform load Q on segment 2 such that $R_2=0$

$$P = 13.728 \text{ kN}$$
 $R_1 = \sigma_1 A_1 = 9.071 \text{ kN}$ $L_2 = 3 \text{ m}$

Equilibrium: $R_1 + R_2 = P - qL_2 < \text{set } R_2 = 0$, solve for req'd $q = \frac{P - R_1}{L_2} = 1.552 \text{ kN/m}$

q = 1.552 kN/m



Pipe 1 is steel. Pipe 2 is aluminum.

EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{vert}} = 0 \qquad R_A + R_B = 2P \tag{Eq. 1}$$

EQUATION OF COMPATIBILITY

$$\delta_{AB} = \delta_{AC} + \delta_{CB} = 0 \tag{Eq. 2}$$

(A positive value of δ means elongation.)

FORCE-DISPLACEMENT RELATIONS

$$\delta_{AC} = \frac{R_A L}{E_s A_s} \quad \delta_{BC} = -\frac{R_B (2L)}{E_o A_o}$$
 (Eqs. 3, 4))

SOLUTION OF EQUATIONS

Substitute Eqs. (3) and (4) into Eq. (2):

$$\frac{R_A L}{E_s A_s} - \frac{R_B (2L)}{E_a A_a} = 0$$
 (Eq. 5)

Solve simultaneously Eqs. (1) and (5):

$$R_A = \frac{4E_s A_s P}{E_a A_a + 2E_s A_s} \quad R_B = \frac{2E_a A_a P}{E_a A_a + 2E_s A_s}$$
 (Eqs. 6, 7)

(a) AXIAL STRESSES

Aluminum:
$$\sigma_a = \frac{R_B}{A_a} = \frac{2E_aP}{E_aA_a + 2E_sA_s}$$
 (compression) (Eq. 8)

Steel:
$$\sigma_s = \frac{R_A}{A_s} = \frac{4E_sP}{E_aA_a + 2E_sA_s} \leftarrow \text{(Eq. 9)}$$

(b) Numerical results

$$P = 12 \text{ k}$$
 $A_a = 8.92 \text{ in.}^2$ $A_s = 1.03 \text{ in.}^2$
 $E_a = 10 \times 10^6 \text{ psi}$ $E_s = 29 \times 10^6 \text{ psi}$
Substitute into Eqs. (8) and (9):
 $\sigma_a = 1,610 \text{ psi}$ (compression) \leftarrow

$$\sigma_s = 9,350 \text{ psi (tension)} \leftarrow$$

Numerical data:

$$W = 800 \text{ N}$$
 $L = 150 \text{ mm}$ $a = 50 \text{ mm}$ $d_S = 2 \text{ mm}$

$$d_A = 4 \text{ mm}$$
 $E_S = 210 \text{ GPa}$

$$E_A = 70 \text{ GPa}$$

$$\sigma_{Sa} = 220 \text{ MPa}$$
 $\sigma_{Aa} = 80 \text{ MPa}$

$$A_A = \frac{\pi}{4} d_A^2 \qquad A_S = \frac{\pi}{4} d_S^2$$

$$A_A = 13 \text{ mm}^2$$
 $A_S = 3 \text{ mm}^2$

(a) $P_{\rm allow}$ AT CENTER OF BAR

One-degree statically indeterminate - use reaction (R_A) at top of aluminum bar as the redundant compatibility: $\delta_1 - \delta_2 = 0$ Statics: $2R_S + R_A = P + W$

$$\delta_1 = \frac{P + W}{2} \left(\frac{L}{E_S A_S} \right)$$
 < downward displacement due to elongation of each steel wire under $P + W$ if aluminum wire is cut at top

$$\delta_2 = R_A \left(\frac{L}{2E_S A_S} + \frac{L}{E_A A_A} \right)$$
 < upward displ. due to shortening of steel wires and elongation of aluminum wire under redundant R_A

Enforce compatibility and then solve for R_A :

$$\delta_1 = \delta_2 \quad \text{so} \quad R_A = \frac{\frac{P+W}{2} \left(\frac{L}{E_S A_S}\right)}{\frac{L}{2E_S A_S} + \frac{L}{E_A A_A}} \quad R_A = (P+W) \frac{E_A A_A}{E_A A_A + 2E_S A_S} \quad \text{and} \quad \sigma_{Aa} = \frac{R_A}{A_A}$$

Now use statics to find R_S :

$$R_S = \frac{P+W-R_A}{2} \qquad R_S = \frac{P+W-(P+W)\frac{E_AA_A}{E_AA_A+2E_SA_S}}{2} \qquad R_S = (P+W)\frac{E_SA_S}{E_AA_A+2E_SA_S}$$
 and
$$\sigma_{Sa} = \frac{R_S}{A_S}$$

Compute stresses and apply allowable stress values:

$$\sigma_{Aa} = (P + W) \frac{E_A}{E_A A_A + 2E_S A_S} \qquad \sigma_{Sa} = (P + W) \frac{E_S}{E_A A_A + 2E_S A_S}$$

Solve for allowable load P:

$$P_{Aa} = \sigma_{Aa} \left(\frac{E_A A_A + 2E_S A_S}{E_A} \right) - W$$
 $P_{Sa} = \sigma_{Sa} \left(\frac{E_A A_A + 2E_S A_S}{E_S} \right) - W$ (lower value of P controls)
 $P_{Aa} = 1713 \text{ N}$ $P_{Sa} = 1504 \text{ N} \leftarrow P_{\text{allow}}$ is controlled by steel wires

(b) P_{allow} if load P at x = a/2

Again, cut aluminum wire at top, then compute elongations of left and right steel wires:

$$\delta_{1L} = \left(\frac{3P}{4} + \frac{W}{2}\right)\left(\frac{L}{E_S A_S}\right) \quad \delta_{1R} = \left(\frac{P}{4} + \frac{W}{2}\right)\left(\frac{L}{E_S A_S}\right)$$

$$\delta_1 = \frac{\delta_{1L} + \delta_{1R}}{2}$$
 $\delta_1 = \frac{P + W}{2} \left(\frac{L}{E_S A_S}\right)$ where $\delta_1 =$ displacement at $x = a$

Use δ_2 from part (a):

$$\delta_2 = R_A \left(\frac{L}{2E_S A_S} + \frac{L}{E_A A_A} \right)$$

So equating
$$\delta_1$$
 and δ_2 , solve for R_A : $R_A = (P + W) \frac{E_A A_A}{E_A A_A + 2E_S A_S}$

^ same as in part (a)

$$R_{SL} = \frac{3P}{4} + \frac{W}{2} - \frac{R_A}{2}$$
 < stress in left steel wire exceeds that in right steel wire

$$R_{SL} = \frac{3P}{4} + \frac{W}{2} - \frac{(P+W)\frac{E_A A_A}{E_A A_A + 2E_S A_S}}{2}$$

$$R_{SL} = \frac{PE_{A}A_{A} + 6PE_{S}A_{S} + 4WE_{S}A_{S}}{4E_{A}A_{A} + 8E_{S}A_{S}} \qquad \sigma_{Sa} = \frac{PE_{A}A_{A} + 6PE_{S}A_{S} + 4WE_{S}A_{S}}{4E_{A}A_{A} + 8E_{S}A_{S}} \left(\frac{1}{A_{S}}\right)$$

Solve for $P_{\rm allow}$ based on allowable stresses in steel and aluminum:

$$P_{Sa} = \frac{\sigma_{Sa}(4A_SE_AA_A + 8E_SA_S^2) - (4WE_SA_S)}{E_AA_A + 6E_SA_S} \qquad P_{Aa} = 1713 \text{ N} \qquad < \text{same as in part(a)}$$

$$P_{Sa} = 820 \text{ N} \qquad \leftarrow \text{ steel controls}$$

(c) P_{allow} if wires are switched as shown and x = a/2

Select R_A as the redundant; statics on the two released structures:

 Cut aluminum wire—apply P and W, compute forces in left and right steel wires, then compute displacements at each steel wire:

$$R_{SL} = \frac{P}{2} \qquad R_{SR} = \frac{P}{2} + W$$

$$\delta_{1L} = \frac{P}{2} \left(\frac{L}{E_S A_S} \right) \quad \delta_{1R} = \left(\frac{P}{2} + W \right) \left(\frac{L}{E_S A_S} \right)$$

By geometry, δ at aluminum wire location at far right is $\delta_1 = \left(\frac{P}{2} + 2W\right)\left(\frac{L}{E_S A_S}\right)$

(2) Next apply redundant R_A at right wire, compute wire force and displacement at aluminum wire:

$$R_{SL} = -R_A$$
 $R_{SR} = 2R_A$ $\delta_2 = R_A \left(\frac{5L}{E_S A_S} + \frac{L}{E_A A_A} \right)$

(3) Compatibility equate δ₁, δ₂ and solve for R_A, then P_{allow} for aluminum wire:

$$R_{A} = \frac{\left(\frac{P}{2} + 2W\right)\left(\frac{L}{E_{S}A_{S}}\right)}{\frac{5L}{E_{S}A_{S}} + \frac{L}{E_{A}A_{A}}} \qquad R_{A} = \frac{E_{A}A_{A}P + 4E_{A}A_{A}W}{10E_{A}A_{A} + 2E_{S}A_{S}} \qquad \sigma_{Aa} = \frac{R_{A}}{A_{A}}$$

$$\sigma_{Aa} = \frac{E_{A}P + 4E_{A}W}{10E_{A}A_{A} + 2E_{S}A_{S}}$$

$$P_{Aa} = \frac{\sigma_{Aa}(10E_{A}A_{A} + 2E_{S}A_{S}) - 4E_{A}W}{E_{A}} \qquad P_{Aa} = 1713 \text{ N}$$

(4) Statics or superposition—find forces in steel wires, then P_{allow} for steel wires:

$$R_{SL} = \frac{P}{2} + R_A \qquad R_{SL} = \frac{P}{2} + \frac{E_A A_A P + 4 E_A A_A W}{10 E_A A_A + 2 E_S A_S}$$

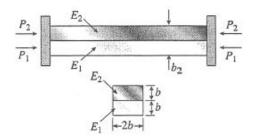
$$R_{SL} = \frac{6 E_A A_A P + P E_S A_S + 4 E_A A_A W}{10 E_A A_A + 2 E_S A_S} \qquad < \text{larger than } R_{SR}, \text{ so use in allowable stress calculations}$$

$$R_{SR} = \frac{P}{2} + W - 2R_{A} \qquad R_{SR} = \frac{P}{2} + W - \frac{E_{A}A_{A}P + 4E_{A}A_{A}W}{5E_{A}A_{A} + E_{S}A_{S}}$$

$$R_{SR} = \frac{3E_{A}A_{A}P + PE_{S}A_{S} + 2E_{A}A_{A}W + 2WE_{S}A_{S}}{10E_{A}A_{A} + 2E_{S}A_{S}}$$

$$\sigma_{Sa} = \frac{R_{SL}}{A_{S}} \qquad P_{Sa} = \sigma_{Sa}A_{S} \left(\frac{10E_{A}A_{A} + 2E_{S}A_{S}}{6E_{A}A_{A} + E_{S}A_{S}}\right) - \frac{4E_{A}A_{A}W}{6E_{A}A_{A} + E_{S}A_{S}}$$

$$P_{Sa} = \frac{10\sigma_{Sa}A_{S}E_{A}A_{A} + 2\sigma_{Sa}A_{S}^{2}E_{S} - 4E_{A}A_{A}W}{6E_{A}A_{A} + E_{S}A_{S}} \qquad P_{Sa} = 703 \text{ N} \leftarrow 6E_{A}A_{A} + E_{S}A_{S}$$



FREE-BODY DIAGRAM

(Plate at right-hand end)

$$\begin{array}{c|c}
\frac{b}{2} & P_2 & P & \downarrow e \\
\hline
\uparrow & \uparrow & P_1 & \uparrow & \uparrow e
\end{array}$$

EQUATIONS OF EQUILIBRIUM

$$\Sigma F = 0 \quad P_1 + P_2 = P$$
 (Eq. 1)

$$\Sigma M = 0 \Leftrightarrow P_{\ell} = P_{\ell} \left(\frac{b}{2}\right) - P_{\ell} \left(\frac{b}{2}\right) = 0 \quad \text{(Eq. 2)}$$

EQUATION OF COMPATIBILITY

$$\delta_2 = \delta_1$$

$$\frac{P_2L}{E_2A} = \frac{P_1L}{E_1A}$$
 or $\frac{P_2}{E_2} = \frac{P_1}{E_1}$ (Eq. 3)

(a) AXIAL FORCES

Solve simultaneously Eqs. (1) and (3):

$$P_1 = \frac{PE_1}{E_1 + E_2}$$
 $P_2 = \frac{PE_2}{E_1 + E_2}$ \leftarrow

(b ECCENTRICITY OF LOAD P Substitute P₁ and P₂ into Eq. (2) and solve for e:

$$e = \frac{b(E_2 - E_1)}{2(E_2 + E_1)} \quad \leftarrow \quad$$

(c) RATIO OF STRESSES

$$\sigma_1 = \frac{P_1}{A}$$
 $\sigma_2 = \frac{P_2}{A}$ $\frac{\sigma_1}{\sigma_2} = \frac{P_1}{P_2} = \frac{E_1}{E_2}$ \leftarrow

NUMERICAL DATA

$$L = 2.5 \text{ m}$$
 $b = 0.71$ $L = 1.775 \text{ m}$ $E = 210 \text{ GPa}$ $A = 3500 \text{ mm}^2$ $P = 185 \text{ kN}$ $\theta_A = 60^\circ$ $\sigma_a = 150 \text{ MPa}$

FIND MISSING DIMENSIONS AND ANGLES IN PLANE TRUSS FIGURE

$$x_{c} = b\cos(\theta_{A}) = 0.8875 \text{ m}$$
 $y_{c} = b\sin(\theta_{A}) = 1.5372 \text{ m}$
$$\frac{b}{\sin(\theta_{B})} = \frac{L}{\sin(\theta_{A})} \quad \text{so} \quad \theta_{B} = a\sin\left(\frac{b\sin(\theta_{A})}{L}\right) = 37.94306^{\circ}$$

$$\theta_{C} = 180^{\circ} - (\theta_{A} + \theta_{B}) = 82.05694^{\circ}$$

$$c = \frac{L}{\sin(\theta_{A})}\sin(\theta_{C}) = 2.85906 \text{ m} \quad \text{or} \quad c = \sqrt{b^{2} + L^{2} - 2bL\cos(\theta_{C})} = 2.85906 \text{ m}$$

(a) Select B_x as the redundant; perform superposition analysis to find B_x then use statics to find remaining reactions. Finally use method of joints to find member forces (see Example 1-1)

 δ_{Bx1} = displacement in x-direction in released structure acted upon by loads P and 2P at joint C:

 $\delta_{Bx1} = 1.2789911 \text{ mm}$ < this displacement equals force in AB divided by flexibility of AB

 δ_{Bx2} = displacement in x-direction in released structure acted upon by redundant B_x : $\delta_{BX2} = B_x \frac{c}{EA}$

Compatibility equation: $\delta_{BX1} + \delta_{BX2} = 0$ so $B_X = \frac{-EA}{c} \delta_{BX1} = -328.8 \text{ kN}$

STATICS: $\Sigma F_X = 0$ $A_X = -B_X - 2P = -41.2 \text{ kN}$ $\Sigma M_A = 0$ $B_y = \frac{1}{c} [2P(b\sin(\theta_A)) + P(b\cos(\theta_A))] = 256.361 \text{ kN}$ $\Sigma F_y = 0$ $A_y = P - B_y = -71.361 \text{ kN}$

REACTIONS:

$$A_x = -41.2 \text{ kN}$$
 $A_y = -71.4 \text{ kN}$ $B_x = -329 \text{ kN}$ $B_y = 256 \text{ kN}$

- (b) Find maximum permissible value of load variable P if allowable normal stress is 150 MPa
 - Use reactions and Method of Joints to find member forces in each member for above loading.

Results: $F_{AB} = 0$ $F_{BC} = -416.929 \text{ kN}$ $F_{AC} = 82.40 \text{ kN}$

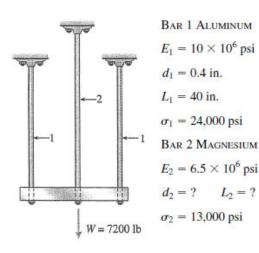
(2) Compute member stresses:

$$\sigma_{AB} = 0$$
 $\sigma_{BC} = \frac{-416.93 \text{ kN}}{A} = -119.123 \text{ MPa}$ $\sigma_{AC} = \frac{82.4 \text{ kN}}{A} = 23.543 \text{ MPa}$

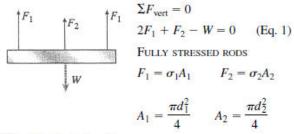
(3) Maximum stress occurs in member BC. For linear analysis, the stress is proportional to the load so

$P_{\text{max}} =$	$\frac{\sigma_a}{\sigma_{BC}}$	P = 233 kN	
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So when downward load P = 233 kN is applied at C and horizontal load 2P = 466 kN is applied to the right at C, the stress in BC is 150 MPa



FREE-BODY DIAGRAM OF RIGID BAR EQUATION OF EQUILIBRIUM



Substitute into Eq. (1):

$$2\sigma_1\left(\frac{\pi d_1^2}{4}\right) + \sigma_2\left(\frac{\pi d_2^2}{4}\right) = W$$

Diameter d_1 is known; solve for d_2 :

$$d_2^2 = \frac{4W}{\pi \sigma_2} - \frac{2\sigma_1 d_1^2}{\sigma_2} \quad \longleftarrow \quad \text{(Eq. 2)}$$

SUBSTITUTE NUMERICAL VALUES:

$$d_2^2 = \frac{4(7200 \text{ lb})}{\pi (13,000 \text{ psi})} - \frac{2(24,000 \text{ psi})(0.4 \text{ in.})^2}{13,000 \text{ psi}}$$
$$= 0.70518 \text{ in.}^2 - 0.59077 \text{ in.}^2 = 0.11441 \text{ in.}^2$$
$$d_2 = 0.338 \text{ in.} \qquad \longleftarrow$$

EQUATION OF COMPATIBILITY

$$\delta_1 = \delta_2$$
 (Eq. 3)

FORCE-DISPLACEMENT RELATIONS

$$\delta_1 = \frac{F_1 L_1}{E_1 A_1} = \sigma_1 \left(\frac{L_1}{E_1}\right)$$
 (Eq. 4)

$$\delta_2 = \frac{F_2 L_2}{E_2 A_2} = \sigma_2 \left(\frac{L_2}{E_2}\right)$$
 (Eq. 5)

Substitute (4) and (5) into Eq. (3):

$$\sigma_1\left(\frac{L_1}{E_1}\right) = \sigma_2\left(\frac{L_2}{E_2}\right)$$

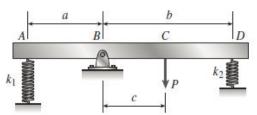
Length L_1 is known; solve for L_2 :

$$L_2 = L_1 \left(\frac{\sigma_1 E_2}{\sigma_2 E_1} \right) \qquad \longleftarrow \tag{Eq. 6}$$

SUBSTITUTE NUMERICAL VALUES:

$$L_2 = (40 \text{ in.}) \left(\frac{24,000 \text{ psi}}{13,000 \text{ psi}} \right) \left(\frac{6.5 \times 10^6 \text{ psi}}{10 \times 10^6 \text{ psi}} \right)$$

= 48.0 in.



NUMERICAL DATA

a = 250 mm

b = 500 mm

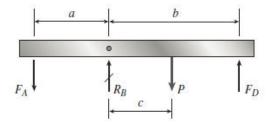
c = 200 mm

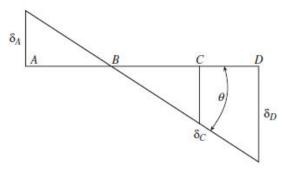
 $k_1 = 10 \text{ kN/m}$

 $k_2 = 25 \text{ kN/m}$

$$\theta_{\text{max}} = 3^{\circ} = \frac{\pi}{60} \text{ rad}$$

FREE-BODY DIAGRAM AND DISPLACEMENT DIAGRAM





EQUATION OF EQUILIBRIUM

$$\sum M_B = 0 + -F_A(a) - P(c) + F_D(b) = 0$$
 (Eq. 1)

EQUATION OF COMPATIBILITY

$$\frac{\delta_A}{a} = \frac{\delta_D}{b}$$
 (Eq. 2)

FORCE-DISPLACEMENT RELATIONS

$$\delta_A = \frac{F_A}{k_1} \quad \delta_D = \frac{F_D}{k_2} \tag{Eqs. 3, 4}$$

SOLUTION OF EQUATIONS

Substitute (3) and (4) into Eq. (2):

$$\frac{F_A}{ak_1} = \frac{F_D}{bk_2} \tag{Eq. 5}$$

Solve simultaneously Eqs. (1) and (5):

$$F_A = \frac{ack_1P}{a^2k_1 + b^2k_2}$$
 $F_D = \frac{bck_2P}{a^2k_1 + b^2k_2}$

ANGLE OF ROTATION

$$\delta_D = \frac{F_D}{k_2} = \frac{bcP}{a^2k_1 + b^2k_2} \qquad \theta = \frac{\delta_D}{b} = \frac{cP}{a^2k_1 + b^2k_2}$$

MAXIMUM LOAD

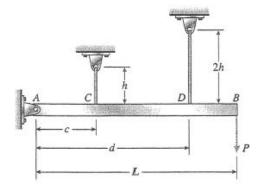
$$P = \frac{\theta}{c} (a^2 k_1 + b^2 k_2)$$

$$P_{\text{max}} = \frac{\theta_{\text{max}}}{c} (a^2 k_1 + b^2 k_2) \quad \leftarrow$$

SUBSTITUTE NUMERICAL VALUES:

$$P_{\text{max}} = \frac{\pi/60 \text{ rad}}{200 \text{ mm}} [(250 \text{ mm})^2 (10 \text{ kN/m}) + (500 \text{ mm})^2 (25 \text{ kN/m})]$$

$$= 1800 \text{ N} \quad \leftarrow$$



h = 18 in.

2h = 36 in.

c = 20 in.

d = 50 in.

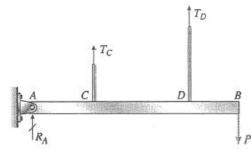
L = 66 in.

 $E = 30 \times 10^{6} \, \text{psi}$

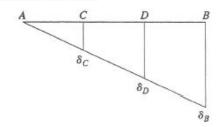
 $A = 0.0272 \text{ in.}^2$

P = 340 lb

FREE-BODY DIAGRAM



DISPLACEMENT DIAGRAM



EQUATION OF EQUILIBRIUM

$$\sum M_A = 0 \Leftrightarrow T_C(c) + T_D(d) = PL$$
 (Eq. 1)

EQUATION OF COMPATIBILITY

$$\frac{\delta_c}{c} = \frac{\delta_D}{d}$$
 (Eq. 2)

FORCE-DISPLACEMENT RELATIONS

$$\delta_C = \frac{T_C h}{FA} \quad \delta_D = \frac{T_D(2h)}{FA}$$
 (Eqs. 3, 4)

SOLUTION OF EQUATIONS

Substitute (3) and (4) into Eq. (2):

$$\frac{T_C h}{cEA} = \frac{T_D(2h)}{dEA} \quad \text{or} \quad \frac{T_C}{c} = \frac{2T_D}{d}$$
 (Eq. 5)

TENSILE FORCES IN THE WIRES

Solve simultaneously Eqs. (1) and (5):

$$T_C = \frac{2cPL}{2c^2 + d^2}$$
 $T_D = \frac{dPL}{2c^2 + d^2}$ (Eqs. 6, 7)

TENSILE STRESSES IN THE WIRES

$$\sigma_C = \frac{T_C}{A} = \frac{2cPL}{A(2c^2 + d^2)}$$
 (Eq. 8)

$$\sigma_D = \frac{T_D}{A} = \frac{dPL}{A(2c^2 + d^2)}$$
 (Eq. 9)

DISPLACEMENT AT END OF BAR

$$\delta_B = \delta_D \left(\frac{L}{d}\right) = \frac{2hT_D}{EA} \left(\frac{L}{d}\right) = \frac{2hPL^2}{EA(2c^2 + d^2)}$$
 (Eq. 10)

SUBSTITUTE NUMERICAL VALUES

$$2c^2 + d^2 = 2(20 \text{ in.})^2 + (50 \text{ in.})^2 = 3300 \text{ in.}^2$$

(a)
$$\sigma_C = \frac{2cPL}{A(2c^2 + d^2)} = \frac{2(20 \text{ in.})(340 \text{ lb})(66 \text{ in.})}{(0.0272 \text{ in.}^2)(3300 \text{ in.}^2)}$$

= 10,000 psi \leftarrow

$$\sigma_D = \frac{dPL}{A(2c^2 + d^2)} = \frac{(50 \text{ in.})(340 \text{ lb})(66 \text{ in.})}{(0.0272 \text{ in.}^2)(3300 \text{ in.}^2)}$$
$$= 12,500 \text{ psi} \qquad \leftarrow$$

(b)
$$\delta_B = \frac{2hPL^2}{EA(2c^2 + d^2)}$$

$$= \frac{2(18 \text{ in.})(340 \text{ lb})(66 \text{ in.})^2}{(30 \times 10^6 \text{ psi})(0.0272 \text{ in.}^2)(3300 \text{ in.}^2)}$$

= 0.0198 in.

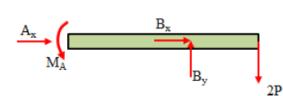
Remove pin at B; draw separate FBD's of beam and column. Find selected forces using statics

From FBD of column DBF

$$\Sigma M_B = D_x \cdot \frac{L}{2} = 0$$
 $D_x = 0$

$$\Sigma F_X = D_X - B_X = 0$$
 $B_X = D_X$

$$B_x = D_x$$



From FBD of beam ABC

$$\Sigma F_{\mathbf{v}} = \mathbf{A}_{\mathbf{v}} + \mathbf{B}_{\mathbf{v}} = 0 \qquad A_{\mathbf{v}} = 0$$

$$A_x = 0$$

$$\Sigma M_{\mathbf{B}} = \mathbf{M_A} - 2\mathbf{P} \cdot \frac{\mathbf{L}}{3} = 0$$
 $M_{\mathbf{A}} = 2\mathbf{P} \cdot \frac{\mathbf{L}}{3}$

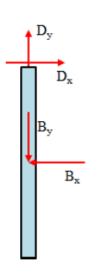
$$M_A = 2P \cdot \frac{L}{3}$$

$$\Sigma F_{V} = B_{V} - 2P = 0$$
 $B_{V} = 2P$

$$B_v = 2P$$

Remove reaction R_F to create the release structure; find vertical displacement at F due to actual load 2P at C

$$\delta_{\text{F1}} = \frac{\text{B}_{\text{y}} \cdot \frac{\text{L}}{2}}{\text{EA}} \qquad \delta_{\text{F1}} = \frac{\text{P} \cdot \text{L}}{\text{EA}}$$



Apply redundant R_F to released structure; find vertical displacement at F

$$B_{y} = 0 \qquad \delta'_{F2} = \frac{-R_{F} \cdot \frac{L}{2}}{2EA} - \frac{R_{F} \cdot \frac{L}{2}}{EA} \qquad \delta'_{F2} = -R_{F} \cdot \left(\frac{L}{4 \cdot EA} + \frac{L}{2EA}\right) \qquad \delta'_{F2} = -R_{F} \cdot \frac{3L}{4 \cdot EA}$$

$$\delta'_{F2} = -R_F \left(\frac{L}{4 \cdot EA} + \frac{L}{2EA} \right)$$

$$\delta'_{F2} = -R_F \cdot \frac{3L}{4 \cdot EA}$$

Compatibility equation - solve for R_F

$$\delta_{F1} + \delta'_{F2} = 0$$

$$\delta_{F1} + \delta'_{F2} = 0$$

$$R_F = \frac{\frac{P \cdot L}{EA}}{\left(\frac{3L}{4 \cdot FA}\right)} \qquad R_F = \frac{4}{3}P$$

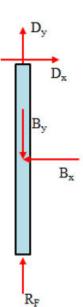
Finally solve for reaction $\mathbf{D}_{\mathbf{y}}$ using FBD of DBF

$$\Sigma F_v = 0$$

$$D_v = B_v - R_t$$

$$\Sigma F_y = 0$$
 $D_y = B_y - R_F$ $D_y = 2P - \frac{4}{3}P$ $D_y = \frac{2}{3}P$

$$D_y = \frac{2}{3}P$$



 B_x

Numerical properties (kips, inches):

$$d_c = 2.25 \text{ in.}$$
 $d_b = 1.75 \text{ in.}$ $d_s = 1.25 \text{ in.}$

$$A_s = \frac{\pi}{4} d_s^2$$

$$E_c = 18,000 \text{ ksi}$$
 $E_b = 16,000 \text{ ksi}$

$$A_b = \frac{\pi}{4} (d_b^2 - d_s^2)$$

$$E_s = 30000 \text{ ksi}$$

$$A_c = \frac{\pi}{4} \left(d_c^2 - d_b^2 \right)$$

$$P = 9 \text{ k}$$

EQUATION OF EQUILIBRIUM

$$\sum F_{\text{vert}} = 0 \qquad P_s + P_b + P_c = P \tag{Eq. 1}$$

EQUATIONS OF COMPATIBILITY

$$\delta_s = \delta_b$$
 $\delta_c = \delta_s$ (Eqs. 2)

FORCE-DISPLACEMENT RELATIONS

$$\delta_s = \frac{P_s L}{E_s A_s} \delta_b = \frac{P_b L}{E_b A_b} \delta_c = \frac{P_c L}{E_c A_c}$$
 (Eqs. 3, 4, 5)

SOLUTION OF EQUATIONS

Substitute (3), (4), and (5) into Eqs. (2):

$$P_b = P_s \frac{E_b A_b}{E_s A_s} \quad P_c = P_s \frac{E_c A_c}{E_s A_s}$$
 (Eqs. 6, 7)

SOLVE SIMULTANEOUSLY EQS. (1), (6), AND (7):

$$P_s = P \frac{E_s A_s}{E_s A_s + E_b A_b + E_c A_c} = 3.95 \text{ k}$$

$$P_b = P \frac{E_b A_b}{E_s A_s + E_b A_b + E_c A_c} = 2.02 \text{ k}$$

$$P_c = P \frac{E_c A_c}{E_s A_s + E_b A_b + E_c A_c} = 3.03 \text{ k}$$

$$P_s + P_b + P_c = 9$$
 statics check

Compressive stresses

Let
$$\Sigma EA = E_s A_s + E_b A_b + E_c A_c$$

$$\sigma_s = \frac{P_s}{A_s} = \frac{PE_s}{\Sigma EA}$$
 $\sigma_s = 3.22 \text{ ksi}$ \leftarrow

$$\sigma_b = \frac{P_b}{A_b} = \frac{PE_b}{\Sigma EA}$$
 $\sigma_b = 1.716 \text{ ksi}$ \leftarrow

$$\sigma_c = \frac{P_c}{A_c} = \frac{PE_c}{\Sigma EA}$$
 $\sigma_c = 1.93 \text{ ksi}$
 \leftarrow

Remove R_F to create <u>released structure</u>; use superposition to find redundant R_F = y-dir reaction at F

Released structure under actual load; use FBD of ABC to find pin force B_v

$$\Sigma M_A = 0$$
 $B_y = \frac{1}{\frac{L}{3}} \cdot (3 \cdot P \cdot L)$ $B_y \to 9 \cdot P$ acts upward on ABC so acts downward on DBF

Find vert, displ. of F in released structure under actual loads
$$\delta_{F1} = \frac{-B_y \cdot L}{2 \cdot EA}$$
 $\delta_{F1} \to \frac{9 \cdot L \cdot P}{2 \cdot EA}$ downward

Apply redundant R_F and find vertical displ. at F in released structure
$$\delta_{F2} = R_F \left(\frac{\frac{L}{3}}{EA} + \frac{L}{2 \cdot EA} \right) \quad \delta_{F2} \to \frac{5 \cdot L \cdot R_F}{6 \cdot EA}$$

Compatibility equ.
$$\delta_{F1} + \delta_{F2} = 0 \qquad R_F = \frac{-\delta_{F1}}{\frac{5}{6} \cdot \frac{L}{EA}} \qquad R_F \to \frac{27 \cdot P}{5}$$

Now use statics to find all remaining reactions FBD of DBF
$$\Sigma M_B = 0$$
 so $D_x = 0$

Entire structure
$$\Sigma F_{\rm X} = 0$$
 $A_{\rm X} = 0$ $\Sigma M_{\rm B} = 0$ $A_{\rm y} = \frac{1}{\frac{L}{3}} \left[-3 \cdot P \cdot \left(\frac{2 \cdot L}{3} \right) \right]$ $A_{\rm y} \rightarrow -6 \cdot P$ $\Sigma F_{\rm y} = 0$ $D_{\rm y} = -R_{\rm F} + 3 \cdot P - A_{\rm y}$ $D_{\rm y} \rightarrow \frac{18 \cdot P}{5}$

The rails are prevented from expanding because of their great length and lack of expansion joints.

Therefore, each rail is in the same condition as a bar with fixed ends (see Example 2-9).

The compressive stress in the rails may be calculated as follows.

$$\Delta T = 120^{\circ} \text{F} - 60^{\circ} \text{F} = 60^{\circ} \text{F}$$

 $\sigma = E\alpha(\Delta T)$
= $(30 \times 10^{6} \text{ psi})(6.5 \times 10^{-6} \text{/°F})(60^{\circ} \text{F})$
 $\sigma = 11,700 \text{ psi}$

INITIAL CONDITIONS

$$L_a = 60 \text{ m}$$
 $T_0 = 10^{\circ}\text{C}$

$$L_s = 60.005 \text{ m}$$
 $T_0 = 10^{\circ}\text{C}$

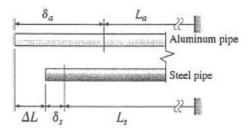
$$\alpha_a = 23 \times 10^{-6} / ^{\circ}\text{C}$$
 $\alpha_s = 12 \times 10^{-6} / ^{\circ}\text{C}$

FINAL CONDITIONS

Aluminum pipe is longer than the steel pipe by the amount $\Delta L = 15$ mm.

 ΔT = increase in temperature

$$\delta_a = \alpha_a(\Delta T)L_a$$
 $\delta_s = \alpha_s(\Delta T)L_s$



From the figure above:

$$\delta_a + L_a = \Delta L + \delta_s + L_s$$

or,
$$\alpha_a(\Delta T)L_a + L_a = \Delta L + \alpha_s(\Delta T)L_s + L_s$$

Solve for ΔT :

$$\Delta T = \frac{\Delta L + (L_s - L_a)}{\alpha_a L_a - \alpha_s L_s} \quad \leftarrow \quad$$

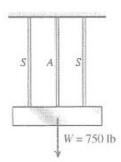
Substitute numerical values:

$$\alpha_a L_a - \alpha_s L_s = 659.9 \times 10^{-6} \text{ m/}^{\circ}\text{C}$$

$$\Delta T = \frac{15 \text{ mm} + 5 \text{ mm}}{659.9 \times 10^{-6} \text{ m/°C} = 30.31°C}$$

$$T = T_0 + \Delta T = 10^{\circ}\text{C} + 30.31^{\circ}\text{C}$$

= 40.3°C \leftarrow



$$S = steel$$
 $A = aluminum$

$$W = 750 \, \text{lb}$$

$$d=\frac{1}{8}$$
 in.

$$A_s = \frac{\pi d^2}{4} = 0.012272 \text{ in.}^2$$

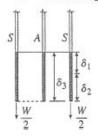
$$E_s = 30 \times 10^6 \text{ psi}$$

$$E_s A_s = 368,155 \text{ lb}$$

$$\alpha_s = 6.5 \times 10^{-6} / {\rm °F}$$

$$\alpha_a = 12 \times 10^{-6} / {\rm °F}$$

L = Initial length of wires



 $δ_1$ = increase in length of a steel wire due to temperature increase ΔT

$$= \alpha_s (\Delta T)L$$

 δ_2 = increase in length of a steel wire due to load W/2

$$=\frac{WL}{2E_{s}A_{s}}$$

 δ_3 = increase in length of aluminum wire due to temperature increase ΔT

$$= \alpha_a(\Delta T)L$$

For no load in the aluminum wire:

$$\delta_1 + \delta_2 = \delta_3$$

$$\alpha_s(\Delta T)L + \frac{WL}{2E_sA_s} = \alpha_a(\Delta T)L$$

or

$$\Delta T = \frac{W}{2E_s A_s (\alpha_n - \alpha_s)} \quad \leftarrow$$

Substitute numerical values:

$$\Delta T = \frac{750 \text{ lb}}{(2)(368,155 \text{ lb})(5.5 \times 10^{-6} \text{/°F})}$$
$$= 185 \text{°F} \qquad \leftarrow$$

NOTE: If the temperature increase is larger than ΔT , the aluminum wire would be in compression, which is not possible. Therefore, the steel wires continue to carry all of the load. If the temperature increase is less than ΔT , the aluminum wire will be in tension and carry part of the load.

NUMERICAL PROPERTIES

$$d_r=15~\mathrm{mm}$$
 $d_b=12~\mathrm{mm}$ $d_w=20~\mathrm{mm}$ $t_c=10~\mathrm{mm}$ $t_{\mathrm{wall}}=18~\mathrm{mm}$ $au_b=45~\mathrm{MPa}$ $lpha=12\,(10^{-6})$ $E=200~\mathrm{GPa}$

(a) Temperature drop resulting in bolt shear stress $\varepsilon = \alpha \Delta T$ $\sigma = E \alpha \Delta T$

Rod force
$$=P=(E\alpha\Delta T)\frac{\pi}{4}d_r^2$$
 and bolt in double shear with shear stress $\tau=\frac{P}{2}$ $\tau=\frac{P}{2\frac{\pi}{4}d_b^2}$
$$\tau_b=\frac{2}{\pi d_b^2}\Big[(E\alpha\Delta T)\frac{\pi}{4}d_r^2\Big] \qquad \tau_b=\frac{E\alpha\Delta T}{2}\Big(\frac{d_r}{d_b}\Big)^2$$

$$\Delta T = \frac{2\tau_b}{E(1000)\alpha} \left(\frac{d_b}{d_r}\right)^2 \qquad \Delta T = 24^{\circ}\text{C} \qquad P = (E\alpha \Delta T)\frac{\pi}{4}d_r^2 \qquad P = 10 \text{ kN}$$

$$\sigma_{\text{rod}} = \frac{P1000}{\frac{\pi}{4}d_r^2} \qquad \boxed{\sigma_{\text{rod}} = 57.6 \text{ MPa}}$$

(b) Bearing stresses

Bolt and clevis
$$\sigma_{bc} = \frac{\frac{P}{2}}{d_b t_c}$$

$$\sigma_{bc} = 42.4 \text{ MPa}$$

$$\text{Washer at wall.} \qquad \sigma_{bw} = \frac{P}{\frac{\pi}{4} (d_w^2 - d_r^2)}$$

$$\sigma_{bw} = 74.1 \text{ MPa}$$

(c) If the connection to the wall at B is changed to an end plate with two bolts (see Fig. b), what is the required diameter d_b of each bolt if temperature drop ΔT = 38°C and the allowable bolt stress is 90 MPa? Find force in rod due to temperature drop.

$$\Delta T = 38^{\circ}\text{C}$$
 $P = (E\alpha \Delta T) \frac{\pi}{4} d_r^2$
$$P = 200 \text{ } GPa \frac{\pi}{4} (15 \text{ mm})^2 \left[12 \left(10^{-6} \right) \right] (38) = 16116 \text{ N} \qquad P = 16.12 \text{ kN}$$

Each bolt carries one half of the force P:

$$d_b = \sqrt{\frac{\frac{16 \, 12 \, \text{kN}}{2}}{\frac{\pi}{4} (90 \, \text{MPa})}} = 10.68 \, \text{mm}) \qquad \boxed{d_b = 10.68 \, \text{mm}}$$

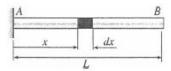
(a) One degree statically indeterminate—use superposition select reaction R_B as the redundant; follow procedure Bar with nonuniform temperature change.

 $0 \xrightarrow{\Delta T} \xrightarrow{\Delta T_B}$

At distance x:

$$\Delta T = \Delta T_B \left(\frac{x^3}{L^3} \right)$$

Remove the support at the end B of the bar:



Consider an element dx at a distance x from end A.

 $d\delta$ = Elongation of element dx

$$d\delta = \alpha(\Delta T)dx = \alpha(\Delta T_B) \left(\frac{x^3}{L^3}\right) dx$$

 $d\delta$ = elongation of bar

$$\delta = \int_0^L d\delta = \int_0^L \alpha(\Delta T_B) \left(\frac{x^3}{L^3}\right) dx = \frac{1}{4} \alpha(\Delta T_B) L$$

Compressive force P required to shorten the bar by the amount δ

$$P = \frac{EA\delta}{I} = \frac{1}{4}EA\alpha(\Delta T_B)$$

COMPRESSIVE STRESS IN THE BAR

$$\sigma_c = \frac{P}{A} = \frac{E\alpha(\Delta T_B)}{4} \leftarrow$$

(b) ONE DEGREE STATICALLY INDETERMINATE—USE SUPERPOSITION.

Select reaction R_B as the redundant then compute bar elongations due to ΔT and due to R_B

 $\delta_{B1} = \alpha \Delta T_B \frac{L}{4}$ due to temperature from above

$$\delta_{B2} = R_B \left(\frac{1}{k} + \frac{L}{EA} \right)$$

Compatibility: solve for R_B : $\delta_{B1} + \delta_{B2} = 0$

$$R_B = \frac{-\left(\alpha \Delta T_B \frac{L}{4}\right)}{\left(\frac{1}{k} + \frac{L}{EA}\right)}$$

$$R_B = -\alpha \Delta T_B \left[\frac{EA}{4\left(\frac{EA}{kL} + 1\right)} \right]$$

So compressive stress in bar is

$$\sigma_c = \frac{R_B}{A}$$
 $\sigma_c = \frac{E\alpha(\Delta T_B)}{4\left(\frac{EA}{kL} + 1\right)} \leftarrow$

NOTE: σ_c in part (b) is the same as in part (a) if spring constant k goes to infinity.

CLICK HERE TO ACCESS THE COMPLETE Solutions

Problem 2.5-6

$$A = 2 \cdot \left(1090 mm^2\right) = 2180 \cdot mm^2 \qquad \qquad k = 1750 \, \frac{kN}{m} \qquad \qquad \Delta T = 45 \qquad \alpha = 12 \cdot \left(10^{-6}\right) \qquad \qquad L = 3m \qquad E = 205 \, GPa$$

Assume that beam and spring are stress free at the start, then apply temperature increase ΔT . Select R_C as the redundant to remove to create the released structure

Apply ΔT to beam in released structure $\delta_{C1} = \alpha \cdot \Delta T \cdot L = 1.62 \cdot mm$

 $\delta_{C2} = R_{C} \cdot \left(\frac{L}{E \cdot A} + \frac{1}{k} \right) \qquad \frac{L}{E \cdot A} + \frac{1}{k} = 0.578 \cdot \frac{mm}{kN}$

Axial normal compressive stress in beam $\sigma_T = \frac{R_C}{A} = -1.285 \cdot MPa$

Displacement at B using superposition $\delta_B = \frac{R_C \cdot L}{E \cdot A} + \alpha \cdot \Delta T \cdot L = 1.601 \cdot mm \qquad \frac{R_C}{k} = -1.601 \cdot mm$ elongation of beam is equal to shortening of spring

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Problem 2.5-7

E = 29000ksi
$$\alpha = 6.5 \cdot 10^{-6}$$
 $\Delta T = 20$ A = 8.24in^2 L = 10ft

Select reaction R_B as the redundant; remove R_B to create released structure. Use superposition - apply ΔT to released structure, then apply redundant. Solve compatibility equation to find R_B then use statics to get R_A

$$\delta_{\text{B1}} = \alpha \cdot \Delta T \cdot L = 0.016 \cdot \text{in}$$
 $\delta_{\text{B2}} = R_{\text{B}} \cdot \frac{L}{EA}$

Compatibility
$$\delta_{B1} + \delta_{B2} = 0$$
 solve for $R_B = \frac{-E \cdot A}{L} \cdot (\alpha \cdot \Delta T \cdot L) = -31.065 \cdot kip$ negative so R_B acts to left

Statics
$$R_A + R_B = 0$$
 so $R_A = -R_B = 31.065 \cdot kip$

Beam is in uniform axial compression due to temperature change; compressive normal stress is $\sigma_T = \frac{R_B}{A} = -3.77 \cdot ksi$

NUMERICAL DATA

$$d_1 = 50 \text{ mm}$$
 $d_2 = 75 \text{ mm}$

$$L_1 = 225 \text{ mm}$$
 $L_2 = 300 \text{ mm}$

$$E = 6.0 \text{ GPa}$$
 $\alpha = 100 \times 10^{-6} \text{/°C}$

$$\Delta T = 30^{\circ}$$
C $k = 50 \text{ MN/m}$

(a) Compressive force N, maximum compressive stress and displacement of pt. C

$$A_1 = \frac{\pi}{4}d_1^2$$
 $A_2 = \frac{\pi}{4}d_2^2$

One-degree statically indeterminate—use R_B as redundant

$$\delta_{B1} = \alpha \Delta T (L_1 + L_2)$$

$$\delta_{B2} = R_B \left(\frac{L_1}{EA_1} + \frac{L_2}{EA_2} \right)$$

Compatibility: $\delta_{B1} = \delta_{B2}$, solve for R_B

$$R_{B} = \frac{\alpha \Delta T (L_{1} + L_{2})}{\frac{L_{1}}{EA_{1}} + \frac{L_{2}}{EA_{2}}} \quad N = R_{B}$$

$$N = 51.8 \text{ kN} \leftarrow$$

Maximum compressive stress in AC since it has the smaller area $(A_1 \le A_2)$:

$$\sigma_{\rm cmax} = \frac{N}{A_1}$$
 $\sigma_{\rm cmax} = 26.4 \, \rm MPa$

Displacement δ_C of point C = superposition of displacements in two released structures at C:

$$\delta_C = \alpha \Delta T(L_1) - R_B \frac{L_1}{EA_1}$$

 $\delta_C = -0.314 \, \text{mm} \quad \leftarrow (-) \, \text{sign means joint } C \, \text{moves}$ left

(b) Compressive force N, maximum compressive stress and displacement of part C for elastic support case

Use R_B as redundant as in part (a):

$$\delta_{B1} = \alpha \Delta T (L_1 + L_2)$$

$$\delta_{B_2} = R_B \left(\frac{L_1}{EA_1} + \frac{L_2}{EA_2} + \frac{1}{k} \right)$$

Now add effect of elastic support; equate δ_{B1} and δ_{B2} then solve for R_B :

$$R_B = \frac{\alpha \Delta T (L_1 + L_2)}{\frac{L_1}{EA_1} + \frac{L_2}{EA_2} + \frac{1}{k}} \quad N = R_B$$

$$N = 31.2 \text{ kN} \leftarrow$$

$$\sigma_{cmax} = \frac{N}{A_1}$$
 $\sigma_{cmax} = 15.91 \text{ MPa} \leftarrow$

Superposition:

$$\delta_C = \alpha \Delta T(L_1) - R_B \left(\frac{L_1}{EA_1} + \frac{1}{k} \right)$$

$$\delta_C = -0.546 \text{ mm} \leftarrow (-) \text{ sign means joint } C$$

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Problem 2.5-9

$$\Delta T = 20$$
 $\alpha = 13 \cdot (10^{-6})$ $\alpha = 10400 \, \text{ksi}$ $\alpha = 2 \, \text{ms}$ $\alpha = 2 \, \text{ms}$

Select reaction R_C as the redundant; remove redundant to create released structure; apply temperature increase and then apply redundant to get displacements at joint C in released structure.

$$\begin{split} \delta_{C1} &= \alpha \cdot \Delta T \cdot L = 9.36 \times 10^{-3} \cdot in \\ \delta_{C2} &= R_C \cdot \left[\frac{\frac{L}{2}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot ln \left(\frac{b_2}{b_1}\right) + \frac{\frac{L}{2}}{E \cdot \left(b_1 \cdot t\right)} \right] \\ &= \frac{\frac{L}{2}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot ln \left(\frac{b_2}{b_1}\right) + \frac{\frac{L}{2}}{E \cdot \left(b_1 \cdot t\right)} = 6.551 \times 10^{-3} \cdot \frac{in}{kip} \end{split}$$

Write compatibility equation then solve for R_C

$$\begin{split} \delta_{\text{C1}} + \delta_{\text{C2}} &= 0 \\ R_{\text{C}} &= \frac{-(\alpha \cdot \Delta T \cdot L)}{\left[\frac{\frac{L}{2}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot ln \left(\frac{b_2}{b_1}\right) + \frac{\frac{L}{2}}{E \cdot \left(b_1 \cdot t\right)}\right]} = -1.429 \cdot \text{kip} \end{split}$$

Statics $R_A + R_C = 0$ $R_A = -R_C = 1.429 \cdot \text{kip}$

Displacement at B using superposition
$$\delta_B = -R_A \cdot \left[\frac{\frac{L}{2}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot \ln \left(\frac{b_2}{b_1}\right) \right] + \alpha \cdot \Delta T \cdot \frac{L}{2} = 2.656 \times 10^{-4} \cdot \text{in}$$
 joint B moves to right

OR
$$\frac{R_{C} \cdot \frac{L}{2}}{E \cdot \left(b_{1} \cdot t\right)} + \alpha \cdot \Delta T \cdot \frac{L}{2} = -2.656 \times 10^{-4} \cdot in$$
 shortening of BC

$$\Delta T = 30$$
 $\alpha = 19 \cdot (10^{-6})$ $L = 2m$ $t = 20mm$ $b_1 = 100mm$ $b_2 = 115mm$ $E = 96GPa$

Select reaction R_C as the redundant; remove redundant to create released structure; apply temperature increase and then apply redundant to get displacements at joint C in released structure.

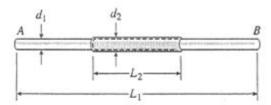
$$\delta_{C1} = \alpha \cdot \Delta T \cdot L = 1.14 \cdot mm \qquad \delta_{C2} = R_C \cdot \left[\frac{\frac{3 \cdot L}{5} + \frac{2 \cdot L}{5}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot ln \left(\frac{b_2}{b_1}\right) \right] \qquad \frac{L}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot ln \left(\frac{b_2}{b_1}\right) = 9.706 \times 10^{-3} \cdot \frac{mm}{kN}$$

Write compatibility equation then solve for R_c

$$\begin{split} & \frac{\delta_{\text{C1}} + \delta_{\text{C2}} = 0}{\left[\frac{L}{\text{E·t·}\left(b_2 - b_1\right)} \cdot \ln\left(\frac{b_2}{b_1}\right)\right]} = -117.457 \cdot \text{kN} \end{split}$$

Statics
$$R_A + R_C = 0$$
 $R_A = -R_C = 117.457 \cdot kN$

Displacement at B using superposition
$$\delta_B = -R_A \left[\frac{\frac{3 \cdot L}{5}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot ln \left(\frac{b_2}{b_1}\right) \right] + \alpha \cdot \Delta T \cdot \frac{3 \cdot L}{5} = 0 \cdot mm$$
 no elongation of AB or
$$R_C \cdot \left[\frac{\frac{2 \cdot L}{5}}{E \cdot t \cdot \left(b_2 - b_1\right)} \cdot ln \left(\frac{b_2}{b_1}\right) \right] + \alpha \cdot \Delta T \cdot \frac{2L}{5} = 0 \cdot mm$$



$$L_1 = 36 \text{ in.}$$
 $L_2 = 12 \text{ in.}$

ELONGATION OF THE TWO OUTER PARTS OF THE BAR

$$\delta_1 = \alpha_s(\Delta T)(L_1 - L_2)$$

= $(6.5 \times 10^{-6})^{\circ}$ F)(500°F)(36 in. - 12 in.)
= 0.07800 in.

ELONGATION OF THE MIDDLE PART OF THE BAR

The steel rod and bronze sleeve lengthen the same
amount, so they are in the same condition as the bolt
and sleeve of Example 2-10. Thus, we can calculate the
elongation from Eq. (2-21):

$$\delta_2 = \frac{(\alpha_s E_s A_s + \alpha_b E_b A_b)(\Delta T)L_2}{E_s A_s + E_b A_b}$$

SUBSTITUTE NUMERICAL VALUES

$$\alpha_s = 6.5 \times 10^{-6} \text{/°F}$$
 $\alpha_b = 11 \times 10^{-6} \text{/°F}$

$$E_s = 30 \times 10^6 \text{ psi}$$
 $E_b = 15 \times 10^6 \text{ psi}$

$$d_1 = 1.0$$
 in.

$$A_s = \frac{\pi}{4} d_1^2 = 0.78540 \text{ in.}^2$$

$$d_2 = 1.25 \text{ in.}$$

$$A_b = \frac{\pi}{4} (d_2^2 - d_1^2) = 0.44179 \text{ in.}^2$$

$$\Delta T = 500^{\circ} \text{F}$$
 $L_2 = 12.0 \text{ in.}$

$$\delta_2 = 0.04493$$
 in.

TOTAL ELONGATION

$$\delta = \delta_1 + \delta_2 = 0.123$$
 in. \leftarrow

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Problem 2.5-12

$$\Delta T = 15$$
 $\alpha_{\overline{T}} = 23 \cdot (10^{-6})$ $L = 1.8 \text{m}$ $r = 36 \text{mm}$ $E = 72 \text{GPa}$ $a = \frac{r}{8} = 4.5 \cdot \text{mm}$

$$E = 72GP$$

$$a = \frac{r}{8} = 4.5 \cdot mn$$

$$A_1 = \pi \cdot r^2 = 4071.504 \cdot mm^2$$

 $A_1 = \pi \cdot r^2 = 4071.504 \cdot mm^2$ Use formulas in **Appendix E, Case 15** for area of slotted segment

$$\alpha = a\cos\left(\frac{a}{r}\right) = 1.445$$

$$b = \sqrt{r^2 - a^2} = 35.718 \cdot mn$$

$$\alpha = a\cos\left(\frac{a}{r}\right) = 1.445$$
 $b = \sqrt{r^2 - a^2} = 35.718 \cdot mm$ $A_2 = 2 \cdot r^2 \cdot \left(\alpha - \frac{a \cdot b}{r^2}\right) = 3425.196 \cdot mm^2$ $\frac{A_2}{A_1} = 0.841$

$$\frac{A_2}{A_1} = 0.84$$

Select reaction R_C as the redundant; remove redundant to create released structure; apply temperature increase and then apply redundant to get displacements at joint C in released structure.

$$\delta_{C1} = \alpha_{T} \Delta T \cdot L = 0.621 \cdot mm$$

$$\delta_{C2} = R_C \left(\frac{2 \cdot \frac{L}{4}}{E \cdot A_1} + \frac{\frac{L}{2}}{E \cdot A_2} \right)$$

$$\delta_{C1} = \alpha_{T} \Delta T \cdot L = 0.621 \cdot \text{mm} \quad \delta_{C2} = R_{C} \cdot \left(\frac{2 \cdot \frac{L}{4}}{E \cdot A_{1}} + \frac{\frac{L}{2}}{E \cdot A_{2}} \right) \qquad \qquad \frac{2 \cdot \frac{L}{4}}{E \cdot A_{1}} + \frac{\frac{L}{2}}{E \cdot A_{2}} = 6.72 \times 10^{-3} \cdot \frac{\text{mm}}{\text{kN}}$$

$$\delta_{C1} + \delta_{C2} = 0$$

Write compatibility equation then solve for R_C
$$\frac{\delta_{C1} + \delta_{C2}}{\left(\frac{2 \cdot \frac{L}{4}}{E \cdot A_1} + \frac{L}{\frac{2}{E \cdot A_2}}\right)} = -92.417 \cdot kN$$

Statics

$$R_A + R_C = 0$$

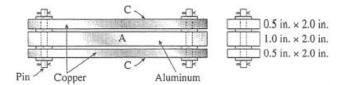
$$R_A + R_C = 0$$
 $R_A = -R_C = 92.417 \text{-kN}$

Thermal compressive stress in solid bar segments

$$\sigma_{T1} = \frac{R_C}{A_1} = -22.698 \cdot MPa$$

and in slotted middle segment

$$\sigma_{T2} = \frac{R_C}{A_2} = -26.982 \cdot MPa$$



Diameter of pin:
$$d_P = \frac{7}{16}$$
 in. = 0.4375 in.

Area of pin:
$$A_P = \frac{\pi}{4} d_P^2 = 0.15033 \text{ in.}^2$$

Copper:
$$E_c = 18,000 \text{ ksi}$$
 $\alpha_c = 9.5 \times 10^{-6} \text{ f}^{\circ}\text{F}$

Aluminum: $E_a = 10,000 \text{ ksi}$

$$\alpha_a = 13 \times 10^{-6} \text{/}^{\circ}\text{F}$$

Use the results of Example 2-10.

Find the forces P_a and P_c in the aluminum bar and copper bar, respectively, from Eq. (2-19).

Replace the subscript "S" in that equation by "a" (for aluminum) and replace the subscript "B" by "c" (for copper), because α for aluminum is larger than α for copper.

$$P_a = P_c = \frac{(\alpha_a - \alpha_c)(\Delta T)E_a A_a E_c A_c}{E_a A_a + E_c A_c}$$

Note that P_a is the compressive force in the aluminum bar and P_c is the combined tensile force in the two copper bars.

$$P_a = P_c = \frac{(\alpha_a - \alpha_c)(\Delta T)E_cA_c}{1 + \frac{E_cA_c}{E_aA_a}}$$

Area of two copper bars: $A_c = 2.0 \text{ in.}^2$

Area of aluminum bar: $A_a = 2.0 \text{ in.}^2$

$$\Delta T = 100^{\circ} F$$

SUBSTITUTE NUMERICAL VALUES:

$$P_a = P_c = \frac{(3.5 \times 10^{-6} \text{f°F})(100^{\circ}\text{F})(18,000 \text{ ksi})(2 \text{ in.}^2)}{1 + \left(\frac{18}{10}\right)\left(\frac{2.0}{2.0}\right)}$$
$$= 4,500 \text{ lb}$$

FREE-BODY DIAGRAM OF PIN AT THE LEFT END



V = shear force in pin

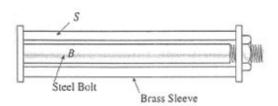
$$=P_c/2$$

$$= 2.250 lb$$

 τ = average shear stress on cross section of pin

$$\tau = \frac{V}{A_P} = \frac{2,250 \text{ lb}}{0.15033 \text{ in.}^2}$$

$$\tau = 15.0 \text{ ksi} \leftarrow$$



Subscript S means "sleeve".

Subscript B means "bolt".

Use the results of Example 2-10.

 σ_S = compressive force in sleeve

EQUATION (2-20a):

$$\sigma_S = \frac{(\alpha_S - \alpha_B)(\Delta T)E_S E_B A_B}{E_S A_S + E_B A_B}$$
(Compression)

Solve for ΔT :

$$\Delta T = \frac{\sigma_S(E_S A_S + E_B A_B)}{(\alpha_S - \alpha_B)E_S E_B A_B}$$

or

$$\Delta T = \frac{\sigma_S}{E_S(\alpha_S - \alpha_B)} \left(1 + \frac{E_S A_S}{E_B A_B} \right) \leftarrow$$

SUBSTITUTE NUMERICAL VALUES:

$$\sigma_S = 25 \text{ MPa}$$

$$d_2 = 36 \text{ mm}$$
 $d_1 = 26 \text{ mm}$ $d_B = 25 \text{ mm}$

$$E_S = 100 \text{ GPa}$$
 $E_B = 200 \text{ GPa}$

$$\alpha_S = 21 \times 10^{-6} / ^{\circ}\text{C}$$
 $\alpha_B = 10 \times 10^{-6} / ^{\circ}\text{C}$

$$A_S = \frac{\pi}{4}(d_2^2 - d_1^2) = \frac{\pi}{4}(620 \text{ mm}^2)$$

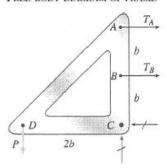
$$A_B = \frac{\pi}{4} (d_B)^2 = \frac{\pi}{4} (625 \text{ mm}^2) 1 + \frac{E_S A_S}{E_B A_B} = 1.496$$

$$\Delta T = \frac{25 \text{ MPa } (1.496)}{(100 \text{ GPa})(11 \times 10^{-6} \text{ pc})}$$

$$\Delta T = 34^{\circ}C \leftarrow$$

(Increase in temperature)

FREE-BODY DIAGRAM OF FRAME

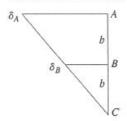


EQUATION OF EQUILIBRIUM

 $\Sigma M_C = 0 \Leftrightarrow \bigcirc$

$$P(2b) - T_A(2b) - T_B(b) = 0$$
 or $2T_A + T_B = 2P$ (Eq. 1)

DISPLACEMENT DIAGRAM



EQUATION OF COMPATIBILITY

$$\delta_A = 2\delta_B$$
 (Eq. 2)

(a) LOAD P ONLY

Force-displacement relations:

$$\delta_A = \frac{T_A L}{EA} \quad \delta_B = \frac{T_B L}{EA} \tag{Eq. 3, 4}$$

(L = length of wires at A and B.)

Substitute (3) and (4) into Eq. (2):

$$\frac{T_A L}{EA} = \frac{2T_B L}{EA}$$
 or $T_A = 2T_B$ (Eq. 5)

Solve simultaneously Eqs. (1) and (5):

$$T_A = \frac{4P}{5}$$
 $T_B = \frac{2P}{5}$ (Eqs. 6, 7)

Numerical values:

P = 500 lb

$$T_A = 400 \text{ lb}$$
 $T_B = 200 \text{ lb} \leftarrow$

(b) Load P and temperature increase ΔT

Force-displacement and temperature-displacement relations:

$$\delta_A = \frac{T_A L}{EA} + \alpha(\Delta T) L \tag{Eq. 8}$$

$$\delta_B = \frac{T_B L}{FA} + \alpha(\Delta T) L \tag{Eq. 9}$$

Substitute (8) and (9) into Eq. (2):

$$\frac{T_AL}{EA} + \alpha(\Delta T)L = \frac{2T_BL}{EA} + 2\alpha(\Delta T)L$$

or
$$T_A - 2T_B = EA\alpha(\Delta T)$$
 (Eq. 10)

Solve simultaneously Eqs. (1) and (10):

$$T_A = \frac{1}{5} [4P + EA\alpha(\Delta T)]$$
 (Eq. 11)

$$T_B = \frac{2}{5}[P - EA\alpha(\Delta T)]$$
 (Eq. 12)

Substitute numerical values:

$$P = 500 \text{ lb}$$
 $EA = 120,000 \text{ lb}$

$$\Delta T = 180^{\circ} F$$

$$\alpha = 12.5 \times 10^{-6} / {}^{\circ}\text{F}$$

$$T_A = \frac{1}{5}(2000 \text{ lb} + 270 \text{ lb}) = 454 \text{ lb} \quad \leftarrow$$

$$T_B = \frac{2}{5}(500 \text{ lb} - 270 \text{ lb}) = 92 \text{ lb} \quad \leftarrow$$

(c) WIRE B BECOMES SLACK

Set
$$T_B = 0$$
 in Eq. (12):

$$P = EA\alpha(\Delta T)$$

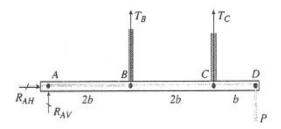
or

$$\Delta T = \frac{P}{EA\alpha} = \frac{500 \text{ lb}}{(120,000 \text{ lb})(12.5 \times 10^{-6} \text{/°F})}$$
$$= 333.3 \text{°F}$$

Further increase in temperature:

$$\Delta T = 333.3^{\circ}F - 180^{\circ}F$$
$$= 153^{\circ}F \quad \leftarrow$$

FREE-BODY DIAGRAM OF BAR ABCD



$$T_B$$
 = force in cable B T_C = force in cable C $d_B = 12 \text{ mm}$ $d_C = 20 \text{ mm}$

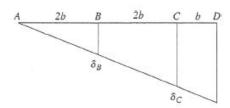
From Table 2-1:

$$A_B = 76.7 \text{ mm}^2$$
 $E = 140 \text{ GPa}$
 $\Delta T = 60^{\circ}\text{C}$ $A_C = 173 \text{ mm}^2$
 $\alpha = 12 \times 10^{-6} / ^{\circ}\text{C}$

EQUATION OF EQUILIBRIUM

$$\Sigma M_A = 0 \quad \Leftrightarrow \quad T_B(2b) + T_C(4b) - P(5b) = 0$$
 or $2T_B + 4T_C = 5P$ (Eq. 1)

DISPLACEMENT DIAGRAM



COMPATIBILITY:

$$\delta_C = 2\delta_B$$
 (Eq. 2)

FORCE-DISPLACEMENT AND TEMPERATURE-DISPLACEMENT

$$\delta_B = \frac{T_B L}{E A_B} + \alpha(\Delta T) L \tag{Eq. 3}$$

$$\delta_C = \frac{T_C L}{E A_C} + \alpha (\Delta T) L \tag{Eq. 4}$$

Substitute Eqs. (3) and (4) into Eq. (2):

$$\frac{T_CL}{EA_C} + \alpha(\Delta T)L = \frac{2T_BL}{EA_B} + 2\alpha(\Delta T)L$$

or

$$2T_B A_C - T_C A_B = -E\alpha(\Delta T) A_B A_C$$
 (Eq. 5)

SUBSTITUTE NUMERICAL VALUES INTO EQ. (5):

$$T_B(346) - T_C(76.7) = -1,338,000$$
 (Eq. 6) in which T_B and T_C have units of newtons.

SOLVE SIMULTANEOUSLY EQS. (1) AND (6):

$$T_B = 0.2494 P - 3,480$$
 (Eq. 7)

$$T_C = 1.1253 P + 1,740$$
 (Eq. 8)

in which P has units of newtons.

Solve Eqs. (7) and (8) for the load P:

$$P_B = 4.0096 T_B + 13,953$$
 (Eq. 9)

$$P_C = 0.8887 T_C - 1,546$$
 (Eq. 10)

ALLOWABLE LOADS

From Table 2-1:

$$(T_B)_{\text{ULT}} = 102,000 \text{ N}$$
 $(T_C)_{\text{ULT}} = 231,000 \text{ N}$

Factor of safety
$$= 5$$

$$(T_B)_{\text{allow}} = 20,400 \text{ N}$$
 $(T_C)_{\text{allow}} = 46,200 \text{ N}$

From Eq. (9):
$$P_B = (4.0096)(20,400 \text{ N}) + 13,953 \text{ N}$$

= 95,700 N

From Eq. (10):
$$P_C = (0.8887)(46,200 \text{ N}) - 1546 \text{ N}$$

= 39,500 N

Cable C governs.

$$P_{\text{allow}} = 39.5 \text{ kN} \leftarrow$$

Numerical data:

$$L=25$$
 in. $d=2$ in. $\delta=0.008$ in. $k=1.2\times(10^6)$ lb/in. $E=16\times(10^6)$ psi $\alpha=9.6\times(10^{-6})$ /°F $\Delta T=50$ °F $A=\frac{\pi}{4}d^2$ $A=3.14159$ in.²

(a) One-degree statically indeterminate if gap closes

$$\Delta = \alpha \Delta T L$$
 $\Delta = 0.012$ in.

Select R_A as redundant and do superposition analysis:

$$\delta_{A1} = \Delta \quad \delta_{A2} = R_A \left(\frac{L}{EA} + \frac{1}{k} \right)$$

Compatibility: $\delta_{A1} + \delta_{A2} = \delta$ $\delta_{A2} = \delta - \delta_{A1}$

$$R_A = \frac{\delta - \Delta}{\frac{L}{EA} + \frac{1}{k}} \quad R_A = -3006 \text{ lb}$$

Compressive stress in bar:

$$\sigma = \frac{R_A}{A}$$
 $\sigma = -957 \text{ psi}$

(b) Force in spring $F_k = R_C$ Statics

$$R_A + R_C = 0$$

 $R_C = -R_A$
 $R_C = 3006 \text{ lb} \leftarrow$

(c) Find compressive stress in BAR if k goes to infinity

$$R_A = \frac{\delta - \Delta}{L}$$
 $R_A = -8042 \text{ lb}$ $\sigma = \frac{R_A}{A}$

FROM EXPRESSION FOR R_A ABOVE, 1/k GOES TO ZERO

$$\sigma = -2560 \text{ psi} \leftarrow$$



Initial prestress: $\sigma_1 = 42 \text{ MPa}$

Initial temperature: $T_1 = 20^{\circ}\text{C}$

$$E = 200 \text{ GPa}$$

$$\alpha = 14 \times 10^{-6} / ^{\circ} \text{C}$$

(a) Stress σ when temperature drops to 0° C

$$T_2 = 0$$
°C $\Delta T = 20$ °C

NOTE: Positive ΔT means a decrease in temperature and an increase in the stress in the wire.

Negative ΔT means an increase in temperature and a decrease in the stress.

Stress σ equals the initial stress σ_1 plus the additional stress σ_2 due to the temperature drop.

$$\sigma_2 = E\alpha(\Delta T)$$

$$\sigma = \sigma_1 + \sigma_2 = \sigma_1 + E\alpha(\Delta T)$$
= 42 MPa + (200 GPa)(14 × 10⁻⁶/°C)(20°C)
= 42 MPa + 56 MPa = 98 MPa \leftarrow

(b) Temperature when stress equals zero

$$\sigma = \sigma_1 + \sigma_2 = 0$$
 $\sigma_1 + E\alpha(\Delta T) = 0$

$$\Delta T = -\frac{\sigma_1}{E\alpha}$$

(Negative means increase in temp.)

$$\Delta T = -\frac{42 \text{ MPa}}{(200 \text{ GPa})(14 \times 10^{-6} \text{/°C})} = -15^{\circ}\text{C}$$
$$T = 20^{\circ}\text{C} + 15^{\circ}\text{C} = 35^{\circ}\text{C} \leftarrow$$

$$n = 1.5$$
 $p = \frac{1}{16}in$ $A_s = 0.85in^2$ $A_A = 4.5in^2$

$$L = 20in$$
 $E_s = 29000ksi$ $E_A = 10600ksi$

Select force in tube as the redundant. Cut through aluminum tube at right end to expose internal force F_A to create released structure. Apply n turns of turnbuckles to released structure to find relative displacement between ends of cut tube.

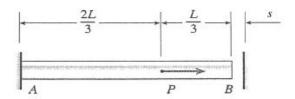
$$\delta_1 = 2 \cdot n \cdot p = 0.187 \cdot in$$
 Note that n turns of a turnbuckle moves ends together by factor of two

Now apply pair of internal forces F_T to ends of tube then again find relative displacement. Force F_A shortens both cables and elongates the tube.

$$\delta_2 = \mathbf{F_A} \cdot \left(\frac{L}{E_A \cdot A_A} + \frac{L}{2 \cdot E_s \cdot A_s} \right) \qquad \qquad \frac{L}{E_A \cdot A_A} + \frac{L}{2 \cdot E_s \cdot A_s} = 8.25 \times 10^{-4} \cdot \frac{in}{kip}$$

$$\text{Compatibility equation} \qquad \frac{\pmb{\delta_1} + \pmb{\delta_2} = 0}{\frac{L}{E_A \cdot A_A} + \frac{L}{2 \cdot E_S \cdot A_S}} = -227.282 \cdot \text{kip}$$

Shortening of aluminum tube
$$\delta_A = \frac{F_A \cdot L}{E_A \cdot A_A} = -0.0953 \cdot in$$



L = length of bar

s = size of gap

EA = axial rigidity

Reactions must be equal; find s.

COMPATIBILITY EQUATION

$$\delta_1 - \delta_2 = s$$
 or
$$\frac{2PL}{3EA} - \frac{R_BL}{EA} = s$$
 (Eq. 1)

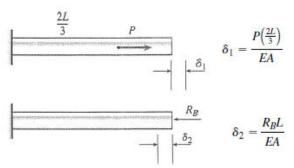
EQUILIBRIUM EQUATION

 R_A = reaction at end A (to the left)

 R_B = reaction at end B (to the left)

$$P = R_A + R_B$$

FORCE-DISPLACEMENT RELATIONS



Reactions must be equal.

$$\therefore R_A = R_B \quad P = 2R_B \quad R_B = \frac{P}{2}$$

Substitute for R_B in Eq. (1):

$$\frac{2PL}{3EA} - \frac{PL}{2EA} = s$$
 or $s = \frac{PL}{6EA} \leftarrow$

NOTE: The gap closes when the load reaches the value P/4. When the load reaches the value P, equal to 6EAs/L, the reactions are equal $(R_A = R_B = P/2)$. When the load is between P/4 and P, R_A is greater than R_B . If the load exceeds P, R_B is greater than R_A .

(a) Find reactions at A and B for applied force P_1 First compute P_1 , required to close gap:

$$P_1 = \frac{E_1 A_1}{L_1} s$$
 $P_1 = 231.4 \text{ k}$ \leftarrow

Statically indeterminate analysis with R_{B} as the redundant:

$$\delta_{B1} = -s \quad \delta_{B2} = R_B \left(\frac{L_1}{E_1 A_1} + \frac{L_2}{E_2 A_2} \right)$$

Compatibility: $\delta_{B1} + \delta_{B2} = 0$

$$R_B = \frac{s}{\left(\frac{L_1}{E_1 A_1} + \frac{L_2}{E_2 A_2}\right)} \quad R_B = 55.2 \text{ k} \quad \leftarrow$$

(b) Find reactions at A and B for applied force P_2

$$P_2 = \frac{E_2 A_2}{\frac{L_2}{2}} s$$
 $P_2 = 145.1 \text{ k}$ \leftarrow

Analysis after removing P_2 is same as in part (a), so reaction forces are the same

(c) Maximum shear stress in pipe 1 or 2 when either P_1 or P_2 $\frac{P_1}{A_1} = \frac{P_1}{2} \quad \tau_{\max a} = 13.39 \text{ ksi} \quad \leftarrow$

 $\tau_{\text{max}b} = \frac{\frac{P_2}{A_2}}{2} \quad \tau_{\text{max}b} = 19.44 \text{ ksi} \quad \leftarrow$

(d) Required ΔT and reactions at A and B

$$\Delta T_{\text{reqd}} = \frac{s}{\alpha_1 L_1 + \alpha_2 L_2} \quad \Delta T_{\text{reqd}} = 65.8^{\circ} \text{F} \quad \leftarrow$$

If pin is inserted but temperature remains at ΔT above ambient temperature, reactions are zero.

(e) If TEMPERATURE RETURNS TO ORIGINAL AMBIENT TEMPERATURE, FIND REACTIONS AT A AND B statically indeterminate analysis with R_B as the redundant Compatibility: $\delta_{B1} + \delta_{B2} = 0$ Analysis is the same as in parts (a) and (b) above since gap s is the same, so reactions are the same.

With gap s closed due to ΔT , structure is one-degree statically-indeterminate; select internal force (Q) at juncture of bar and spring as the redundant. Use superposition of two released structures in the solution.

 δ_{rel1} = relative displacement between end of bar at C and end of spring due to ΔT

 $\delta_{\text{rell}} = \alpha \Delta T (L_1 + L_2)$ $\delta_{\text{rell}} \text{ is greater than gap length } s$

 δ_{rel2} = relative displacement between ends of bar and spring due to pair of forces Q, one on end of bar at C and the other on end of spring

$$\delta_{\text{rel2}} = Q\left(\frac{L_1}{EA_1} + \frac{L_2}{EA_2}\right) + \frac{Q}{k_3}$$

$$\delta_{\text{rel2}} = Q \left(\frac{L_1}{EA_1} + \frac{L_2}{EA_2} + \frac{1}{k_3} \right)$$

Compatibility: $\delta_{rel1} + \delta_{rel2} = s \quad \delta_{rel2} = s - \delta_{rel1}$

$$\delta_{\text{rel2}} = s - \alpha \Delta T (L_1 + L_2)$$

$$Q = \frac{s - \alpha \Delta T (L_1 + L_2)}{\frac{L_1}{EA_1} + \frac{L_2}{EA_2} + \frac{1}{k_3}}$$

$$Q = \frac{EA_1A_2k_3}{L_1A_2k_3 + L_2A_1k_3 + EA_1A_2}$$

$$[s - \alpha \Delta T(L_1 + L_2)]$$

(a) Reactions at A and D

Statics:
$$R_A = -Q$$
 $R_D = Q$

$$R_{A} = \frac{-s + \alpha \Delta T(L_{1} + L_{2})}{\frac{L_{1}}{EA_{1}} + \frac{L_{2}}{EA_{2}} + \frac{1}{k_{3}}} \leftarrow$$

$$R_D = -R_A \leftarrow$$

(b) DISPLACEMENTS AT B AND C Use superposition of displacements in the two released structures:

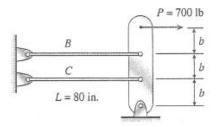
$$\delta_B = \alpha \Delta T(L_1) - R_A \left(\frac{L_1}{EA_1}\right) \leftarrow$$

$$\delta_B = \alpha \Delta T(L_1) - \frac{[-s + \alpha \Delta T(L_1 + L_2)]}{\frac{L_1}{EA_1} + \frac{L_2}{EA_2} + \frac{1}{k_3}} \left(\frac{L_1}{EA_1}\right)$$

$$\delta_C = \alpha \Delta T (L_1 + L_2) - R_A \left(\frac{L_1}{EA_1} + \frac{L_2}{EA_2} \right) \leftarrow$$

$$\delta_C = \alpha \Delta T (L_1 + L_2) -$$

$$\frac{[-s + \alpha \Delta T(L_1 + L_2)]}{\frac{L_1}{EA_1} + \frac{L_2}{EA_2} + \frac{1}{k_3}} \left(\frac{L_1}{EA_1} + \frac{L_2}{EA_2}\right)$$



$$P = 700 \text{ lb}$$

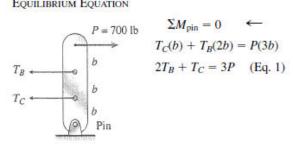
$$A = 0.03 \text{ in.}^2$$

$$E = 30 \times 10^{6} \, \text{psi}$$

$$L_R = 79.98 \text{ in.}$$

$$L_C = 79.95$$
 in.

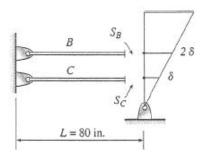
EQUILIBRIUM EQUATION



DISPLACEMENT DIAGRAM

$$S_B = 80 \text{ in.} - L_B = 0.02 \text{ in.}$$

$$S_C = 80 \text{ in. } -L_C = 0.05 \text{ in.}$$



Elongation of wires:

$$\delta_{\rm B} = S_B + 2\delta$$

$$+2\delta$$
 (Eq. 2)

$$\delta_C = S_C + \delta$$
 (Eq. 3)

FORCE-DISPLACEMENT RELATIONS

$$\delta_B = \frac{T_B L}{EA} \quad \delta_C = \frac{T_C L}{EA}$$
 (Eqs. 4, 5)

SOLUTION OF EQUATIONS

Combine Eqs. (2) and (4):

$$\frac{T_B L}{F_A} = S_B + 2\delta \tag{Eq. 6}$$

Combine Eqs. (3) and (5):

$$\frac{T_C L}{F_A} = S_C + \delta \tag{Eq. 7}$$

Eliminate δ between Eqs. (6) and (7):

$$T_B - 2T_C = \frac{EAS_B}{L} - \frac{2EAS_C}{L}$$
 (Eq. 8)

Solve simultaneously Eqs. (1) and (8):

$$T_B = \frac{6P}{5} + \frac{EAS_B}{5L} - \frac{2EAS_C}{5L} \quad \longleftarrow$$

$$T_C = \frac{3P}{5} - \frac{2EAS_B}{5L} + \frac{4EAS_C}{5L} \leftarrow$$

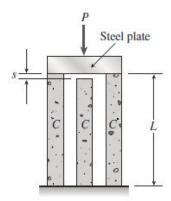
SUBSTITUTE NUMERICAL VALUES:

$$\frac{EA}{5L} = 2250 \text{ lb/in.}$$

$$T_B = 840 \text{ lb} + 45 \text{ lb} - 225 \text{ lb} = 660 \text{ lb}$$

$$T_C = 420 \text{ lb} - 90 \text{ lb} + 450 \text{ lb} = 780 \text{ lb}$$

(Both forces are positive, which means tension, as required for wires.)



s = size of gap = 1.0 mm

L = length of posts = 2.0 m

 $A = 40,000 \text{ mm}^2$

 $\sigma_{\text{allow}} = 20 \text{ MPa}$

E = 30 GPa

C = concrete post

DOES THE GAP CLOSE?

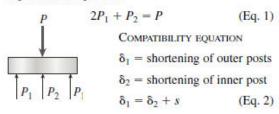
Stress in the two outer posts when the gap is just closed:

$$\sigma = E\varepsilon = E\left(\frac{s}{L}\right) = (30 \text{ GPa})\left(\frac{1.0 \text{ mm}}{2.0 \text{ m}}\right)$$

$$= 15 \text{ MPa}$$

Since this stress is less than the allowable stress, the allowable force *P* will close the gap.

EQUILIBRIUM EQUATION



FORCE-DISPLACEMENT RELATIONS

$$\delta_1 = \frac{P_1 L}{EA} \quad \delta_2 = \frac{P_2 L}{EA}$$
 (Eqs. 3, 4)

SOLUTION OF EQUATIONS

Substitute (3) and (4) into Eq. (2):

$$\frac{P_1L}{EA} = \frac{P_2L}{EA} + s \quad \text{or} \quad P_1 - P_2 = \frac{EAs}{L} \quad \text{(Eq. 5)}$$

Solve simultaneously Eqs. (1) and (5):

$$P = 3P_1 - \frac{EAs}{L}$$

By inspection, we know that P_1 is larger than P_2 . Therefore, P_1 will control and will be equal to $\sigma_{\text{allow}} A$.

$$P_{\text{allow}} = 3\sigma_{\text{allow}} A - \frac{EAs}{L}$$

$$= 2400 \text{ kN} - 600 \text{ kN} = 1800 \text{ kN}$$

$$= 1.8 \text{ MN} \qquad \longleftarrow$$

The figure shows a section through the pipe, cap and rod Numerical properties

$$L_{ci} = 48 \text{ in.}$$
 $E_s = 30000 \text{ ksi}$ $E_b = 14,000 \text{ ksi}$ $E_c = 12,000 \text{ ksi}$ $t_c = 1 \text{ in.}$ $p = 52 \times (10^{-3}) \text{ in.}$ $n = \frac{1}{4}$ $d_w = \frac{3}{4} \text{ in.}$ $d_r = \frac{1}{2} \text{ in.}$ $d_o = 6 \text{ in.}$ $d_i = 5.625 \text{ in.}$

(a) Forces and stresses in PIPE and ROD

One degree statically indeterminate—cut rod at cap and use force in rod (Q) as the redundant:

 δ_{rel1} = relative displacement between cut ends of rod due to 1/4 turn of nut

$$\delta_{\text{rel1}} = -np$$
 Ends of rod move apart, not together, so this is $(-)$.

 δ_{rel2} = relative displacement between cut ends of rod due pair of forces Q

$$\delta_{\text{rel2}} = Q \left(\frac{L + 2t_c}{E_b A_{\text{rod}}} + \frac{L_{ci}}{E_c A_{\text{pipe}}} \right)$$

$$A_{\text{rod}} = \frac{\pi}{4} d_r^2 \qquad A_{\text{pipe}} = \frac{\pi}{4} (d_o^2 - d_i^2)$$

$$A_{\text{rod}} = 0.196 \text{ in.}^2$$
 $A_{\text{pipe}} = 3.424 \text{ in.}^2$
Compatibility equation: $\delta_{\text{rel}1} + \delta_{\text{rel}2} = 0$

$$Q = \frac{np}{\frac{L_{ci} + 2t_c}{E_b A_{\text{rod}}} + \frac{L_{ci}}{E_c A_{\text{pipe}}}}$$

$$Q = 0.672 \text{ k} \qquad F_{\text{rod}} = Q$$

Statics:
$$F_{pipe} = -Q$$

Stresses:
$$\sigma_c = \frac{F_{\text{pipe}}}{A_{\text{pipe}}}$$
 $\sigma_c = -0.196 \text{ ksi} \leftarrow$

$$\sigma_b = \frac{F_{\text{rod}}}{A_{\text{prd}}}$$
 $\sigma_b = 3.42 \text{ ksi} \leftarrow$

(b) Bearing and shear stresses in steel cap

$$\sigma_b = \frac{F_{\text{rod}}}{\frac{\pi}{4}(d_w^2 - d_r^2)} \qquad \sigma_b = 2.74 \text{ ksi} \qquad \longleftarrow$$

$$\tau_c = \frac{F_{\rm rod}}{\pi d_w t_c}$$
 $\tau_c = 0.285 \text{ ksi}$ \leftarrow



 $L = 200 \, \text{mm}$

P = 1.0 mm

 $E_x = 200 \text{ GPa}$

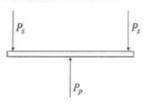
 $A_s = 36.0 \text{ mm}^2 \text{ (for one bolt)}$

 $E_p = 7.5 \text{ GPa}$

 $A_p = 960 \text{ mm}^2$

n = 1 (See Eq. 2-22)

EQUILIBRIUM EQUATION

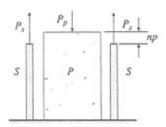


 P_x = tensile force in one steel bolt

 P_p = compressive force in plastic cylinder

$$P_p = 2P_s$$

COMPATIBILITY EQUATION



 δ_{ν} = elongation of steel bolt

 δ_p = shortening of plastic cylinder

$$\delta_s + \delta_p = np$$
 (Eq. 2)

FORCE-DISPLACEMENT RELATIONS

$$\delta_s = \frac{P_s L}{E_s A_s} \quad \delta_p = \frac{P_p L}{E_p A_p} \quad \text{(Eq. 3, Eq. 4)}$$

SOLUTION OF EQUATIONS

Substitute (3) and (4) into Eq. (2):

$$\frac{P_s L}{E_s A_s} + \frac{P_p L}{E_n A_n} = np$$
 (Eq. 5)

Solve simultaneously Eqs. (1) and (5):

$$P_p = \frac{2npE_sA_sE_pA_p}{L(E_pA_p + 2E_sA_s)}$$

STRESS IN THE PLASTIC CYLINDER

$$\sigma_p = \frac{P_p}{A_p} = \frac{2np E_s A_s E_p}{L(E_p A_p + 2E_s A_s)}$$

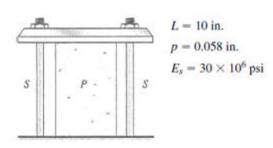
SUBSTITUTE NUMERICAL VALUES:

$$N = E_s A_s E_p = 54.0 \times 10^{15} \text{ N}^2/\text{m}^2$$

$$D = E_p A_p + 2E_s A_s = 21.6 \times 10^6 \text{ N}$$

$$\sigma_p = \frac{2np}{L} \left(\frac{N}{D} \right) = \frac{2(1)(1.0 \text{ mm})}{200 \text{ mm}} \left(\frac{N}{D} \right)$$
= 25.0 MPa

(Eq. 1)



 $A_s = 0.06 \text{ in.}^2$ (for one bolt)

 $E_p = 500 \text{ ksi}$

$$A_p = 1.5 \text{ in.}^2$$

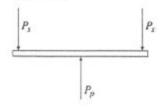
n = 1 (see Eq. 2-22)

EQUILIBRIUM EQUATION

 P_s = tensile force in one steel bolt

 P_p = compressive force in plastic cylinder

$$P_p = 2P_s$$
 (Eq. 1)

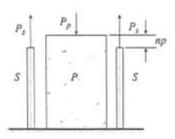


COMPATIBILITY EQUATION

 δ_s = elongation of steel bolt

 δ_p = shortening of plastic cylinder

$$\delta_s + \delta_p = np$$
 (Eq. 2)



FORCE-DISPLACEMENT RELATIONS

$$\delta_s = \frac{P_s L}{E_s A_s} \quad \delta_p = \frac{P_p L}{E_p A_p}$$
 (Eq. 3, Eq. 4)

SOLUTION OF EQUATIONS

Substitute (3) and (4) into Eq. (2):

$$\frac{P_s L}{E_s A_s} + \frac{P_p L}{E_p A_p} = np$$
 (Eq. 5)

Solve simultaneously Eqs. (1) and (5):

$$P_{p} = \frac{2 \, np \, E_{s} \, A_{s} \, E_{p} \, A_{p}}{L(E_{p} \, A_{p} \, + \, 2E_{s} \, A_{s})}$$

STRESS IN THE PLASTIC CYLINDER

$$\sigma_p = \frac{P_p}{A_p} = \frac{2 np E_s A_s E_p}{L(E_p A_p + 2E_s A_s)} \quad \leftarrow$$

SUBSTITUTE NUMERICAL VALUES:

$$N = E_s A_s E_p = 900 \times 10^9 \text{ lb}^2/\text{in.}^2$$

 $D = E_p A_p + 2E_s A_s = 4350 \times 10^3 \text{ lb}$

$$\sigma_p = \frac{2np}{L} \left(\frac{N}{D} \right) = \frac{2(1)(0.058 \text{in.})}{10 \text{ in.}} \left(\frac{N}{D} \right)$$

$$= 2400 \text{ psi} \qquad \longleftarrow$$

The figure shows a section through the sleeve, cap, and bolt.

NUMERICAL PROPERTIES

$$n = \frac{1}{2}$$
 $p = 1.0 \text{ mm}$ $\Delta T = 30^{\circ}\text{C}$

$$E_c = 120 \text{ GPa}$$
 $\alpha_c = 17 \times (10^{-6})^{\circ}\text{C}$

$$E_s = 200 \text{ GPa}$$
 $\alpha_s = 12 \times (10^{-6})^{\circ}\text{C}$

$$\tau_{ai} = 18.5 \text{ MPa}$$
 $s = 26 \text{ mm}$ $d_b = 5 \text{ mm}$

$$L_1 = 40 \text{ mm}$$
 $t_1 = 4 \text{ mm}$ $L_2 = 50 \text{ mm}$ $t_2 = 3 \text{ mm}$

$$d_1 = 25 \text{ mm}$$
 $d_1 - 2t_1 = 17 \text{ mm}$ $d_2 = 17 \text{ mm}$

$$A_b = \frac{\pi}{4}d_b^2$$
 $A_1 = \frac{\pi}{4}[d_1^2 - (d_1 - 2t_1)^2]$

$$A_b = 19.635 \text{ mm}^2$$
 $A_1 = 263.894 \text{ mm}^2$

$$A_2 = \frac{\pi}{4} [d_2^2 - (d_2 - 2t_2)^2]$$
 $A_2 = 131.947 \text{ mm}^2$

(a) Forces in Sleeve and Bolt One-degree statically indeterminate—cut bolt and use force in bolt (P_R) as redundant (see sketches):

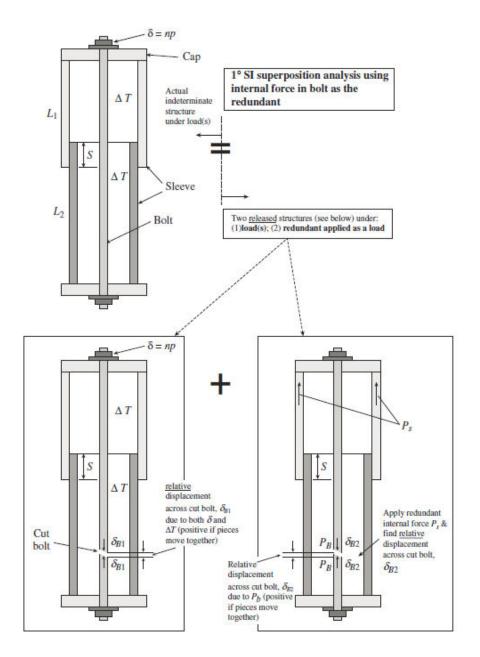
$$\delta_{B1} = -np + \alpha_s \Delta T (L_1 + L_2 - s)$$

$$\delta_{B2} = P_B \left[\frac{L_1 + L_2 - s}{E_s A_b} + \frac{L_1 - s}{E_c A_1} + \frac{L_2 - s}{E_c A_2} + \frac{s}{E_c (A_1 + A_2)} \right]$$

Compatibility: $\delta_{B1} + \delta_{B2} = 0$

$$P_{B} = \frac{-[-np + \alpha_{s}\Delta T(L_{1} + L_{2} - s)]}{\left[\frac{L_{1} + L_{2} - s}{E_{s}A_{b}} + \frac{L_{1} - s}{E_{c}A_{1}} + \frac{L_{2} - s}{E_{c}A_{2}} + \frac{s}{E_{c}(A_{1} + A_{2})}\right]} \quad P_{B} = 25.4 \text{ kN} \quad \longleftarrow \quad P_{s} = -P_{B} \quad \longleftarrow$$

Sketches illustrating superposition procedure for statically-indeterminate analysis



(b) REQUIRED LENGTH OF SOLDER JOINT≈

$$au=rac{P}{A_s}$$
 $A_s=\pi d_2 s$
$$s_{\mathrm{reqd}}=rac{P_B}{\pi d_2 au_{aj}}$$
 $s_{\mathrm{reqd}}=25.7~\mathrm{mm}$

(c) Final elongation

 δ_f = net of elongation of bolt (δ_b) and shortening of sleeve (δ_s)

$$\delta_b = P_B \left(\frac{L_1 + L_2 - s}{E_s A_b} \right) \qquad \delta_b = 0.413 \text{ mm}$$

$$\delta_s = P_s \left[\frac{L_1 - s}{E_c A_1} + \frac{L_2 - s}{E_c A_2} + \frac{s}{E_c (A_1 + A_2)} \right]$$

$$\delta_s = -0.064 \text{ mm}$$

$$\delta_f = \delta_b + \delta_s \qquad \delta_f = 0.35 \text{ mm} \qquad \longleftarrow$$

The figure shows a section through the tube, cap, and spring.

Properties and dimensions:

$$d_o = 6 \text{ in.}$$
 $t = \frac{1}{8} \text{ in.}$ $E_t = 100 \text{ ksi}$

$$A_t = \frac{\pi}{4} [d_o^2 - (d_o - 2t)^2]$$
 $A_t = 2.307 \text{ in.}^2$

$$L_1 = 12.125 \text{ in.} > L = 12 \text{ in.}$$
 $k = 1.5 \text{ k/in.}$

Spring is 1/8 in. longer than tube

$$\delta = L_1 - L$$
 $\delta = 0.125$ in.

$$\alpha_k = 6.5(10^{-6})/{}^{\circ}F$$
 < $\alpha_t = 80 \times (10^{-6})/{}^{\circ}F$

$$\Delta T = 0$$
 < note that Q result below is for zero temperature (until part(d))

(a) Force in spring F_{κ} = redundant Q

Flexibilities:
$$f = \frac{1}{k}$$
 $f_t = \frac{L}{E_t A_t}$

 δ_2 = relative displacement across cut spring due to redundant = $Q(f + f_t)$

 δ_1 = relative displacement across cut spring due to precompression and $\Delta T = \delta + \alpha_k \Delta T L_1 - \alpha_t \Delta T L$

Compatibility: $\delta_1 + \delta_2 = 0$

Solve for redundant Q:

$$Q = \frac{-\delta + \Delta T(-\alpha_k L_1 + \alpha_i L)}{f + f_t} = F_k$$

$$F_k = -0.174 \text{ k} \leftarrow \text{compressive force in}$$

 compressive force in spring (F_k) and also tensile force in tube

(b) $F_t = \text{force in tube} = -Q \longleftarrow$

NOTE: If tube is rigid, $F_k = -k\delta = -0.1875 \text{ k}$

(c) Final length of tube

$$L_f = L + \delta_{c1} + \delta_{c2}$$
 < i.e., add displacements
for the two released structures
to initial tube length L

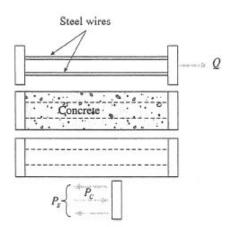
$$L_f = L - Qf_t + \alpha_t(\Delta T)L$$
 $L_f = 12.01$ in.

(d) Set Q=0 to find ΔT required to reduce spring force to zero

$$\Delta T_{\rm reqd} = \frac{\delta}{(-\alpha_k L_1 + \alpha_t L)}$$

$$\Delta T_{\text{regd}} = 141.9^{\circ}\text{F}$$

Since $\alpha_t > \alpha_k$, a temp. increase is req'd to expand tube so that spring force goes to zero.



EQUILIBRIUM EQUATION

$$P_s = P_c$$
 (Eq. 1)
Compatibility equation and force-displacement relations

 δ_1 = initial elongation of steel wires

$$=\frac{QL}{E_s A_s} = \frac{\sigma_0 L}{E_s}$$

 δ_2 = final elongation of steel wires

$$=\frac{P_sL}{E_sA_s}$$

 δ_3 = shortening of concrete

$$= \frac{P_c L}{E_c A_c}$$

$$\delta_1 - \delta_2 = \delta_3$$
 or $\frac{\sigma_0 L}{E_s} - \frac{P_s L}{E_s A_s} = \frac{P_c L}{E_c A_c}$

Solve simultaneously Eqs. (1) and (3):

$$P_s = P_c = \frac{\sigma_0 A_s}{1 + \frac{E_s A_s}{E_c A_c}}$$

$$L = length$$

 σ_0 = initial stress in wires

$$= \frac{Q}{A_s} = 620 \text{ MPa}$$

 A_s = total area of steel wires

 A_c = area of concrete

$$= 50 A_s$$

$$E_s = 12 E_c$$

 P_s = final tensile force in steel wires

 P_c = final compressive force in concrete

STRESSES

$$\sigma_s = \frac{P_s}{A_s} = \frac{\sigma_0}{1 + \frac{E_s A_s}{E_c A_c}} \leftarrow$$

$$\sigma_c = \frac{P_c}{A_c} = \frac{\sigma_0}{\frac{A_c}{A_s} + \frac{E_s}{E_c}} \quad \leftarrow$$

SUBSTITUTE NUMERICAL VALUES:

$$\sigma_0 = 620 \text{ MPa}$$
 $\frac{E_s}{E_c} = 12 \frac{A_s}{A_c} = \frac{1}{50}$

$$\sigma_s = \frac{620 \text{ MPa}}{1 + \frac{12}{50}} = 500 \text{ MPa (Tension)} \quad \leftarrow$$

$$\sigma_c = \frac{620 \text{ MPa}}{50 + 12} = 10 \text{ MPa (Compression)} \leftarrow$$

(Eq. 2, Eq. 3)

The figure shows a section through the tube, cap, and spring.

Properties and dimensions:

$$d_o = 6 \text{ in.}$$
 $t = \frac{1}{8} \text{ in.}$ $E_t = 100 \text{ ksi}$
 $L = 12 \text{ in.} > L_1 = 11.875 \text{ in.}$ $k = 1.5 \text{ k/in.}$
 $\alpha_k = 6.5(10^{-6}) < \alpha_t = 80 \times (10^{-6})$
 $A_t = \frac{\pi}{4} [d_o^2 - (d_o - 2t)^2]$
 $A_t = 2.307 \text{ in.}^2$

 $F_k = 0.174 \text{ k}$ also the compressive force in the

- (b) Force in tube $F_t = -Q = -0.174 \,\mathrm{k}$
- (c) Final length of tube and spring $L_{\!f} = L + \delta_{c1} + \delta_{c2}$

$$L_f = L - Qf_t + \alpha_t(\Delta T)L$$
 $L_f = 11.99$ in.

Pretension and temperature: Spring is 1/8 in. shorter than tube.

 $\delta = L - L_1$ $\delta = 0.125$ in. Δ

Note that Q result below is for zero temperature (until part (d)).

Flexibilities: $f = \frac{1}{k}$ $f_t = \frac{L}{E_t A_t}$

(a) Force in spring (F_k) = redundant (Q)

Follow solution procedure outlined in Prob. 2.5-29 solution:

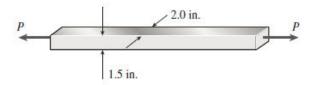
$$Q = \frac{\delta + \Delta T(-\alpha_k L_1 + \alpha_t L)}{f + f_t} = F_k$$

(d) Set Q=0 to find ΔT required to reduce spring force to zero

$$\Delta T_{\text{reqd}} = \frac{-\delta}{(-\alpha_k L_1 + \alpha_t L)}$$

$$\Delta T_{\text{read}} = -141.6^{\circ}\text{F}$$

Since $\alpha_t > \alpha_k$, a temperature drop is required to shrink tube so that spring force goes to zero.



NUMERICAL DATA

$$A = 3 \text{ in.}^2$$
 $\sigma_a = 14500 \text{ psi}$

$$\tau_a = 7100 \text{ psi}$$

MAXIMUM LOAD-TENSION

$$P_{\text{max}1} = \sigma_a A$$
 $P_{\text{max}1} = 43500 \text{ lbs}$

MAXIMUM LOAD—SHEAR

$$P_{\text{max}2} = 2\tau_a A$$
 $P_{\text{max}2} = 42,600 \text{ lbs}$

Because $\tau_{\rm allow}$ is less than one-half of $\sigma_{\rm allow}$, the shear stress governs.



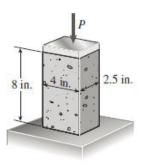
Numerical data
$$P=3.5~\mathrm{kN}$$
 $\sigma_a=118~\mathrm{MPa}$ $\tau_a=48~\mathrm{MPa}$

Find $P_{\rm max}$ then rod diameter. since τ_a is less than 1/2 of σ_a , shear governs.

$$P_{\text{max}} = 2\tau_a \left(\frac{\pi}{4} d_{\text{min}}^2\right)$$

$$d_{\min} = \sqrt{\frac{2}{\pi \tau_a} P}$$

$$d_{\min} = 6.81 \text{ mm} \quad \leftarrow$$



 $A = 2.5 \text{ in.} \times 4.0 \text{ in.} = 10.0 \text{ in.}^2$ Maximum normal stress:

$$\sigma_x = \frac{P}{A}$$

Maximum shear stress:

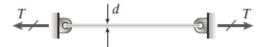
$$\tau_{\text{max}} = \frac{\sigma_x}{2} = \frac{P}{2A}$$

$$\sigma_{\rm ult} = 3600 \ {\rm psi}$$
 $\tau_{\rm ult} = 1200 \ {\rm psi}$

Because $au_{
m ult}$ is less than one-half of $\sigma_{
m ult}$, the shear stress governs.

$$\tau_{\text{max}} = \frac{P}{2A}$$
 or $P_{\text{max}} = 2A\tau_{\text{ult}}$

$$P_{\rm max} = 2(10.0 \ {\rm in.^2})(1200 \ {\rm psi}) = 24,000 \ {\rm lb} \quad \leftarrow$$



NUMERICAL DATA

$$d = 2.42 \text{ mm}$$
 $T = 98 \text{ N}$
 $\alpha = 19.5 (10^{-6}) / ^{\circ}\text{C}$ $E = 110 \text{ GPa}$

(a) $\Delta T_{\rm max}$ (drop in temperature)

$$\sigma = \frac{T}{A} - (E\alpha \Delta T) \qquad \tau_{\text{max}} = \frac{\sigma}{2}$$

$$\tau_a = \frac{T}{2A} - \frac{E\alpha \Delta T}{2}$$

$$au_a=60~\mathrm{MPa}$$
 $A=rac{\pi}{4}d^2$
$$\Delta T_{\mathrm{max}}=rac{T}{A}-2\, au_a$$

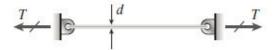
$$\Delta T_{\mathrm{max}}=-46^{\circ}\mathrm{C}~(\mathrm{drop})$$

(b) ΔT at which wire goes slack Increase ΔT until $\sigma=0$: $\Delta T=\frac{T}{E\,\alpha A}$

 $\Delta T = 9.93^{\circ}$ C (increase)

CLICK HERE TO ACCESS THE COMPLETE Solutions

Problem 2.6-5



NUMERICAL DATA

$$d = \frac{1}{16}$$
 in. $T = 37$ lb $\alpha = 10.6 \times (10^{-6})^{6}$ F

$$E = 15 \times (10^6) \text{ psi}$$
 $\Delta T = -60^\circ \text{F}$
 $A = \frac{\pi}{4} d^2$

(a) au_{max} (due to drop in temperature)

$$\tau_{\text{max}} = \frac{\sigma_x}{2}$$
 $\tau_{\text{max}} = \frac{\frac{T}{A} - (E \alpha \Delta T)}{2}$
 $\tau_{\text{max}} = 10,800 \text{ psi}$

(b) $\Delta T_{\rm max}$ for allowable shear stress $au_a = 10000~{
m psi}$

$$\Delta T_{\text{max}} = \frac{\frac{T}{A} - 2\tau_a}{E\alpha}$$

$$\Delta T_{\text{max}} = -49.9^{\circ} \text{F} \quad \longleftarrow$$

(c) ΔT AT WHICH WIRE GOES SLACK Increase ΔT until $\sigma = 0$:

$$\Delta T = \frac{T}{E \alpha A}$$

$$\Delta T = 75.9^{\circ} \text{F (increase)} \quad \longleftarrow$$

CLICK HERE TO ACCESS THE COMPLETE Solutions

Problem 2.6-6

(a)
$$d = 12 \text{ mm}$$
 $P = 9.5 \text{ kN}$ $A = \frac{\pi}{4}d^2 = 1.131 \times 10^{-4} \text{ m}^2$

$$\sigma_x = \frac{P}{A} = 84 \text{ MPa}$$

(b)
$$\tau_{\text{max}} = \frac{\sigma_x}{2} = 42 \text{ MPa}$$
 On plane stress element rotated 45°

(c) Rotated stress element (45°) has normal tensile stress $\sigma_x/2$ on all faces, $-T_{\max}$ (CW) on +x-face, and $+T_{\text{max}}$ (CCW) on + y-face

$$\tau_{xy|y|} = \tau_{max}$$
 $\sigma_{x|} = \frac{\sigma_x}{2}$
 $\sigma_{y|} = \sigma_{x|}$

On rotated x-face:
$$\sigma_{x1} = 42 \text{ MPa}$$

On rotated x-face:
$$\sigma_{x1} = 42 \text{ MPa}$$
 $\sigma_{x1y1} = 42 \text{ MPa}$ On rotated y-face: $\sigma_{y1} = 42 \text{ MPa}$

(d) $\theta = 22.5^{\circ}$ < CCW rotation of element

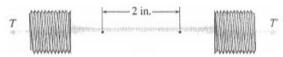
$$\sigma_{\theta} = \sigma_x \cos(\theta)^2 = 71.7 \text{ MPa}$$
 < on rotated x face $\sigma_y = \sigma_x \cos\left(\theta + \frac{\pi}{2}\right)^2 = 12.3 \text{ MPa}$ < on rotated y face

Eq. 2-29b
$$au_{\theta} = \frac{-\sigma_x}{2} \sin(2\theta) = -29.7 \text{ MPa}$$
 < CW on rotated x-face

On rotated x-face:
$$\sigma_{x1} = 71.7 \text{ MPa}$$
 $\sigma_{x1} = 71.7 \text{ MPa}$ $\sigma_{y1} = 12.3 \text{ MPa}$ $\sigma_{y1} = 12.3 \text{ MPa}$

$$\tau_{x|y|} = -29.7 \text{ MPa}$$

d y-face:
$$\sigma_{yl} = 12.3 \text{ MPa}$$



Elongation: $\delta = 0.00120$ in.

(2 in. gage length)

Strain:
$$\varepsilon = \frac{\delta}{L} = \frac{0.00120 \text{ in.}}{2 \text{ in.}} = 0.00060$$

Hooke's law:
$$\sigma_x = E\varepsilon = (30 \times 10^6 \text{ psi})(0.00060)$$

= 18,000 psi

(a) MAXIMUM NORMAL STRESS

 σ_x is the maximum normal stress.

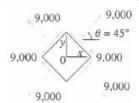
$$\sigma_{\text{max}} = 18,000 \text{ psi} \quad \leftarrow$$

(b) MAXIMUM SHEAR STRESS

The maximum shear stress is on a 45° plane and equals $\sigma_x/2$.

$$\tau_{\text{max}} = \frac{\sigma_x}{2} = 9,000 \text{ psi}$$

(c) Stress element at $\theta = 45^{\circ}$



NOTE: All stresses have units of psi.

(a)
$$\alpha = 17.5 \left(10^{-6}\right)$$
 $\Delta T = 50$ $E = 120$ GPa
$$\sigma_x = -E\alpha \Delta T = -105 \text{ MPa} \qquad \tau_{\text{max}} = \frac{\sigma_x}{2} = -52.5 \text{ MPa} \qquad < \text{at } \theta = 45^\circ$$
 (compression) Element $A: \sigma_x = 105 \text{ MPa}$ (compression); Element $B: \tau_{\text{max}} = 52.5 \text{ MPa}$

(b)
$$\tau_{\theta} = 48 \text{ MPa}$$
 Eq. 2-29b
$$\tau_{\theta} = \frac{-\sigma_{x}}{2} \sin(2\theta)$$

so
$$\theta = \frac{1}{2}a\sin\left(\frac{2\tau_{\theta}}{-\sigma_{x}}\right) = 33.1^{\circ} < \text{CCW rotation of element}$$
 $\theta = 33.1^{\circ}$

$$\sigma_{\theta} = \sigma_{x}\cos(\theta)^{2} = -73.8 \text{ MPa} < \text{on rotated } x \text{ face}$$

$$\sigma_{y} = \sigma_{x}\cos\left(\theta + \frac{\pi}{2}\right)^{2} = -31.2 \text{ MPa} < \text{on rotated } y \text{ face}$$

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Problem 2.6-9

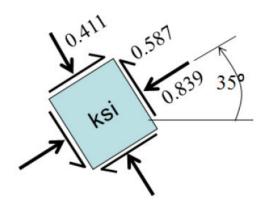
$$P = 10 \text{kip}$$
 $L = 3 \text{ft}$ $A = 8 \cdot \text{in}^2$ $\theta = 35 \text{deg}$

Normal compressive stress
$$\sigma_{x} = \frac{-p}{A} = -1.25 \cdot ksi$$

$$\sigma_{\theta}(\theta) = \sigma_{x} \cdot \cos(\theta)^{2} \qquad \qquad \sigma_{\theta}(\theta) = -0.839 \cdot ksi \qquad \qquad \sigma_{\theta}\left(\theta + \frac{\pi}{2}\right) = -0.411 \cdot ksi$$

$$\tau_{\theta}(\theta) = -\sigma_{\mathbf{x}} \cdot \sin(\theta) \cdot \cos(\theta) \qquad \qquad \tau_{\theta}(\theta) = 0.587 \cdot \mathrm{ksi} \qquad \qquad \tau_{\theta}\left(\theta + \frac{\pi}{2}\right) = -0.587 \cdot \mathrm{ksi}$$

Rotated stress element



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Problem 2.6-10

$$L = 1m$$
 $A = 1200mm^2$ $\Delta T = 25$ $\alpha = 12 \cdot (10^{-6})$ $\theta = 45deg$ $E = 200GPa$

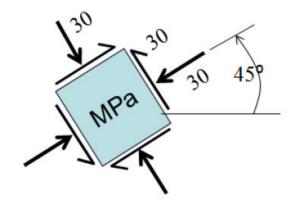
Compressive thermal stress
$$\sigma_T = E \cdot \alpha \cdot \Delta T = 60 \cdot MPa$$

Support reactions
$$R_A = \sigma_{T} \cdot A = 72 \cdot kN$$
 $R_B = -R_A$

Plane stress transformations
$$\sigma_{_{X}} = \frac{R_{B}}{_{\Delta}} = -60 \cdot \text{MPa}$$

$$\sigma_{\pmb{\theta}} = \sigma_{\pmb{x}} \cdot \cos(\pmb{\theta})^2 = -30 \cdot \text{MPa} \qquad \qquad \sigma_{\pmb{x}} \cdot \cos\left(\pmb{\theta} + \frac{\pi}{2}\right)^2 = -30 \cdot \text{MPa} \qquad \qquad \tau_{\pmb{\theta}} = -\sigma_{\pmb{x}} \cdot \sin(\pmb{\theta}) \cdot \cos(\pmb{\theta}) = 30 \cdot \text{MPa}$$

Rotated stress element



NUMERICAL DATA

$$L = 10 \text{ ft}$$
 $b = 0.71 L$ $P = 49 \text{ k}$ $\sigma_a = 14 \text{ ksi}$ $\tau_a = 7.5 \text{ ksi}$ $A = 5.87 \text{ in.}^2$

(a) FOR LINEAR ANALYSIS, MEMBER FORCES ARE PROPORTIONAL TO LOADING

(solution for
$$P=35$$
 k)
$$F_{AC}=\frac{P}{35}\,15.59=21.826\,\mathrm{k} \qquad F_{AB}=\frac{P}{35}\,62.2=87.08\,\mathrm{k}$$

$$F_{BC}=\frac{P}{35}\,(-78.9) \qquad F_{BC}=-110.46\,\mathrm{k}$$
 Normal stresses in each member:
$$\sigma_{AC}=\frac{F_{AC}}{A}=3.718\,\mathrm{ksi} \qquad \sigma_{AB}=\frac{F_{AB}}{A}=14.835\,\mathrm{ksi}$$
 From Eq. 2-31:
$$\sigma_{BC}=\frac{F_{BC}}{A}=-18.818\,\mathrm{ksi}$$

$$\tau_{\text{max}AC} = \frac{\sigma_{AC}}{2} = 1.859 \text{ ksi}$$
 $\tau_{\text{max}AB} = \frac{\sigma_{AB}}{2} = 7.42 \text{ ksi}$

$$\tau_{\text{max}BC} = \frac{\sigma_{BC}}{2} = -9.41 \text{ ksi}$$

(b) $\sigma_a < 2 \times T_a$ so normal stress will control; lowest value governs here

Member AC:
$$P_{\text{max}\sigma} = \frac{P}{F_{AC}}(\sigma_a A) = 184.496 \text{ k}$$
 $P_{\text{max}\tau} = \frac{P}{F_{AC}}(2\tau_a A) = 197.675 \text{ k}$ Member AB: $P_{\text{max}\sigma} = \frac{P}{F_{AB}}(\sigma_a A) = 46.243 \text{ k}$ $P_{\text{max}\tau} = \frac{P}{F_{AB}}(2\tau_a A) = 49.546 \text{ k}$ Member BC: $P_{\text{max}\sigma} = \left|\frac{P}{F_{BC}}\right|(\sigma_a A) = 36.5 \text{ k}$ $P_{\text{max}\tau} = \left|\frac{P}{F_{BC}}\right|(2\tau_a A) = 39.059 \text{ k}$

NUMERICAL DATA

$$d = 32 \text{ mm} \qquad \qquad A = \frac{\pi}{4} d^2$$

$$P = 190 \text{ N}$$
 $A = 804.25 \text{ mm}^2$

$$a = 100 \, \text{mm}$$

$$b = 300 \, \text{mm}$$

(a) STATICS—FIND COMPRESSIVE FORCE F AND STRESSES IN PLASTIC BAR

$$F = \frac{P(a+b)}{a} \qquad F = 760 \,\mathrm{N}$$

$$\sigma_x = \frac{F}{A}$$
 $\sigma_x = 0.945 \text{ MPa}$ or $\sigma_x = 945 \text{ kPa}$

From (1), (2), and (3) below:

$$\sigma_{\text{max}} = \sigma_x$$
 $\sigma_{\text{max}} = -945 \text{ kPa}$

$$\tau_{\text{max}} = 472 \text{ kPa}$$
 $\frac{\sigma_x}{2} = -472 \text{ kPa}$

(1)
$$\theta = 0^{\circ}$$
 $\sigma_x = -945 \text{ kPa}$ \leftarrow

(2)
$$\theta = 22.50^{\circ}$$

On
$$+x$$
-face:

$$\sigma_{\theta} = \sigma_{x} \cos(\theta)^{2}$$
 $\sigma_{\theta} = -807 \text{ kPa} \quad \longleftarrow$

$$\tau_{\theta} = -\sigma_{x} \sin(\theta) \cos(\theta)$$

$$\tau_{\theta} = 334 \text{ kPa} \leftarrow$$

On +y-face:
$$\theta = \theta + \frac{\pi}{2}$$

$$\sigma_{\theta} = \sigma_{x} \cos(\theta)^{2}$$

$$\sigma_{\theta} = -138.39 \text{ kPa}$$

$$\tau_{\theta} = -\sigma_x \sin(\theta) \cos(\theta)$$

 $\tau_{\theta} = -334.1 \text{ kPa}$

(3)
$$\theta = 45^{\circ}$$

On +x-face:

$$\sigma_{\theta} = \sigma_{x} \cos(\theta)^{2}$$
 $\sigma_{\theta} = -472 \text{ kPa} \longleftrightarrow$

$$\tau_{\theta} = -\sigma_{x} \sin(\theta) \cos(\theta)$$

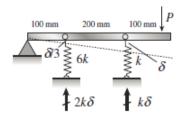
$$\tau_{\theta} = 472 \text{ kPa} \longleftrightarrow$$

On +y-face:
$$\theta = \theta + \frac{\pi}{2}$$

$$\sigma_{\theta} = \sigma_{x} \cos(\theta)^{2}$$
 $\sigma_{\theta} = -472.49 \text{ kPa}$

$$\tau_{\theta} = -\sigma_x \sin(\theta) \cos(\theta)$$
 $\tau_{\theta} = -472.49 \text{ kPa}$

(b) ADD SPRING—FIND MAXIMUM NORMAL AND SHEAR STRESSES IN PLASTIC BAR



$$\sum M_{\rm pin} = 0$$

$$P(400) = [2k\delta(100) + k\delta(300)]$$

$$\delta = \frac{4}{5} \frac{P}{k}$$

Force in plastic bar: $F = (2k) \left(\frac{4}{5} \frac{P}{k} \right)$ $F = \frac{8}{5} P \qquad F = 304 \text{ N}$

Normal and shear stresses in plastic bar:

$$\sigma_x = \frac{F}{A}$$
 $\sigma_x = 0.38$ $\sigma_{\text{max}} = -378 \text{ kPa}$ \leftarrow

$$\tau_{\text{max}} = \frac{\sigma_x}{2}$$
 $\tau_{\text{max}} = -189 \,\text{kPa}$ \leftarrow

NUMERICAL DATA

$$b = 1.5 \text{ in.}$$
 $h = 3 \text{ in.}$ $A = bh$ $\Delta T = (160 - 68)^{\circ} \text{F}$

$$\Delta T = 92^{\circ}F$$

$$A = 4.5 \text{ in.}^2$$
 $\sigma_{pq} = -1700 \text{ psi}$

$$\alpha = 60 \times (10^{-6})^{\circ} \text{F}$$

$$E = 450 \times (10^3) \text{ psi}$$

(a) SHEAR STRESS ON PLANE PQ Statically indeterminate analysis gives, for reaction at right support:

$$R = -EA\alpha\Delta T$$
 $R = -11178$ lb

$$R = -11178 \text{ lb}$$

$$\sigma_x = \frac{R}{\Lambda}$$
 $\sigma_x = -2484 \text{ psi}$

Using $\sigma_{\theta} = \sigma_{x} \cos(\theta)^{2}$: $\cos(\theta)^{2} = \frac{\sigma_{pq}}{\sigma_{rq}}$

$$\theta = a\cos\left(\sqrt{\frac{\sigma_{pq}}{\sigma_x}}\right)$$
 $\theta = 34.2^{\circ}$

Now with θ , can find shear stress on plane pq:

$$\tau_{pq} = -\sigma_x \sin(\theta)\cos(\theta)$$
 $\tau_{pq} = 1154 \text{ psi}$

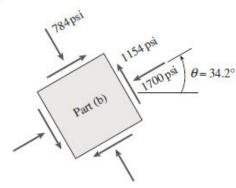
$$\tau_{pq} = 1154 \text{ psi}$$

$$\sigma_{pq} = \sigma_x \cos(\theta)^2$$
 $\sigma_{pq} = -1700 \text{ psi}$

Stresses at $\theta + \pi/2$ (y-face):

$$\sigma_y = \sigma_x cos \left(\theta + \frac{\pi}{2}\right)^2$$
 $\sigma_y = -784 \text{ psi}$

(b) STRESS ELEMENT FOR PLANE PQ



(c) Maximum load at quarter point $\sigma_a = 3400 \text{ psi}$

$$\tau_a = 1650 \text{ psi}$$

$$2\tau_a = 3300$$

< less than σ_a ,

so shear controls

Statically indeterminate analysis for P at L/4 gives for reactions:

$$R_{R2} = \frac{-P}{4}$$
 $R_{L2} = \frac{-3}{4}P$

(tension for 0 to L/4 and compression for rest of bar)

From part (a) (for temperature increase ΔT):

$$R_{R1} = -EA\alpha\Delta T$$
 $R_{L1} = -EA\alpha\Delta T$

Stresses in bar (0 to L/4):

$$\sigma_x = -E\alpha\Delta T + \frac{3P}{4A}$$
 $\tau_{\text{max}} = \frac{\sigma_x}{2}$

Set $\tau_{\text{max}} = \tau_a$ and solve for $P_{\text{max}1}$:

$$\tau_a = \frac{-E\alpha\Delta T}{2} + \frac{3P}{8A}$$

$$P_{\text{max}1} = \frac{4A}{3}(2\tau_a + E\alpha\Delta T)$$

$$P_{\text{max}1} = 34,704 \text{ lb}$$

$$\tau_{\text{max}} = \frac{-E\alpha\Delta T}{2} + \frac{3P_{\text{max}1}}{8A}$$

$$\tau_{\text{max}} = 1650 \text{ psi}$$
 < check

$$\sigma_x = -E\alpha\Delta T + \frac{3P_{\text{max}1}}{4A}$$

$$\sigma_{\rm r} = 3300 \, \rm psi$$
 < less than σ_a

Stresses in bar (L/4 to L):

$$\sigma_x = -E\alpha\Delta T - \frac{P}{4A}$$
 $\tau_{\text{max}} = \frac{\sigma_x}{2}$

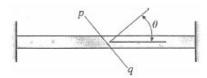
Set $\tau_{\text{max}} = \tau_a$ and solve for $P_{\text{max}2}$:

$$P_{\text{max}2} = -4A(-2\tau_a + E\alpha\Delta T)$$

$$P_{\text{max2}} = 14,688 \text{ lb} \leftarrow \text{shear in segment } (L/4 \text{ to } L) \text{ controls}$$

$$au_{\text{max}} = \frac{-E \alpha \Delta T}{2} - \frac{P_{\text{max}2}}{8A} \quad au_{\text{max}} = -1650 \text{ psi}$$

$$\sigma_x = -E\alpha\Delta T - \frac{P_{\text{max}2}}{4A}$$
 $\sigma_x = -3300 \text{ psi}$



NUMERICAL DATA

$$\theta = 55 \left(\frac{\pi}{180}\right) \text{ rad}$$

$$b = 18 \text{ mm} \qquad h = 40 \text{ mm}$$

$$A = bh \qquad A = 720 \text{ mm}^2$$

$$\sigma_{pqa} = 60 \text{ MPa} \qquad \tau_{pqa} = 30 \text{ MPa}$$

$$\alpha = 17 \times (10^{-6})^{\circ}\text{C} \qquad E = 120 \text{ GPa}$$

$$\Delta T = 20^{\circ}\text{C} \qquad P = 15 \text{ kN}$$

(a) Find $\Delta T_{
m max}$ based on allowable normal and shear stress values on plane pq

$$\sigma_x = -E\alpha\Delta T_{\text{max}}$$
 $\Delta T_{\text{max}} = \frac{-\sigma_x}{E\alpha}$
 $\sigma_{pq} = \sigma_x \cos(\theta)^2$ $\tau_{pq} = -\sigma_x \sin(\theta)\cos(\theta)$
Set each equal to corresponding allowable and solve for σ_x :
$$\sigma_{x1} = \frac{\sigma_{pqa}}{\cos(\theta)^2}$$
 $\sigma_{x1} = 182.38 \text{ MPa}$

$$\sigma_{x2} = \frac{\tau_{pqa}}{-\sin(\theta)\cos(\theta)}$$
 $\sigma_{x2} = -63.85 \text{ MPa}$

Lesser value controls, so allowable shear stress governs.

$$\Delta T_{\text{max}} = \frac{-\sigma_{x2}}{E\alpha}$$
 $\Delta T_{\text{max}} = 31.3^{\circ}\text{C} \leftarrow$

(b) STRESSES ON PLANE PQ FOR MAXIMUM TEMPERATURE

$$\sigma_x = -E\alpha\Delta T_{\rm max}$$
 $\sigma_x = -63.85 \text{ MPa}$

$$\sigma_{pq} = \sigma_x \cos(\theta)^2 \qquad \sigma_{pq} = -21.0 \text{ MPa} \quad \leftarrow$$

$$\tau_{pq} = -\sigma_x \sin(\theta) \cos(\theta) \qquad \tau_{pq} = 30 \text{ MPa} \quad \leftarrow$$

(c) ADD LOAD P IN +x-DIRECTION TO TEMPERATURE CHANGE AND FIND LOCATION OF LOAD

$$\Delta T = 28^{\circ}C$$

P = 15 kN from one-degree statically indeterminate analysis, reactions R_A and R_B due to load P:

$$R_A = -(1 - \beta)P$$
 $R_B = \beta P$
Now add normal stresses due to P to thermal stresses due to ΔT (tension in segment 0 to βL , compression in segment βL to L).

Stresses in bar (0 to βL):

$$\sigma_x = -E\alpha\Delta T + \frac{R_A}{A}$$
 $\tau_{\text{max}} = \frac{\sigma_x}{2}$

Shear controls so set $\tau_{\text{max}} = \tau_a$ and solve for β :

$$2\tau_a = -E\alpha\Delta T + \frac{(1-\beta)P}{A}$$

$$\beta = 1 - \frac{A}{P}[2\tau_a + E\alpha\Delta T]$$

$$\beta = -5.1$$
 Impossible so evaluate segment (\beta L to L):

Stresses in bar $(\beta L \text{ to } L)$:

$$\sigma_x = -E\alpha\Delta T - \frac{R_B}{A}$$
 $\tau_{\text{max}} = \frac{\sigma_x}{2}$

set $\tau_{\text{max}} = \tau_a$ and solve for $P_{\text{max}2}$

$$2\tau_a = -E\alpha\Delta T - \frac{\beta P}{A}$$
$$\beta = \frac{-A}{P} \left[-2\tau_a + E\alpha\Delta T \right]$$
$$\beta = 0.62 \leftarrow$$

NUMERICAL DATA

$$P = 5 \text{ k}$$
 $\alpha = 36^{\circ}$ $\sigma_a = 13.5 \text{ ksi}$

$$\tau_a = 6.5 \text{ ksi}$$

$$\theta = \frac{\pi}{2} - \alpha$$
 $\theta = 54^{\circ}$

$$\sigma_{ja} = 6.0 \text{ ksi}$$

$$\tau_{ia} = 3.0 \text{ ksi}$$

Tensile force N_{AC} using Method of Joints at C:

$$N_{AC} = \frac{P}{\sin(60^{\circ})}$$
 (tension)

$$N_{AC} = 5.77 \text{ k} \leftarrow$$

Minimum required diameter of bar AC:

(1) Check tension and shear in bars; $\tau_a < \sigma_a/2$ so shear

controls
$$\tau_{\text{max}} = \frac{\sigma_x}{2}$$
:

$$2\tau_a = \frac{N_{AC}}{A}$$
 $\sigma_x = 2\tau_a = 13$ ksi

$$A_{\rm reqd} = \frac{N_{AC}}{2\tau_a}$$
 $A_{\rm reqd} = 0.44 \text{ in.}^2$

$$d_{\min} = \sqrt{\frac{4}{\pi}} A_{\text{reqd}}$$
 $d_{\min} = 0.75 \text{ in.}$

(2) Check tension and shear on brazed joint:

$$\sigma_x = \frac{N_{AC}}{A}$$
 $\sigma_x = \frac{N_{AC}}{\frac{\pi}{4} d^2}$ $d_{\text{reqd}} = \sqrt{\frac{4}{\pi} \frac{N_{AC}}{\sigma_X}}$

Tension on brazed joint:

$$\sigma_{\theta} = \sigma_{r} \cos(\theta)^{2}$$

Set equal to σ_{ja} and solve for σ_x , then d_{regd} :

$$\sigma_x = \frac{\sigma_{ja}}{\cos(\theta)^2}$$
 $\sigma_x = 17.37 \text{ ksi}$

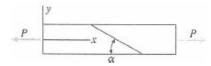
$$d_{\text{reqd}} = \sqrt{\frac{4}{\pi} \frac{N_{AC}}{\sigma_{\text{r}}}}$$
 $d_{\text{reqd}} = 0.65 \text{ in.}$

Shear on brazed joint:

$$\tau_{\theta} = -\sigma_x \sin(\theta) \cos(\theta)$$

$$\sigma_x = \left| \frac{\tau_{ja}}{-(\sin(\theta)\cos(\theta))} \right|$$
 $\sigma_x = -6.31 \text{ ksi}$

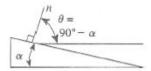
$$d_{\mathrm{reqd}} = \sqrt{\frac{4}{\pi} \frac{N_{AC}}{\sigma_X}}$$
 $d_{\mathrm{reqd}} = 1.08 \text{ in.}$ \leftarrow



$$10^{\circ} \le \alpha \le 40^{\circ}$$

Due to load P: $\sigma_x = 4.9 \text{ MPa}$

(a) Stresses on joint when $\alpha = 20^{\circ}$



$$\theta = 90^{\circ} - \alpha = 70^{\circ}$$

$$\sigma_{\theta} = \sigma_{x} \cos^{2}\theta = (4.9 \text{ MPa})(\cos 70^{\circ})^{2}$$

$$= 0.57 \text{ MPa} \qquad \longleftarrow$$

$$\tau_{\theta} = -\sigma_{x} \sin \theta \cos \theta$$

$$= (-4.9 \text{ MPa})(\sin 70^{\circ})(\cos 70^{\circ})$$

$$= -1.58 \text{ MPa} \qquad \longleftarrow$$

(b) Largest angle α if $\tau_{\rm allow} = 2.25$ MPa

$$\tau_{\rm allow} = -\sigma_x \sin \theta \cos \theta$$

The shear stress on the joint has a negative sign. Its numerical value cannot exceed $\tau_{allow} = 2.25$ MPa. Therefore.

$$-2.25 \text{ MPa} = -(4.9 \text{ MPa})(\sin \theta)(\cos \theta) \text{ or } \sin \theta \cos \theta = 0.4592$$

From trigonometry: $\sin\theta\cos\theta = \frac{1}{2}\sin 2\theta$

Therefore:
$$\sin 2\theta = 2(0.4592) = 0.9184$$

Solving: $2\theta = 66.69^{\circ}$ or 113.31°
 $\theta = 33.34^{\circ}$ or 56.66°
 $\alpha = 90^{\circ} - \theta$ $\therefore \alpha = 56.66^{\circ}$ or 33.34°

Since α must be between 10° and 40° , we select

$$\alpha = 33.3^{\circ} \leftarrow$$

NOTE: If α is between 10° and 33.3°,

$$|\tau_{\theta}| < 2.25 \text{ MPa}.$$

If α is between 33.3° and 40°,

$$|\tau_{\theta}| > 2.25 \text{ MPa}.$$

(c) What is α if $\tau_{\theta} = 2\sigma_{\theta}$?

Numerical values only:

$$|\tau_{\theta}| = \sigma_x \sin \theta \cos \theta$$
 $|\sigma_{\theta}| = \sigma_x \cos^2 \theta$

$$\left|\frac{\tau_0}{\sigma_0}\right| = 2$$

$$\sigma_x \sin \theta \cos \theta = 2\sigma_x \cos^2 \theta$$

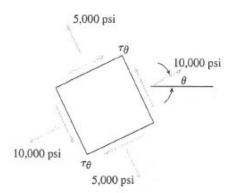
$$\sin \theta = 2 \cos \theta$$
 or $\tan \theta = 2$

$$\theta = 63.43^{\circ}$$
 $\alpha = 90^{\circ} - \theta$

$$\alpha = 26.6^{\circ} \leftarrow$$

NOTE: For $\alpha = 26.6^{\circ}$ and $\theta = 63.4^{\circ}$, we find $\sigma_{\theta} = 0.98$ MPa and $\tau_{\theta} = -1.96$ MPa.

Thus,
$$\left| \frac{\tau_0}{\sigma_0} \right| = 2$$
 as required.



(a) Angle θ and shear stress τ_{θ}

$$\sigma_{\theta} = \sigma_x \cos^2 \theta$$

 $\sigma_{\theta} = 10,000 \text{ psi}$

$$\sigma_x = \frac{\sigma_0}{\cos^2 \theta} = \frac{10,000 \text{ psi}}{\cos^2 \theta} \tag{1}$$

Plane at angle $\theta + 90^{\circ}$

$$\sigma_{\theta} + 90^{\circ} = \sigma_{x} [\cos(\theta + 90^{\circ})]^{2} = \sigma_{x} [-\sin \theta]^{2}$$
$$= \sigma_{x} \sin^{2} \theta$$

$$\sigma_{\theta} + 90^{\circ} = 5,000 \text{ psi}$$

$$\sigma_x = \frac{\sigma_{\theta+90}^{\circ}}{\sin^2 \theta} = \frac{5,000 \text{ psi}}{\sin^2 \theta}$$
 (2)

Equate (1) and (2):

$$\frac{10,000 \text{ psi}}{\cos^2 \theta} = \frac{5,000 \text{ psi}}{\sin^2 \theta}$$

$$\tan^2\theta = \frac{1}{2} \quad \tan\theta = \frac{1}{\sqrt{2}} \quad \theta = 35.26^{\circ} \quad \longleftarrow$$

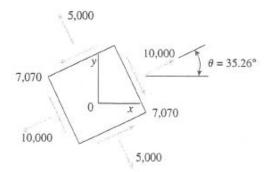
From Eq. (1) or (2):

$$\sigma_x = 15,000 \text{ psi}$$

$$\tau_{\theta} = -\sigma_x \sin \theta \cos \theta$$

$$= (-15,000 \text{ psi})(\sin 35.26^{\circ})(\cos 35.26^{\circ})$$

Minus sign means that τ_{θ} acts clockwise on the plane for which $\theta = 35.26^{\circ}$.

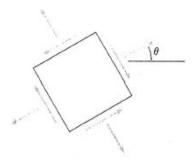


NOTE: All stresses have units of psi.

(b) MAXIMUM NORMAL AND SHEAR STRESSES

$$\sigma_{\text{max}} = \sigma_x = 15,000 \text{ psi} \quad \leftarrow$$

$$\tau_{\text{max}} = \frac{\sigma_x}{2} = 7,500 \text{ psi}$$



Find θ and σ_x for stress state shown in figure.

$$\sigma_{\theta} = \sigma_{x} \cos(\theta)^{2}$$
 $\cos(\theta) = \sqrt{\frac{\sigma_{\theta}}{\sigma_{x}}}$
so $\sin(\theta) = \sqrt{1 - \frac{\sigma_{\theta}}{\sigma_{x}}}$

$$\tau_{\theta} = -\sigma_x \sin(\theta) \cos(\theta)$$

$$\frac{\tau_{\theta}}{\sigma_{x}} = -\sqrt{1 - \frac{\sigma_{\theta}}{\sigma_{x}}} \sqrt{\frac{\sigma_{\theta}}{\sigma_{x}}}$$

$$\left(\frac{\tau_{\theta}}{\sigma_{x}}\right)^{2} = \frac{\sigma_{\theta}}{\sigma_{x}} - \left(\frac{\sigma_{\theta}}{\sigma_{x}}\right)$$

$$\left(\frac{23}{\sigma_x}\right)^2 = \frac{65}{\sigma_x} - \left(\frac{65}{\sigma_x}\right)^2$$

$$\left(\frac{65}{\sigma_x}\right)^2 - \left(\frac{65}{\sigma_x}\right) + \left(\frac{23}{\sigma_x}\right)^2 = 0$$

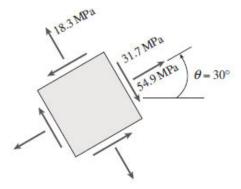
$$\frac{-(-4754 + 65\sigma_x)}{\sigma_x^2} = 0$$

$$4754$$

$$\sigma_x = \frac{4754}{65}$$

$$\sigma_x = 73.1 \text{ MPa}$$
 $\sigma_\theta = 65 \text{ MPa}$

$$\theta = a\cos\left(\sqrt{\frac{\sigma_{\theta}}{\sigma_{x}}}\right) \quad \theta = 19.5^{\circ}$$

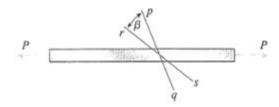


Now find σ_{θ} and τ_{θ} for $\theta = 30^{\circ}$:

$$\sigma_{\theta 1} = \sigma_x \cos(\theta)^2$$
 $\sigma_{\theta 1} = 54.9 \text{ MPa}$ \leftarrow

$$\tau_{\theta} = -\sigma_x \sin(\theta) \cos(\theta)$$
 $\tau_{\theta} = -31.7 \text{ MPa}$ \leftarrow

$$\sigma_{\theta 2} = \sigma_x \cos\left(\theta + \frac{\pi}{2}\right)^2$$
 $\sigma_{\theta 2} = 18.3 \text{ MPa}$ \leftarrow



Eq. (2-29a)

$$\sigma_{\theta} = \sigma_{x} \cos^{2} \theta$$

$$\beta = 30^{\circ}$$

PLANE
$$pq$$
: $\sigma_1 = \sigma_x \cos^2 \theta_1$ $\sigma_1 = 7500 \text{ psi}$

$$\sigma_1 = 7500 \text{ ps}$$

Plane rs:
$$\sigma_2 = \sigma_x \cos^2(\theta_1 + \beta)$$
 $\sigma_2 = 2500 \text{ psi}$

$$\sigma_2 = 2500 \text{ psi}$$

Equate σ_x from σ_1 and σ_2 :

$$\sigma_x = \frac{\sigma_1}{\cos^2 \theta_1} = \frac{\sigma_2}{\cos^2 (\theta_1 + \beta)}$$
 (Eq. 1)

$$\frac{\cos^2\theta_1}{\cos^2(\theta_1 + \beta)} = \frac{\sigma_1}{\sigma_2} \frac{\cos\theta_1}{\cos(\theta_1 + \beta)} = \sqrt{\frac{\sigma_1}{\sigma_2}} \quad \text{(Eq. 2)}$$

SUBSTITUTE NUMERICAL VALUES INTO Eq. (2):

$$\frac{\cos\theta_1}{\cos(\theta_1 + 30^\circ)} = \sqrt{\frac{7500 \text{ psi}}{2500 \text{ psi}}} = \sqrt{3} = 1.7321$$

Solve by iteration or a computer program:

$$\theta_1 = 30^{\circ}$$

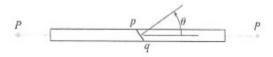
MAXIMUM NORMAL STRESS (FROM EQ. 1)

$$\sigma_{\text{max}} = \sigma_x = \frac{\sigma_1}{\cos^2 \theta_1} = \frac{7500 \text{ psi}}{\cos^2 30^\circ}$$

= 10,000 psi \leftarrow

MAXIMUM SHEAR STRESS

$$\tau_{\text{max}} = \frac{\sigma_x}{2} = 5,000 \text{ psi} \quad \longleftarrow$$



$$25^{\circ} < \theta < 45^{\circ}$$

$$A = 225 \text{ mm}^2$$

On glued joint: $\sigma_{\text{allow}} = 5.0 \text{ MPa}$

$$\tau_{\rm allow} = 3.0 \, \text{MPa}$$

ALLOWABLE STRESS σ_x IN TENSION

$$\sigma_{\theta} = \sigma_{x} \cos^{2} \theta$$
 $\sigma_{x} = \frac{\sigma_{\theta}}{\cos^{2} \theta} = \frac{5.0 \text{ MPa}}{\cos^{2} \theta}$ (1)

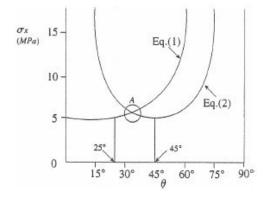
$$\tau_{\theta} = -\sigma_x \sin \theta \cos \theta$$

Since the direction of τ_{θ} is immaterial, we can write: $\tau_{\theta} \mid = \sigma_x \sin \theta \cos \theta$

or

$$\sigma_x = \frac{|\tau_\theta|}{\sin\theta\cos\theta} = \frac{3.0 \text{ MPa}}{\sin\theta\cos\theta} \quad (2$$

Graph of Eqs. (1) and (2)



(a) Determine angle Θ for largest load

Point A gives the largest value of σ_x and hence the largest load. To determine the angle θ corresponding to point A, we equate Eqs. (1) and (2).

$$\frac{5.0 \text{ MPa}}{\cos^2 \theta} = \frac{3.0 \text{ MPa}}{\sin \theta \cos \theta}$$
$$\tan \theta = \frac{3.0}{5.0} \quad \theta = 30.96^{\circ} \quad \longleftarrow$$

(b) DETERMINE THE MAXIMUM LOAD

From Eq. (1) or Eq. (2):

$$\sigma_x = \frac{5.0 \text{ MPa}}{\cos^2 \theta} = \frac{3.0 \text{ MPa}}{\sin \theta \cos \theta} = 6.80 \text{ MPa}$$

$$P_{\text{max}} = \sigma_x A = (6.80 \text{ MPa})(225 \text{ mm}^2)$$

= 1.53 kN \leftarrow

NUMERICAL DATA

$$\alpha = 55\,(10^{-6})$$
 $E = 400~{
m ksi}$ $L = 2~{
m ft}$ $\Delta T = 100$ $k = 18~{
m k/in}$. $b = 0.75~{
m in}$. $h = 1.5~{
m in}$. $\sigma_{\theta} = -950~{
m psi}$ $\sigma_{a} = -1000~{
m psi}$ $\sigma_{a} = -560~{
m psi}$ $L_{\theta} = 1.5~{
m ft}$ $A = b~h$ $f = \frac{1}{k} = 5.556 \times 10^{-5}~{
m in/lbs}$

(a) Find θ and T_{θ}

$$R_2 = \text{redundant}$$
 $R_2 = \frac{-\alpha \Delta T L}{\left(\frac{L}{EA}\right) + f} = -1.212 \times 10^3 \text{ lb}$ $\sigma_x = \frac{R_2}{A} = -1077.551 \text{ psi}$ $\sqrt{\frac{\sigma_\theta}{\sigma_x}} = 0.939 \text{ m}$

$$\theta = a\cos\left(\sqrt{\frac{\sigma_{\theta}}{\sigma_{x}}}\right) = 0.351$$
 $\cos(2\theta) = 0.763$ $\theta = 20.124^{\circ}$

$$\sigma_x \cos(\theta)^2 = -950 \text{ psi}$$
 or $\frac{\sigma_x}{2} (1 + \cos(2\theta)) = -950 \text{ psi}$ $\sigma_y = \sigma_x \cos\left(\theta + \frac{\pi}{2}\right)^2 = -127.551 \text{ psi}$

$$\theta = 0.351$$
 $\theta = 20.124^{\circ}$ $\sigma_x = -1077.551 \text{ psi}$ $2\theta = 0.702$

$$\tau_{\theta} = -\sigma_x \sin(\theta)\cos(\theta) = 348.1 \text{ psi}$$
 or $\tau_{\theta} = \frac{-\sigma_x}{2}\sin(2\theta) = 348.1 \text{ psi}$

$$\tau_{\theta} = 348 \text{ psi}$$
 $\theta = 20.1^{\circ}$

(b) Find σ_{v1} and σ_{v1}

$$\sigma_{x1} = \sigma_x \cos(\theta)^2$$
 $\sigma_{y1} = \sigma_x \cos\left(\theta + \frac{\pi}{2}\right)^2$

$$\sigma_{x1} = -950 \text{ psi}$$
 $\sigma_{y1} = -127.6 \text{ psi}$

(c) Given L = 2 ft, find k_{max}

$$k_{\rm max1} = \frac{\sigma_a A}{-\alpha \, \Delta T L - \sigma_a A \left(\frac{L}{EA}\right)} = 15625 \, {\rm lb/in.} < {\rm controls} \, ({\rm based} \, {\rm on} \, \sigma_{\rm allow})$$

or
$$k_{\text{max}2} = \frac{2\tau_a A}{-\alpha \Delta T L - 2\tau_a A \left(\frac{L}{EA}\right)} = 19444.444 \text{ lb/in.} < \text{based on allowable shear stress}$$

$$k_{\text{max}} = 15625 \text{ lb/in}.$$

(d) Given allowable normal and shear stresses, find $L_{
m max}$

k = 18000 lb/in.

$$\sigma_{x} = \frac{R_{2}}{A} \qquad \sigma_{a}A = \frac{-\alpha \Delta TL}{\left(\frac{L}{EA}\right) + f} \qquad L_{\text{max}1} = \frac{\sigma_{a}A(f)}{-\left(\alpha \Delta T + \frac{\sigma_{a}}{E}\right)} = 1.736\,\text{ft} < \text{controls (based on } \sigma_{\text{allow}})$$

or
$$L_{\text{max}2} = \frac{2\tau_a A(f)}{-\left(\alpha \Delta T + \frac{2\tau_a}{E}\right)} = 2.16 \text{ ft} < \text{based on } T_{\text{allow}}$$

$$L_{\text{max}} = 1.736 \text{ ft}$$

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(e) Find
$$\Delta T_{\rm max}$$
 given L , k , and allowable stresses $k=18000\,{\rm lb/in}.$ $L=2\,{\rm ft}$ $\sigma_a=-1000\,{\rm psi}$ $\tau_a=-560\,{\rm psi}$
$$\Delta T_{\rm max1}=\frac{\left(\frac{L}{EA}+f\right)\sigma_aA}{-\alpha L}=92.803^{\circ}{\rm F} \qquad <{\rm based \ on \ }\sigma_{\rm allow} \qquad \Delta T=100$$

$$\Delta T_{\rm max2}=\frac{\left(\frac{L}{EA}+f\right)2\tau_aA}{-\alpha L}=103.939^{\circ}{\rm F} <{\rm based \ on \ }T_{\rm allow}$$

$$\Delta T_{\rm max}=92.8^{\circ}{\rm F}$$

$$b = 50 \text{mm}$$
 $\alpha = 35 \text{deg}$

$$\sigma_{a} = 11.5 \text{MPa}$$
 $\sigma_{a} = 4.5 \text{MPa}$

$$\sigma_{\rm ga}$$
 = 3.5MPa $\tau_{\rm ga}$ = 1.25MPa

Rotate stress element CW by angle θ to align with glue joint (see fig.)

$$\theta = \alpha - 90 \text{deg} = -55 \cdot \text{deg}$$

Plane stress transformations
$$\sigma_{x} = 0$$

$$\sigma_{\rm X} = \frac{\rm P}{\Delta}$$
 $A = b^2 = 2500 \cdot {\rm mm}$

$$\begin{split} \text{Plane stress transformations} & \sigma_{x} = \frac{p}{A} \qquad A = \, b^{2} = 2500 \cdot mm^{2} \\ & \sigma_{\theta} = \, \sigma_{x} \cdot \cos(\theta)^{2} \qquad \quad \tau_{\theta} = -\sigma_{x} \cdot \sin(\theta) \cdot \cos(\theta) \end{split}$$

Equate σ_{θ} and τ_{θ} to allowable values and solve for P - min. P controls

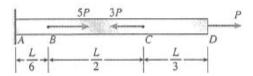
$$\sigma_{\text{max}} = \sigma_{\text{x}}$$

$$P_{\text{max}1} = \sigma_{\text{a}} \cdot A = 28.75 \cdot \text{kN}$$

$$\tau_{max} = -\left(\frac{P}{A}\right) \cdot \sin(45\text{deg}) \cdot \cos(45\text{deg}) \qquad \qquad P_{max2} = \frac{\tau_a}{2} \cdot A = 5.625 \cdot kN \qquad < \text{shear in wood controls}$$

$$P_{\text{max3}} = \frac{\sigma_{\text{ga}} \cdot A}{\cos(\theta)^2} = 26.597 \cdot kN$$

$$P_{\text{max}4} = \frac{\tau_{\text{ga}} \cdot A}{-\sin(\theta) \cdot \cos(\theta)} = 6.651 \cdot \text{kN}$$



$$P = 6 \text{ k}$$

$$L = 52 \text{ in.}$$

$$E = 10.4 \times 10^6 \text{ psi}$$

$$A = 2.76 \text{ in.}^2$$

INTERNAL AXIAL FORCES

$$N_{AB} = 3P \qquad N_{BC} = -2P \qquad N_{CD} = P$$

LENGTHS

$$L_{AB} = \frac{L}{6}$$
 $L_{BC} = \frac{L}{2}$ $L_{CD} = \frac{L}{3}$

(a) STRAIN ENERGY OF THE BAR (Eq. 2-40)

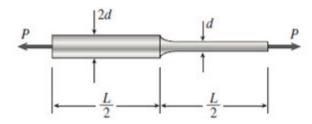
$$U = \sum \frac{N_i^2 L_i}{2E_i A_i}$$

$$= \frac{1}{2EA} \left[(3P)^2 \left(\frac{L}{6} \right) + (-2P)^2 \left(\frac{L}{2} \right) + (P)^2 \left(\frac{L}{3} \right) \right]$$

$$= \frac{P^2 L}{2EA} \left(\frac{23}{6} \right) = \frac{23P^2 L}{12EA} \quad \leftarrow$$

(b) Substitute numerical values:

$$U = \frac{23(6 \text{ k})^2(52 \text{ in.})}{12(10.4 \times 10^6 \text{ psi})(2.76 \text{ in.}^2)}$$
$$= 125 \text{ in.-lb} \quad \leftarrow$$



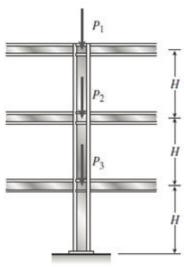
(a) STRAIN ENERGY OF THE BAR

Add the strain energies of the two segments of the bar (see Eq. 2-40).

$$U = \sum_{i=1}^{2} \frac{N_i^2 L_i}{2 E_i A_i} = \frac{P^2 (L/2)}{2E} \left[\frac{1}{\frac{\pi}{4} (2d)^2} + \frac{1}{\frac{\pi}{4} (d^2)} \right]$$
$$= \frac{P^2 L}{\pi E} \left(\frac{1}{4d^2} + \frac{1}{d^2} \right) = \frac{5P^2 L}{4\pi E d^2} \leftarrow$$

(b) Substitute numerical values:

$$P = 27 \text{ kN}$$
 $L = 600 \text{ mm}$
 $d = 40 \text{ mm}$ $E = 105 \text{ GPa}$
 $U = \frac{5(27 \text{ kN}^2)(600 \text{ mm})}{4\pi(105 \text{ GPa})(40 \text{ mm})^2}$
 $= 1.036 \text{ N} \cdot \text{m} = 1.036 \text{ J} \leftarrow$

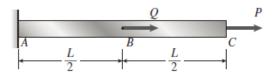


$$H = 10.5 \text{ ft}$$
 $E = 30 \times 10^6 \text{ psi}$
 $A = 15.5 \text{ in.}^2$ $P_1 = 40 \text{ k}$

$$P_2 = P_3 = 60 \text{ k}$$

 $P_2 = P_3 = 60 \text{ k}$ To find the strain energy of the column, add the strain energies of the three segments (see Eq. 2-40).

Upper segment: $N_1 = -P_1$ Middle segment: $N_2 = -(P_1 + P_2)$ Lower segment: $N_3 = -(P_1 + P_2 + P_3)$ STRAIN ENERGY $U = \sum \frac{N_i^2 L_i}{2E \cdot A_i}$ $= \frac{H}{2EA} [P_1^2 + (P_1 + P_2)^2 + (P_1 + P_2 + P_3)^2]$ $=\frac{H}{2EA}[Q]$ $[O] = (40 \text{ k})^2 + (100 \text{ k})^2 + (160 \text{ k})^2 = 37,200 \text{ k}^2$ $2EA = 2(30 \times 10^6 \text{ psi})(15.5 \text{ in.}^2) = 930 \times 10^6 \text{ lb}$ $U = \frac{(10.5 \text{ ft})(12 \text{ in./ft})}{930 \times 10^6 \text{ lb}} [37,200 \text{ k}^2]$ = 5040 in.-lb ←



(a) Force P acts alone (Q = 0)

$$U_1 = \frac{P^2L}{2EA} \leftarrow$$

(b) Force Q acts alone (P = 0)

$$U_2 = \frac{Q^2(L/2)}{2EA} = \frac{Q^2L}{4EA} \quad \leftarrow$$

(c) Forces P and Q act simultaneously

Segment BC:
$$U_{BC} = \frac{P^2(L/2)}{2EA} = \frac{P^2L}{4EA}$$

Segment AB:
$$U_{AB} = \frac{(P+Q)^2(L/2)}{2EA}$$

$$= \frac{P^2L}{4EA} + \frac{PQL}{2EA} + \frac{Q^2L}{4EA}$$

$$U_3 = U_{BC} + U_{AB} = \frac{P^2L}{2EA} + \frac{PQL}{2EA} + \frac{Q^2L}{4EA} \leftarrow$$

(Note that U_3 is *not* equal to $U_1 + U_2$. In this case, $U_3 > U_1 + U_2$. However, if Q is reversed in direction, $U_3 < U_1 + U_2$. Thus, U_3 may be larger or smaller than $U_1 + U_2$.)

DATA:			
Material	Weight density (lb/in.3)	Modulus of elasticity (ksi)	Proportional limit (psi)
Mild steel	0.284	30,000	36,000
Tool steel	0.284	30,000	75,000
Aluminum	0.0984	10,500	60,000
Rubber (soft)	0.0405	0.300	300

STRAIN ENERGY PER UNIT VOLUME

$$U = \frac{P^2L}{2EA} \qquad \text{Volume } V = AL$$

Stress
$$\sigma = \frac{P}{A}$$

$$u = \frac{U}{V} = \frac{\sigma_{PL}^2}{2E}$$

At the proportional limit:

 $u = u_R =$ modulus of resistance

$$u_R = \frac{\sigma_{PL}^2}{2E}$$
 (Eq. 1)

STRAIN ENERGY PER UNIT WEIGHT

$$U = \frac{P^2L}{2EA} \quad \text{Weight } W = \gamma AL$$

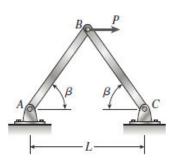
 γ = weight density

$$u_W = \frac{U}{W} = \frac{\sigma^2}{2\gamma E}$$

At the proportional limit:

$$u_W = \frac{\sigma_{PL}^2}{2\gamma E}$$
 (Eq. 2)

RESULTS			
	u_R (psi)	u _w (in.)	
Mild steel	22	76	
Tool steel	94	330	
Aluminum	171	1740	
Rubber (soft)	150	3700	



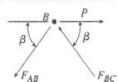
$$\beta = 60^{\circ}$$

$$L_{AB} = L_{BC} = L$$

$$\sin \beta = \sqrt{3/2}$$

 $\cos \beta = 1/2$

FREE-BODY DIAGRAM OF JOINT B



$$\Sigma F_{\mathrm{vert}} = 0$$
 \uparrow_{+} \downarrow^{-} $-F_{AB} \sin \beta + F_{BC} \sin \beta = 0$ $F_{AB} = F_{BC}$ (Eq. 1)

$$\Sigma F_{\text{horiz}} = 0 \rightarrow \leftarrow$$

$$-F_{AB}\cos\beta - F_{BC}\cos\beta + P = 0$$

$$F_{AB} = F_{BC} = \frac{P}{2\cos\beta} = \frac{P}{2(1/2)} = P$$
 (Eq. 2)

Axial forces: $N_{AB} = P$ (tension)

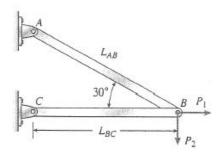
$$N_{BC} = -P$$
 (compression)

(a) STRAIN ENERGY OF TRUSS (Eq. 2-40)

$$U = \sum \frac{N_i^2 L_i}{2E_i A_i} = \frac{(N_{AB})^2 L}{2EA} + \frac{(N_{BC})^2 L}{2EA} = \frac{P^2 L}{EA} \leftarrow$$

(b) HORIZONTAL DISPLACEMENT OF JOINT B (Eq. 2-42)

$$\delta_B = \frac{2U}{P} = \frac{2}{P} \left(\frac{P^2 L}{EA} \right) = \frac{2PL}{EA} \leftarrow$$



$$P_1 = 300 \text{ lb}$$

$$P_2 = 900 \text{ lb}$$

$$A = 2.4 \text{ in.}^2$$

$$E = 30 \times 10^{6} \, \text{psi}$$

$$L_{BC} = 60 \text{ in.}$$

$$\beta = 30^{\circ}$$

$$\sin \beta = \sin 30^\circ = \frac{1}{2}$$

$$\cos \beta = \cos 30^{\circ} = \frac{\sqrt{3}}{2}$$

$$L_{AB} = \frac{L_{BC}}{\cos 30^{\circ}} = \frac{120}{\sqrt{3}}$$
 in. = 69.282 in.

$$2EA = 2(30 \times 10^6 \text{ psi})(2.4 \text{ in.}^2) = 144 \times 10^6 \text{ lb}$$

Forces F_{AB} and F_{BC} in the bars From equilibrium of joint B:

$$F_{AR} = 2P_2 = 1800 \text{ lb}$$

$$F_{BC} = P_1 - P_2\sqrt{3} = 300 \text{ lb} - 1558.8 \text{ lb}$$

Force	P_1 alone	P ₂ alone	P_1 and P_2
F_{AR}	0	1800 Ib	1800 lb
F_{BC}	300 lb	-1558.8 lb	-1258.8 lb

(a) LOAD P_1 ACTS ALONE

$$U_1 = \frac{(F_{BC})^2 L_{BC}}{2EA} = \frac{(300 \text{ lb})^2 (60 \text{ in.})}{144 \times 10^6 \text{ lb}}$$
$$= 0.0375 \text{ in.-lb} \quad \leftarrow$$

(b) Load P_2 acts alone

$$U_2 = \frac{1}{2EA} \left[(F_{AB})^2 L_{AB} + (F_{BC})^2 L_{BC} \right]$$

$$= \frac{1}{2EA} \left[(1800 \text{ lb})^2 (69.282 \text{ in.}) + (-1558.8 \text{ lb})^2 (60 \text{ in.}) \right]$$

$$= \frac{370.265 \times 10^6 \text{ lb}^2 - \text{in.}}{144 \times 10^6 \text{ lb}} = 2.57 \text{ in.-lb} \leftarrow$$

(c) Loads P_1 and P_2 act simultaneously

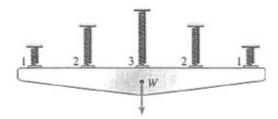
$$U_3 = \frac{1}{2EA} \Big[(F_{AB})^2 L_{AB} + (F_{BC})^2 L_{BC} \Big]$$

$$= \frac{1}{2EA} \left[(1800 \text{ lb})^2 (69.282 \text{ in.}) + (-1258.8 \text{ lb})^2 (60 \text{ in.}) \right]$$

$$= \frac{319.548 \times 10^6 \text{ lb}^2 - \text{in.}}{144 \times 10^6 \text{ lb}}$$

$$= 2.22 \text{ in.-lb} \leftarrow$$

NOTE: The strain energy U_3 is not equal to $U_1 + U_2$.



$$k_1 = 3k$$

$$k_2 = 1.5k$$

$$k_3 = k$$

 δ = downward displacement of rigid bar

For a spring:
$$U = \frac{k\delta^2}{2}$$
 Eq. (2-38b)

(a) STRAIN ENERGY U OF ALL SPRINGS

$$U = 2\left(\frac{3k\delta^2}{2}\right) + 2\left(\frac{1.5k\delta^2}{2}\right) + \frac{k\delta^2}{2} = 5k\delta^2 \leftarrow$$

(b) DISPLACEMENT δ

Work done by the weight W equals $\frac{W\delta}{2}$

Strain energy of the springs equals $5k\delta^2$

$$\therefore \frac{W\delta}{2} = 5k\delta^2 \quad \text{and} \quad \delta = \frac{W}{10k} \quad \leftarrow$$

(c) Forces in the springs

$$F_1 = 3k\delta = \frac{3 \text{ W}}{10} \quad F_2 = 1.5k\delta = \frac{3W}{20} \quad \leftarrow$$

$$F_3 = k\delta = \frac{W}{10} \leftarrow$$

(d) NUMERICAL VALUES

$$W = 600 \text{ N}$$
 $k = 7.5 \text{ N/mm} = 7500 \text{ N/mm}$

$$U = 5k\delta^2 = 5k \left(\frac{W}{10k}\right)^2 = \frac{W^2}{20k}$$

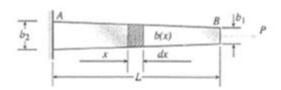
$$\delta = \frac{W}{10k} = 8.0 \text{ mm} \leftarrow$$

$$F_1 = \frac{3W}{10} = 180 \text{ N} \leftarrow$$

$$F_2 = \frac{3W}{20} = 90 \text{ N} \leftarrow$$

$$F_3 = \frac{W}{10} = 60 \text{ N} \leftarrow$$

NOTE:
$$W = 2F_1 + 2F_2 + F_3 = 600 \text{ N (Check)}$$



$$b(x) = b_2 - \frac{(b_2 - b_1)x}{L}$$

$$A(x) = tb(x)$$

$$= t \left[b_2 - \frac{(b_2 - b_1)x}{L} \right]$$

(a) STRAIN ENERGY OF THE BAR

$$U = \int \frac{[N(x)]^2 dx}{2EA(x)} \quad \text{(Eq. 2-41)}$$

$$= \int_0^L \frac{P^2 dx}{2Etb(x)} = \frac{P^2}{2Et} \int_0^L \frac{dx}{b_2 - (b_2 - b_1)_L^t} \quad \text{(1)}$$
From Appendix C:
$$\int \frac{dx}{a + bx} = \frac{1}{b} \ln (a + bx)$$

Apply this integration formula to Eq. (1):

$$U = \frac{P^2}{2Et} \left[\frac{1}{-(b_2 - b_1)(\frac{1}{L})} \ln \left[b_2 - \frac{(b_2 - b_1)x}{L} \right] \right]_0^L$$

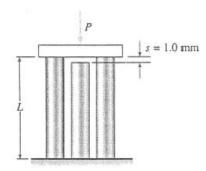
$$= \frac{P^2}{2Et} \left[\frac{-L}{(b_2 - b_1)} \ln b_1 - \frac{-L}{(b_2 - b_1)} \ln b_2 \right]$$

$$U = \frac{P^2L}{2Et(b_2 - b_1)} \ln \frac{b_2}{b_1} \leftarrow$$

(b) ELONGATION OF THE BAR (Eq. 2-42)

$$\delta = \frac{2U}{P} = \frac{PL}{Et(b_2 - b_1)} \ln \frac{b_2}{b_1} \leftarrow$$

NOTE: This result agrees with the formula derived in Prob. 2.3-17.



 $s = 1.0 \, \text{mm}$

L = 1.0 m

For each bar:

 $A = 3000 \text{ mm}^2$

E = 45 GPa

$$\frac{EA}{I} = 135 \times 10^6 \text{ N/m}$$

(a) Load P_1 required to close the gap

In general,
$$\delta = \frac{PL}{EA}$$
 and $P = \frac{EA\delta}{L}$

For two bars, we obtain:

$$P_1 = 2\left(\frac{EAs}{L}\right) = 2(135 \times 10^6 \text{ N/m})(1.0 \text{ mm})$$

$$P_1 = 270 \text{ kN} \leftarrow$$

(b) Displacement δ for P = 400 kN

Since $P > P_1$, all three bars are compressed. The force P equals P_1 plus the additional force required to compress all three bars by the amount $\delta - s$.

$$P = P_1 + 3\left(\frac{EA}{L}\right)(\delta - s)$$

or $400 \text{ kN} = 270 \text{ kN} + 3(135 \times 10^6 \text{ N/m})$ $(\delta - 0.001 \text{ m})$

Solving, we get $\delta = 1.321 \text{ mm}$

(c) Strain energy U for P = 400 kN

$$U = \sum \frac{EA\delta^2}{2L}$$

Outer bars:

 $\delta = 1.321 \, \text{mm}$

Middle bar:

 $\delta = 1.321 \, \text{mm} - s$

= 0.321 mm

$$U = \frac{EA}{2L} [2(1.321 \text{ mm})^2 + (0.321 \text{ mm})^2]$$

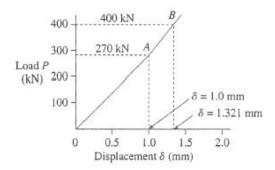
$$= \frac{1}{2} (135 \times 10^6 \text{ N/m}) (3.593 \text{ mm}^2)$$

(d) Load-displacement diagram

$$U = 243 \text{ J} = 243 \text{ N} \cdot \text{m}$$

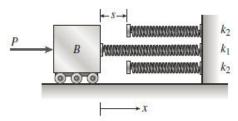
$$\frac{P\delta}{2} = \frac{1}{2} (400 \text{ kN})(1.321 \text{ mm}) = 264 \text{ N} \cdot \text{m}$$

The strain energy U is *not* equal to $\frac{P\delta}{2}$ = because the load-displacement relation is not linear.



U = area under line OAB.

 $\frac{P\delta}{2}$ = area under a straight line from O to B, which is larger than U.



Force P_0 required to close the gap:

$$P_0 = k_1 s \tag{1}$$

FORCE-DISPLACEMENT RELATION BEFORE GAP IS CLOSED

$$P = k_1 x$$
 $(0 \le x \le s)(0 \le P \le P_0)$ (2)

FORCE-DISPLACEMENT RELATION AFTER GAP IS CLOSED

All three springs are compressed. Total stiffness equals $k_1 + 2k_2$. Additional displacement equals x - s. Force P equals P_0 plus the force required to compress all three springs by the amount x - s.

$$P = P_0 + (k_1 + 2k_2)(x - s)$$

$$= k_1 s + (k_1 + 2k_2)x - k_1 s - 2k_2 s$$

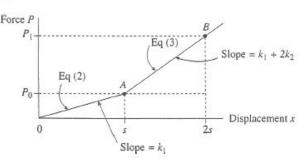
$$P = (k_1 + 2k_2)x - 2k_2 s \quad (x \ge s); (P \ge P_0)$$

$$P_1 = \text{force } P \text{ when } x = 2s$$
(3)

Substitute x = 2s into Eq. (3):

$$P_1 = 2(k_1 + k_2)s (4)$$

(a) Force-displacement diagram



(b) Strain energy U_1 when x = 2s

 U_1 = Area below force-displacement curve

$$= \underbrace{\frac{1}{2}P_0s + P_0s + \frac{1}{2}(P_1 - P_0)s} + \underbrace{\frac{1}{2}P_1s}$$

$$= k_1s^2 + (k_1 + k_2)s^2$$

$$U_1 = (2k_1 + k_2)s^2 \leftarrow (5)$$

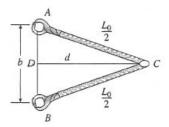
(c) Strain energy U_1 is not equal to $\frac{P\delta}{2}$ For $\delta=2s$: $\frac{P\delta}{2}=\frac{1}{2}P_1(2\ s)=P_1s=2(k_1+k_2)s^2$ (This quantity is greater than U_1 .)

 U_1 = area under line OAB.

 $\frac{P\delta}{2}$ = area under a straight line from O to B, which is larger than U_1 .

Thus, $\frac{P\delta}{2}$ is *not* equal to the strain energy because the force-displacement relation is not linear.

DIMENSIONS BEFORE THE LOAD P IS APPLIED



$$L_0 = 760 \text{ mm}$$
 $\frac{L_0}{2} = 380 \text{ mm}$

 $b = 380 \, \text{mm}$

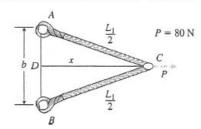
Bungee cord:

$$k = 140 \text{ N/m}$$

From triangle ACD:

$$d = \frac{1}{2}\sqrt{L_0^2 - b^2} = 329.09 \text{ mm} \tag{1}$$

DIMENSIONS AFTER THE LOAD P IS APPLIED



Let x = distance CD

Let L_1 = stretched length of bungee cord

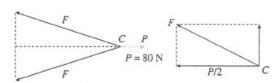
From triangle ACD:

$$\frac{L_1}{2} = \sqrt{\left(\frac{b}{2}\right)^2 + x^2} \tag{2}$$

$$L_1 = \sqrt{b^2 + 4x^2} \tag{3}$$

EQUILIBRIUM AT POINT C

Let F = tensile force in bungee cord



$$\frac{F}{P/2} = \frac{L_1/2}{x} \quad F = \left(\frac{P}{2}\right) \left(\frac{L_1}{2}\right) \left(\frac{1}{x}\right)$$

$$= \frac{P}{2} \sqrt{1 + \left(\frac{b}{2x}\right)^2} \tag{4}$$

ELONGATION OF BUNGEE CORD

Let δ = elongation of the entire bungee cord

$$\delta = \frac{F}{k} = \frac{P}{2k} \sqrt{1 + \frac{b^2}{4x^2}} \tag{5}$$

Final length of bungee cord = original length + δ

$$L_1 = L_0 + \delta = L_0 + \frac{P}{2k} \sqrt{1 + \frac{b^2}{4x^2}}$$
 (6)

SOLUTION OF EQUATIONS

Combine Eqs. (6) and (3):

$$L_1 = L_0 + \frac{P}{2k}\sqrt{1 + \frac{b^2}{4x^2}} = \sqrt{b^2 + 4x^2}$$

or
$$L_1 = L_0 + \frac{P}{4kx} \sqrt{b^2 + 4x^2} = \sqrt{b^2 + 4x^2}$$

$$L_0 = \left(1 - \frac{P}{4kx}\right)\sqrt{b^2 + 4x^2} \tag{7}$$

This equation can be solved for x.

SUBSTITUTE NUMERICAL VALUES INTO Eq. (7):

760 mm =
$$\left[1 - \frac{(80 \text{ N})(1000 \text{ mm/m})}{4(140 \text{ N/m})x}\right]$$
$$\times \sqrt{(380 \text{ mm})^2 + 4x^2} \tag{8}$$

$$760 = \left(1 - \frac{142.857}{x}\right) \sqrt{144,400 + 4x^2} \tag{9}$$

Units: x is in millimeters

Solve for x (Use trial-and-error or a computer program):

x = 497.88 mm

(a) Strain energy U of the bungee cord

$$U = \frac{k\delta^2}{2}$$
 $k = 140 \text{ N/m}$ $P = 80 \text{ N}$

From Eq. (5)

$$\delta = \frac{P}{2k} \sqrt{1 + \frac{b^2}{4x^2}} = 305.81 \text{ mm}$$

$$U = \frac{1}{2} (140 \text{ N/m})(305.81 \text{ mm})^2 = 6.55 \text{ N} \cdot \text{m}$$

$$U = 6.55 \,\mathrm{J} \leftarrow$$

(b) Displacement δ_C of point C

$$\delta_C = x - d = 497.88 \text{ mm} - 329.09 \text{ mm}$$

= 168.8 mm \leftarrow

CLICK HERE TO ACCESS THE COMPLETE Solutions

(c) Comparison of Strain energy U with the quantity $P\delta_{C}/2$

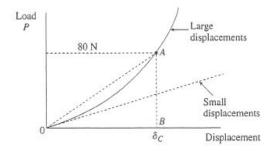
$$U = 6.55 \text{ J}$$

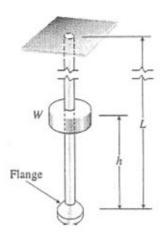
$$\frac{P\delta_C}{2} = \frac{1}{2} (80 \text{ N})(168.8 \text{ mm}) = 6.75 \text{ J}$$

The two quantities are not the same. The work done by the load P is not equal to $P\delta_C/2$ because the load-displacement relation (see below) is non-linear when the displacements are large. (The work done by the load P is equal to the strain energy because the bungee cord behaves elastically and there are no energy losses.)

$$U = \text{area } OAB \text{ under the curve } OA.$$

$$\frac{P\delta_C}{2}$$
 = area of triangle *OAB*, which is greater than *U*.





$$W = 150 \text{ lb}$$

$$h = 2.0 \text{ in}$$

$$h = 2.0 \text{ in.}$$
 $L = 4.0 \text{ ft} = 48 \text{ in.}$ $E = 30 \times 10^6 \text{ psi}$ $A = 0.75 \text{ in.}^2$

$$E = 30 \times 10^{6} \, \text{psi}$$

$$A = 0.75 \text{ in.}^2$$

(a) DOWNWARD DISPLACEMENT OF FLANGE

$$\delta_{st} = \frac{WL}{EA} = 0.00032 \text{ in.}$$

Eq. (2-53):

$$\delta_{\text{max}} = \delta_{st} \left[1 + \left(1 + \frac{2h}{\delta_{st}} \right)^{1/2} \right]$$

$$= 0.0361 \text{ in.} \quad \leftarrow$$

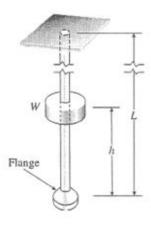
(b) MAXIMUM TENSILE STRESS (Eq. 2-55)

$$\sigma_{\max} = \frac{E\delta_{\max}}{L} = 22,600 \text{ psi} \quad \leftarrow$$

(c) IMPACT FACTOR (Eq. 2-61)

Impact factor =
$$\frac{\delta_{\text{max}}}{\delta_{st}} = \frac{0.0361 \text{ in.}}{0.00032 \text{ in.}}$$

= 113



$$M = 80 \text{ kg}$$

 $W = Mg = (80 \text{ kg})(9.81 \text{ m/s}^2)$
 $= 784.8 \text{ N}$
 $h = 0.5 \text{ m}$ $L = 3.0 \text{ m}$
 $E = 170 \text{ GPa}$ $A = 350 \text{ mm}^2$

(a) DOWNWARD DISPLACEMENT OF FLANGE

$$\delta_{st} = \frac{WL}{EA} = 0.03957 \text{ mm}$$

$$\text{Eq. (2-53):} \quad \delta_{\text{max}} = \delta_{st} \left[1 + \left(1 + \frac{2h}{\delta_{st}} \right)^{1/2} \right]$$

$$= 6.33 \text{ mm} \qquad \leftarrow$$

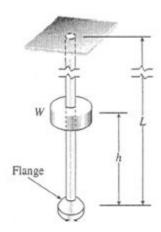
(b) MAXIMUM TENSILE STRESS (Eq. 2-55)

$$\sigma_{\text{max}} = \frac{E\delta_{\text{max}}}{L} = 359 \,\text{MPa} \quad \leftarrow$$

(c) IMPACT FACTOR (Eq. 2-61)

Impact factor =
$$\frac{\delta_{\text{max}}}{\delta_{st}} = \frac{6.33 \text{ mm}}{0.03957 \text{ mm}}$$

= $160 \leftarrow$



$$W = 50 \text{ lb}$$
 $h = 2.0 \text{ in.}$
 $L = 3.0 \text{ ft} = 36 \text{ in.}$
 $E = 30,000 \text{ psi}$ $A = 0.25 \text{ in.}^2$

(a) DOWNWARD DISPLACEMENT OF FLANGE

$$\delta_{st} = \frac{WL}{EA} = 0.00024 \text{ in.}$$

Eq. (2-53):
$$\delta_{\text{max}} = \delta_{st} \left[1 + \left(1 + \frac{2h}{\delta_{st}} \right)^{1/2} \right]$$

= 0.0312 in.

(b) MAXIMUM TENSILE STRESS (Eq. 2-55)

$$\sigma_{\text{max}} = \frac{E\delta_{\text{max}}}{L} = 26,000 \text{ psi} \quad \leftarrow$$

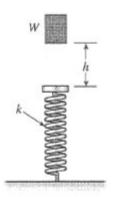
(c) IMPACT FACTOR (Eq. 2-61)

Impact factor =
$$\frac{\delta_{\text{max}}}{\delta_{\text{st}}} = \frac{0.0312 \text{ in.}}{0.00024 \text{ in.}}$$

= 130 \leftarrow

CLICK HERE TO ACCESS THE COMPLETE Solutions

Problem 2.8-4



$$W = 5.0 \text{ N}$$
 $h = 200 \text{ mm}$ $k = 90 \text{ N/m}$

(a) MAXIMUM SHORTENING OF THE SPRING

$$\delta_{st} = \frac{W}{k} = \frac{5.0 \text{ N}}{90 \text{ N/m}} = 55.56 \text{ mm}$$

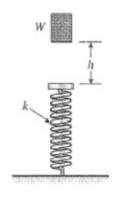
Eq. (2-53): $\delta_{\text{max}} = \delta_{st} \left[1 + \left(1 + \frac{2h}{\delta_{st}} \right)^{1/2} \right]$

= 215 mm \leftarrow

(b) IMPACT FACTOR (Eq. 2-61)

Impact factor =
$$\frac{\delta_{\text{max}}}{\delta_{st}} = \frac{215 \text{ mm}}{55.56 \text{ mm}}$$

= 3.9 \leftarrow



$$W = 1.0 \text{ lb}$$
 $h = 12 \text{ in.}$ $k = 0.5 \text{ lb/in.}$

(a) MAXIMUM SHORTENING OF THE SPRING

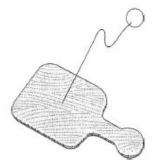
$$\delta_{st} = \frac{W}{k} = \frac{1.0 \text{ lb}}{0.5 \text{ lb/in.}} = 2.0 \text{ in.}$$
Eq. (2-53):
$$\delta_{\text{max}} = \delta_{st} \left[1 + \left(1 + \frac{2h}{\delta_{st}} \right)^{1/2} \right]$$

$$= 9.21 \text{ in.} \quad \leftarrow$$

(b) IMPACT FACTOR (Eq. 2-61)

Impact factor =
$$\frac{\delta_{\text{max}}}{\delta_{st}} = \frac{9.21 \text{ in.}}{2.0 \text{ in.}}$$

= 4.6 \leftarrow



$$g = 9.81 \text{ m/s}^2$$
 $E = 2.0 \text{ MPa}$

$$E = 2.0 \text{ MPa}$$

$$A = 1.6 \text{ mm}^2$$
 $L_0 = 200 \text{ mm}$

$$L_0 = 200 \text{ mm}$$

$$L_1 = 900 \text{ mm}$$
 $W = 450 \text{ mN}$

$$W = 450 \text{ mN}$$

WHEN THE BALL LEAVES THE PADDLE

$$KE = \frac{Wv^2}{2g}$$

WHEN THE RUBBER CORD IS FULLY STRETCHED:

$$U = \frac{EA\delta^2}{2L_0} = \frac{EA}{2L_0}(L_1 - L_0)^2$$

CONSERVATION OF ENERGY

$$KE = U \frac{Wv^2}{2g} = \frac{EA}{2L_0}(L_1 - L_0)^2$$

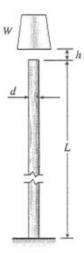
$$v^2 = \frac{gEA}{WL_0}(L_1 - L_0)^2$$

$$v = (L_1 - L_0) \sqrt{\frac{gEA}{WL_0}} \quad \leftarrow$$

SUBSTITUTE NUMERICAL VALUES:

$$v = (700 \text{ mm}) \sqrt{\frac{(9.81 \text{ m/s}^2) (2.0 \text{ MPa}) (1.6 \text{ mm}^2)}{(450 \text{ mN}) (200 \text{ mm})}}$$

= 13.1 m/s \leftarrow



$$W = 4500 \text{ lb}$$
 $d = 12 \text{ in.}$
 $L = 15 \text{ ft} = 180 \text{ in.}$
 $A = \frac{\pi d^2}{4} = 113.10 \text{ in.}^2$

$$E = 1.6 \times 10^6 \text{ psi}$$

 $\sigma_{\text{allow}} = 2500 \text{ psi } (= \sigma_{\text{max}})$

Find hmax

STATIC STRESS

$$\sigma_{st} = \frac{W}{A} = \frac{4500 \text{ lb}}{113.10 \text{ in.}^2} = 39.79 \text{ psi}$$

MAXIMUM HEIGHT hmax

Eq. (2-59):
$$\sigma_{\text{max}} = \sigma_{st} \left[1 + \left(1 + \frac{2hE}{L\sigma_{st}} \right)^{1/2} \right]$$

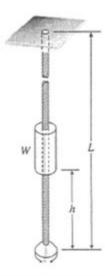
or

$$\frac{\sigma_{\text{max}}}{\sigma_{st}} - 1 = \left(1 + \frac{2hE}{L\sigma_{st}}\right)^{1/2}$$

Square both sides and solve for h

$$h = h_{\text{max}} = \frac{L\sigma_{\text{max}}}{2E} \left(\frac{\sigma_{\text{max}}}{\sigma_{\text{st}}} - 2 \right) \quad \leftarrow$$

$$h_{\text{max}} = \frac{(180 \text{ in.}) (2500 \text{ psi})}{2(1.6 \times 10^6 \text{ psi})} \left(\frac{2500 \text{ psi}}{39.79 \text{ psi}} - 2\right)$$
$$= 8.55 \text{ in.} \leftarrow$$



$$W = Mg = (35 \text{ kg})(9.81 \text{ m/s}^2) = 343.4 \text{ N}$$

 $A = 40 \text{ mm}^2$ $E = 130 \text{ GPa}$
 $h = 1.0 \text{ m}$ $\sigma_{\text{allow}} = \sigma_{\text{max}} = 500 \text{ MPa}$
Find minimum length L_{min} .

STATIC STRESS

$$\sigma_{st} = \frac{W}{A} = \frac{343.4 \text{ N}}{40 \text{ mm}^2} = 8.585 \text{ MPa}$$

MINIMUM LENGTH Lmin

Eq. (2-59):
$$\sigma_{\text{max}} = \sigma_{\text{st}} \left[1 + \left(1 + \frac{2hE}{L\sigma_{\text{st}}} \right)^{1/2} \right]$$

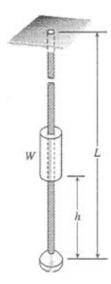
or

$$\frac{\sigma_{\text{max}}}{\sigma_{st}} - 1 = \left(1 + \frac{2hE}{L\sigma_{st}}\right)^{1/2}$$

Square both sides and solve for L:

$$L = L_{\min} = \frac{2Eh\sigma_{st}}{\sigma_{\max}(\sigma_{\max} - 2\sigma_{st})} \leftarrow$$

$$L_{\min} = \frac{2(130 \text{ GPa}) (1.0 \text{ m}) (8.585 \text{ MPa})}{(500 \text{ MPa}) [500 \text{ MPa} - 2(8.585 \text{ MPa})]}$$
$$= 9.25 \text{ m} \leftarrow$$



$$W=100~{
m lb}$$
 $A=0.080~{
m in.}^2$ $E=21 imes10^6~{
m psi}$ $h=45~{
m in}$ $\sigma_{
m allow}=\sigma_{
m max}=70~{
m ksi}$ Find minimum length $L_{
m min}$.

STATIC STRESS

$$\sigma_{st} = \frac{W}{A} = \frac{100 \text{ lb}}{0.080 \text{ in.}^2} = 1250 \text{ psi}$$

MINIMUM LENGTH Lmin

Eq. (2-59):
$$\sigma_{\text{max}} = \sigma_{st} \left[1 + \left(1 + \frac{2hE}{L\sigma_{st}} \right)^{1/2} \right]$$

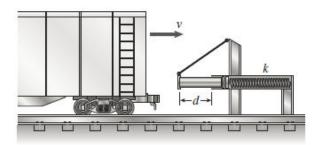
or

$$\frac{\sigma_{\text{max}}}{\sigma_{st}} - 1 = \left(1 + \frac{2hE}{L\sigma_{st}}\right)^{1/2}$$

Square both sides and solve for L:

$$L = L_{\min} = \frac{2Eh\sigma_{st}}{\sigma_{\max}(\sigma_{\max} - 2\sigma_{st})} \leftarrow$$

$$L_{\min} = \frac{2(21 \times 10^6 \text{ psi}) (45 \text{ in.}) (1250 \text{ psi})}{(70,000 \text{ psi}) [70,000 \text{ psi} - 2(1250 \text{ psi})]}$$
$$= 500 \text{ in.} \quad \leftarrow$$



$$k = 8.0 \text{ MN/m}$$
 $W = 545 \text{ kN}$

d = maximum displacement of spring

$$d = \delta_{\text{max}} = 450 \text{ mm}$$

Find ν_{max} .

KINETIC ENERGY BEFORE IMPACT

$$KE = \frac{Mv^2}{2} = \frac{Wv^2}{2g}$$

STRAIN ENERGY WHEN SPRING IS COMPRESSED TO THE MAXIMUM ALLOWABLE AMOUNT

$$U = \frac{k\delta_{\text{max}}^2}{2} = \frac{kd^2}{2}$$

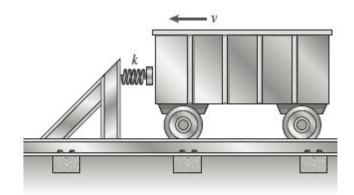
CONSERVATION OF ENERGY

$$KE = U \frac{Wv^2}{2g} = \frac{kd^2}{2} \quad v^2 = \frac{kd^2}{W/g}$$

$$v = v_{\text{max}} = d\sqrt{\frac{k}{W/g}} \quad \leftarrow$$

$$v_{\text{max}} = (450 \text{ mm}) \sqrt{\frac{8.0 \text{ MN/m}}{(545 \text{ kN})/(9.81 \text{ m/s}^2)}}$$

= 5400 mm/s = 5.4 m/s \leftarrow



$$k = 1120 \text{ lb/in.}$$
 $W = 3450 \text{ lb}$

$$\nu = 7 \text{ mph} = 123.2 \text{ in./sec}$$

$$g = 32.2 \text{ ft/sec}^2 = 386.4 \text{ in./sec}^2$$

Find the shortening δ_{max} of the spring.

KINETIC ENERGY JUST BEFORE IMPACT

$$KE = \frac{Mv^2}{2} = \frac{Wv^2}{2g}$$

STRAIN ENERGY WHEN SPRING IS FULLY COMPRESSED

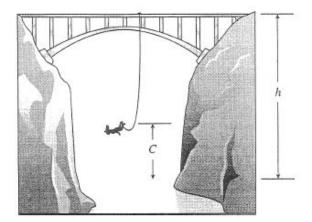
$$U = \frac{k\delta_{\max}^2}{2}$$

Conservation of energy

$$KE = U \frac{Wv^2}{2g} = \frac{k\delta_{\text{max}}^2}{2}$$

Solve for
$$\delta_{\text{max}}$$
: $\delta_{\text{max}} = \sqrt{\frac{Wv^2}{gk}}$ \leftarrow

$$\delta_{\text{max}} = \sqrt{\frac{(3450 \text{ lb}) (123.2 \text{ in./sec})^2}{(386.4 \text{ in./sec}^2) (1120 \text{ lb/in.})}}$$
= 11.0 in. \leftarrow



$$W = Mg = (55 \text{ kg})(9.81 \text{ m/s}^2)$$

= 539.55 N

$$EA = 2.3 \text{ kN}$$

Height: h = 60 m

Clearance: C = 10 m

Find length L of the bungee cord.

P.E. = Potential energy of the jumper at the top of bridge (with respect to lowest position)

$$= W(L + \delta_{\text{max}})$$

U = strain energy of cord at lowest position

$$=\frac{EA\delta_{\max}^2}{2L}$$

CONSERVATION OF ENERGY

$$P.E. = U$$
 $W(L + \delta_{\text{max}}) = \frac{EA\delta_{\text{max}}^2}{2L}$

or
$$\delta_{\text{max}}^2 - \frac{2WL}{FA}\delta_{\text{max}} - \frac{2WL^2}{FA} = 0$$

Solve quadratic equation for δ_{max} :

$$\delta_{\text{max}} = \frac{WL}{EA} + \left[\left(\frac{WL}{EA} \right)^2 + 2L \left(\frac{WL}{EA} \right) \right]^{1/2}$$
$$= \frac{WL}{EA} \left[1 + \left(1 + \frac{2EA}{W} \right)^{1/2} \right]$$

VERTICAL HEIGHT

$$h = C + L + \delta_{\text{max}}$$

$$h - C = L + \frac{WL}{EA} \left[1 + \left(1 + \frac{2EA}{W} \right)^{1/2} \right]$$

Solve for L:

$$L = \frac{h - C}{1 + \frac{W}{EA} \left[1 + \left(1 + \frac{2EA}{W} \right)^{1/2} \right]} \quad \leftarrow$$

$$\frac{W}{EA} = \frac{539.55 \text{ N}}{2.3 \text{ kN}} = 0.234587$$

Numerator =
$$h - C = 60 \text{ m} - 10 \text{ m} = 50 \text{ m}$$

Denominator =
$$1 + (0.234587)$$

$$\times \left[1 + \left(1 + \frac{2}{0.234587}\right)^{1/2}\right]$$

= 1.9586

$$L = \frac{50 \text{ m}}{1.9586} = 25.5 \text{ m} \leftarrow$$





$$W = Weight$$

Properties of elastic cord:

E =modulus of elasticity

A = cross-sectional area

L = original length

 δ_{max} = elongation of elastic cord

P.E. = potential energy of weight before fall (with respect to lowest position)

$$P.E. = W(L + \delta_{max})$$

Let U = strain energy of cord at lowest position.

$$U = \frac{EA\delta_{\text{max}}^2}{2L}$$

CONSERVATION OF ENERGY

$$P.E. = U$$
 $W(L + \delta_{\text{max}}) = \frac{EA\delta_{\text{max}}^2}{2L}$

or
$$\delta_{\text{max}}^2 - \frac{2WL}{EA}\delta_{\text{max}} - \frac{2WL^2}{EA} = 0$$

Solve quadratic equation for δ_{max} :

$$\delta_{\text{max}} = \frac{WL}{EA} + \left[\left(\frac{WL}{EA} \right)^2 + 2L \left(\frac{WL}{EA} \right) \right]^{1/2}$$

STATIC ELONGATION

$$\delta_{st} = \frac{WL}{FA}$$

IMPACT FACTOR

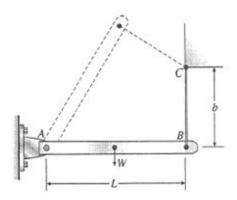
$$\frac{\delta_{\max}}{\delta_{st}} = 1 + \left[1 + \frac{2EA}{W}\right]^{1/2} \quad \leftarrow$$

NUMERICAL VALUES

$$\delta_{st} = (2.5\%)(L) = 0.025L$$

$$\delta_{st} = \frac{WL}{EA}$$
 $\frac{W}{EA} = 0.025$ $\frac{EA}{W} = 40$

Impact factor =
$$1 + [1 + 2(40)]^{1/2} = 10$$
 \leftarrow



RIGID BAR:

$$W = Mg = (1.0 \text{ kg})(9.81 \text{ m/s}^2)$$

= 9.81 N

$$L = 0.5 \text{ m}$$

NYLON CORD:

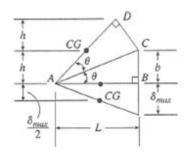
$$A = 30 \text{ mm}^2$$

$$b = 0.25 \text{ m}$$

$$E = 2.1 \text{ GPa}$$

Find maximum stress σ_{max} in cord BC.

GEOMETRY OF BAR AB AND CORD BC



$$\overline{CD} = \overline{CB} = b$$

$$\overline{AD} = \overline{AB} = L$$

h =height of center of gravity of raised bar AD

 δ_{max} = elongation of cord

From triangle ABC:sin
$$\theta = \frac{b}{\sqrt{b^2 + L^2}}$$

$$\cos \theta = \frac{L}{\sqrt{b^2 + L^2}}$$

From line AD:
$$\sin 2\theta = \frac{2h}{AD} = \frac{2h}{L}$$

From Appendix C: $\sin 2\theta = 2 \sin \theta \cos \theta$

$$\therefore \frac{2h}{L} = 2\left(\frac{b}{\sqrt{b^2 + L^2}}\right)\left(\frac{L}{\sqrt{b^2 + L^2}}\right) = \frac{2bL}{b^2 + L^2}$$
and $h = \frac{bL^2}{b^2 + L^2}$ (Eq. 1)

CONSERVATION OF ENERGY

P.E. = potential energy of raised bar AD

$$= W\left(h + \frac{\delta_{\text{max}}}{2}\right)$$

 $U = \text{strain energy of stretched cord} = \frac{EA\delta_{\text{max}}^2}{2h}$

$$P.E. = U \quad W\left(h + \frac{\delta_{\text{max}}}{2}\right) = \frac{EA\delta_{\text{max}}^2}{2b}$$
 (Eq. 2)

For the cord: $\delta_{\max} = \frac{\sigma_{\max} b}{E}$

Substitute into Eq. (2) and rearrange:

$$\sigma_{\text{max}}^2 - \frac{W}{A}\sigma_{\text{max}} - \frac{2WhE}{bA} = 0 \qquad \text{(Eq. 3)}$$

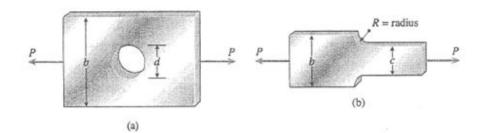
Substitute from Eq. (1) into Eq. (3):

$$\sigma_{\text{max}}^2 - \frac{W}{A}\sigma_{\text{max}} - \frac{2WL^2E}{A(b^2 + L^2)} = 0$$
 (Eq. 4)

Solve for σ_{\max} :

$$\sigma_{\text{max}} = \frac{W}{2A} \left[1 + \sqrt{1 + \frac{8L^2 EA}{W(b^2 + L^2)}} \right] \leftarrow$$

$$\sigma_{\text{max}} = 33.3 \text{ MPa} \quad \leftarrow$$



$$P = 3.0 \text{ k}$$
 $t = 0.25 \text{ in.}$

(a) BAR WITH CIRCULAR HOLE (b = 6 in.)

Obtain K from Fig. 2-84

For
$$d = 1$$
 in.: $c = b - d = 5$ in.

$$\sigma_{\text{nom}} = \frac{P}{ct} = \frac{3.0 \text{ k}}{(5 \text{ in.}) (0.25 \text{ in.})} = 2.40 \text{ ksi}$$

$$d/b = \frac{1}{6} \quad K \approx 2.60$$

$$\sigma_{\text{max}} = k\sigma_{\text{nom}} \approx 6.2 \text{ ksi} \leftarrow$$
For $d = 2$ in.: $c = b - d = 4$ in.

$$\sigma_{\text{nom}} = \frac{P}{ct} = \frac{3.0 \text{ k}}{(4 \text{ in.}) (0.25 \text{ in.})} = 3.00 \text{ ksi}$$

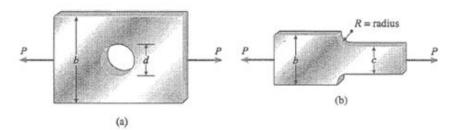
$$d/b = \frac{1}{3} \quad K \approx 2.31$$

 $\sigma_{\text{max}} = K\sigma_{\text{nom}} \approx 6.9 \text{ ksi} \leftarrow$

(b) STEPPED BAR WITH SHOULDER FILLETS

$$b = 4.0 \text{ in.}$$
 $c = 2.5 \text{ in.; Obtain } k \text{ from Fig. 2-86}$

$$\sigma_{\text{nom}} = \frac{P}{ct} = \frac{3.0 \text{ k}}{(2.5 \text{ in.}) (0.25 \text{ in.})} = 4.80 \text{ ksi}$$
For $R = 0.25 \text{ in.: } R/c = 0.1$ $b/c = 1.60$ $k \approx 2.30 \, \sigma_{\text{max}} = K\sigma_{\text{nom}} \approx 11.0 \text{ ksi} \leftarrow$
For $R = 0.5 \text{ in.: } R/c = 0.2$ $b/c = 1.60$ $K \approx 1.87$ $\sigma_{\text{max}} = K\sigma_{\text{nom}} \approx 9.0 \text{ ksi} \leftarrow$



$$P = 2.5 \text{ kN}$$
 $t = 5.0 \text{ mm}$

(a) BAR WITH CIRCULAR HOLE (b = 60 mm) Obtain K from Fig. 2-84

For
$$d = 12$$
 mm: $c = b - d = 48$ mm
$$\sigma_{\text{nom}} = \frac{P}{ct} = \frac{2.5 \text{ kN}}{(48 \text{ mm}) (5 \text{ mm})} = 10.42 \text{ MPa}$$

$$d/b = \frac{1}{5} \quad K \approx 2.51$$

$$\sigma_{\text{max}} = K\sigma_{\text{nom}} \approx 26 \text{ MPa} \quad \leftarrow$$
For $d = 20$ mm: $c = b - d = 40$ mm

$$\sigma_{\text{nom}} = \frac{P}{ct} = \frac{2.5 \text{ kN}}{(40 \text{ mm}) (5 \text{ mm})} = 12.50 \text{ MPa}$$

$$d/b = \frac{1}{3} \quad K \approx 2.31$$

$$\sigma_{\text{max}} = K\sigma_{\text{nom}} \approx 29 \text{ MPa} \quad \leftarrow$$

(b) STEPPED BAR WITH SHOULDER FILLETS

$$b = 60 \text{ mm}$$
 $c = 40 \text{ mm}$;
Obtain K from Fig 2-86
 $\sigma_{\text{nom}} = \frac{P}{ct} = \frac{2.5 \text{ kN}}{(40 \text{ mm}) (5 \text{ mm})} = 12.50 \text{ MPa}$
For $R = 6 \text{ mm}$: $R/c = 0.15$ $b/c = 1.5$
 $K \approx 2.00$ $\sigma_{\text{max}} = K\sigma_{\text{nom}} \approx 25 \text{ MPa} \leftarrow$
For $R = 10 \text{ mm}$: $R/c = 0.25$ $b/c = 1.5$

 $K \approx 1.75$ $\sigma_{\text{max}} = K\sigma_{\text{nom}} \approx 22 \text{ MPa} \leftarrow$



t =thickness

 σ_t = allowable tensile stress

Find Pmax

Find K from Fig. 2-84

$$P_{\text{max}} = \sigma_{\text{nom}} ct = \frac{\sigma_{\text{max}}}{K} ct = \frac{\sigma_t}{K} (b - d)t$$
$$= \frac{\sigma_t}{K} bt \left(1 - \frac{d}{b}\right)$$

Because σ_t , b, and t are constants, we write:

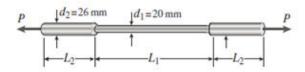
$$P^* = \frac{P_{\text{max}}}{\sigma_t b t} = \frac{1}{K} \left(1 - \frac{d}{b} \right)$$

d		- 20	
\overline{b}	K	P*	
0	3.00	0.333	
0.1	2.73	0.330	
0.2	2.50	0.320	
0.3	2.35	0.298	
0.4	2.24	0.268	

We observe that $P_{\rm max}$ decreases as d/b increases. Therefore, the maximum load occurs when the hole becomes very small.

$$\left(\frac{d}{b} \to 0 \text{ and } K \to 3\right)$$

$$P_{\text{max}} = \frac{\sigma_t bt}{3} \leftarrow$$



$$E = 100 \text{ GPa}$$

$$\delta = 0.12 \, \text{mm}$$

$$L_2 = 0.1 \text{ m}$$

$$L_1 = 0.3 \text{ m}$$

$$R = \text{radius of fillets} = \frac{26 \text{ mm} - 20 \text{ mm}}{2} = 3 \text{ mm}$$

$$\delta = 2\left(\frac{PL_2}{EA_2}\right) + \frac{PL_1}{EA_1}$$

Solve for *P*:
$$P = \frac{\delta E A_1 A_2}{2L_2 A_1 + L_1 A_2}$$

Use Fig. 2-87 for the stress-concentration factor:

$$\sigma_{\text{nom}} = \frac{P}{A_1} = \frac{\delta E A_2}{2L_2 A_1 + L_1 A_2} = \frac{\delta E}{2L_2 \left(\frac{A_1}{A_2}\right) + L_1}$$
$$= \frac{\delta E}{2L_2 \left(\frac{d_1}{d_2}\right)^2 + L_1}$$

SUBSTITUTE NUMERICAL VALUES:

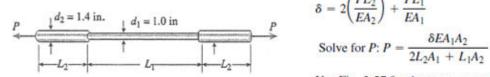
$$\sigma_{\text{nom}} = \frac{(0.12 \text{ mm}) (100 \text{ GPa})}{2(0.1 \text{ m}) \left(\frac{20}{26}\right)^2 + 0.3 \text{ m}} = 28.68 \text{ MPa}$$

$$\frac{R}{D_1} = \frac{3 \text{ mm}}{20 \text{ mm}} = 0.15$$

Use the dashed curve in Fig. 2-87. $K \approx 1.6$

$$\sigma_{\text{max}} = K\sigma_{\text{nom}} \approx (1.6) (28.68 \text{ MPa})$$

 $\approx 46 \text{ MPa} \quad \leftarrow$



$$E = 25 \times 10^{6} \, \text{psi}$$

$$\delta = 0.0040 \text{ in.}$$

$$L_1 = 20 \text{ in.}$$

$$L_2 = 5 \text{ in.}$$

$$R = \text{radius of fillets}$$
 $R = \frac{1.4 \text{ in.} - 1.0 \text{ in.}}{2}$
= 0.2 in.

$$\delta = 2\left(\frac{PL_2}{EA_2}\right) + \frac{PL_1}{EA_1}$$
Solve for $P: P = \frac{\delta EA_1A_2}{\epsilon}$

Use Fig. 2-87 for the stress-concentration factor.

$$\sigma_{\text{nom}} = \frac{P}{A_1} = \frac{\delta E A_2}{2L_2 A_1 + L_1 A_2} = \frac{\delta E}{2L_2 \left(\frac{A_1}{A_2}\right) + L_1}$$
$$= \frac{\delta E}{2L_2 \left(\frac{d_1}{d_2}\right)^2 + L_1}$$

SUBSTITUTE NUMERICAL VALUES:

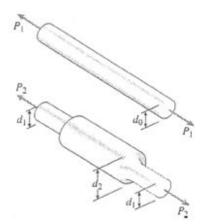
$$\sigma_{\text{nom}} = \frac{(0.0040 \text{ in.})(25 \times 10^6 \text{ psi})}{2(5 \text{ in.}) \left(\frac{1.0}{1.4}\right)^2 + 20 \text{ in.}} = 3,984 \text{ psi}$$

$$\frac{R}{D_1} = \frac{0.2 \text{ in.}}{1.0 \text{ in.}} = 0.2$$

Use the dashed curve in Fig. 2-87. $K \approx 1.53$

$$\sigma_{\text{max}} = K\sigma_{\text{nom}} \approx (1.53)(3984 \text{ psi})$$

 $\approx 6100 \text{ psi} \quad \leftarrow$



$$d_0 = 20 \text{ mm}$$

$$d_1 = 20 \text{ mm}$$

$$d_2 = 25 \text{ mm}$$

Fillet radius: R = 2 mm

Allowable stress: $\sigma_t = 80 \text{ MPa}$

(a) COMPARISON OF BARS

Prismatic bar:
$$P_1 = \sigma_t A_0 = \sigma_t \left(\frac{\pi d_0^2}{4}\right)$$

= $(80 \text{ MPa}) \left(\frac{\pi}{4}\right) (20 \text{mm})^2 = 25.1 \text{ kN} \leftarrow$

Stepped bar: See Fig. 2-87 for the stress-concentration factor.

$$R = 2.0 \text{ mm}$$
 $D_1 = 20 \text{ mm}$

$$D_1 = 20 \text{ mm}$$

$$D_2 = 25 \text{ mm}$$

$$R/D_1 = 0.10$$

$$R/D_1 = 0.10$$
 $D_2/D_1 = 1.25$ $K \approx 1.75$

$$\sigma_{\text{nom}} = \frac{P_2}{\frac{\pi}{4}d_1^2} = \frac{P_2}{A_1} \quad \sigma_{\text{nom}} = \frac{\sigma_{\text{max}}}{K}$$

$$P_2 = \sigma_{\text{nom}} A_1 = \frac{\sigma_{\text{max}}}{K} A_1 = \frac{\sigma_t}{K} A_1$$
$$= \left(\frac{80 \text{ MPa}}{1.75}\right) \left(\frac{\pi}{4}\right) (20 \text{ mm})^2$$
$$\approx 14.4 \text{ kN} \leftarrow$$

Enlarging the bar makes it weaker, not stronger. The ratio of loads is $P_1/P_2 = K = 1.75$

(b) DIAMETER OF PRISMATIC BAR FOR THE SAME ALLOWABLE LOAD

$$P_1 = P_2 \quad \sigma_t \left(\frac{\pi d_0^2}{4}\right) = \frac{\sigma_t}{K} \left(\frac{\pi d_1^2}{4}\right) \quad d_0^2 = \frac{d_1^2}{K}$$

$$d_0 = \frac{d_1}{\sqrt{K}} \approx \frac{20 \text{ mm}}{\sqrt{1.75}} \approx 15.1 \text{ mm} \quad \leftarrow$$



b = 2.4 in.

c = 1.6 in.

Fillet radius: R = 0.2 in.

Find dmax

BASED UPON FILLETS (Use Fig. 2-86)

b = 2.4 in. c = 1.6 in. R = 0.2 in.

R/c = 0.125 b/c = 1.5 $K \approx 2.10$

$$P_{\text{max}} = \sigma_{\text{nom}} ct = \frac{\sigma_{\text{max}}}{K} ct = \frac{\sigma_{\text{max}}}{K} \left(\frac{c}{b}\right) (bt)$$

 $\approx 0.317 \, bt \, \sigma_{\text{max}}$

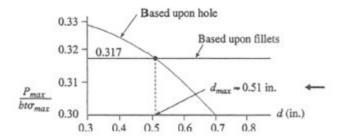
BASED UPON HOLE (Use Fig. 2-84)

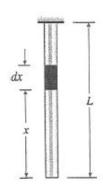
b = 2.4 in. d = diameter of the hole (in.)

 $c_1 = b - d$

$$P_{\text{max}} = \sigma_{\text{nom}} c_1 t = \frac{\sigma_{\text{max}}}{K} (b - d) t$$
$$= \frac{1}{K} \left(1 - \frac{d}{b} \right) b t \sigma_{\text{max}}$$

d(in.)	d/b	K	$P_{\rm max}/bt\sigma_{\rm max}$
0.3	0.125	2.66	0.329
0.4	0.167	2.57	0.324
0.5	0.208	2.49	0.318
0.6	0.250	2.41	0.311
0.7	0.292	2.37	0.299





Let A = cross-sectional area Let N = axial force at distance x $N = \gamma A x$ $C = \frac{N}{A} = \gamma x$

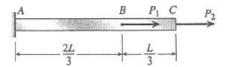
STRAIN AT DISTANCE X

$$\varepsilon = \frac{\sigma}{E} + \frac{\sigma_0 \alpha}{E} \left(\frac{\sigma}{\sigma_0}\right)^m = \frac{\gamma x}{E} + \frac{\sigma_0}{\alpha E} \left(\frac{\gamma x}{\sigma_0}\right)^m$$

ELONGATION OF BAR

$$\delta = \int_0^L \varepsilon dx = \int_0^L \frac{\gamma x}{E} dx + \frac{\sigma_0 \alpha}{E} \int_0^L \left(\frac{\gamma x}{\sigma_0}\right)^m dx$$

$$= \frac{\gamma L^2}{2E} + \frac{\sigma_0 \alpha L}{(m+1)E} \left(\frac{\gamma L}{\sigma_0}\right)^m \qquad \text{Q.E.D.} \quad \leftarrow$$



$$L = 1.8 \text{ m}$$
 $A = 480 \text{ mm}^2$
 $P_1 = 30 \text{ kN}$ $P_2 = 60 \text{ kN}$

Ramberg-Osgood equation:

$$\varepsilon = \frac{\sigma}{45,000} + \frac{1}{618} \left(\frac{\sigma}{170}\right)^{10} (\sigma = \text{MPa})$$

Find displacement at end of bar.

(a) P_1 ACTS ALONE

AB:
$$\sigma = \frac{P_1}{A} = \frac{30 \text{ kN}}{480 \text{ mm}^2} = 62.5 \text{ MPa}$$

$$\varepsilon = 0.001389$$

$$\delta_c = \varepsilon \left(\frac{2L}{3}\right) = 1.67 \text{ mm} \quad \leftarrow$$

(b) P_2 ACTS ALONE

$$ABC: \sigma = \frac{P_2}{A} = \frac{60 \text{ kN}}{480 \text{ mm}^2} = 125 \text{ MPa}$$

 $\varepsilon = 0.002853$
 $\delta_c = \varepsilon L = 5.13 \text{ mm} \leftarrow$

(c) Both P_1 and P_2 are acting $AB:\sigma = \frac{P_1 + P_2}{A} = \frac{90 \text{ kN}}{480 \text{ mm}^2} = 187.5 \text{ MPa}$ $\varepsilon = 0.008477$ $\delta_{AB} = \varepsilon \left(\frac{2L}{3}\right) = 10.17 \text{ mm}$

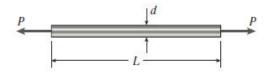
$$BC: \sigma = \frac{P_2}{A} = \frac{60 \text{ kN}}{480 \text{ mm}^2} = 125 \text{ MPa}$$

 $\varepsilon = 0.002853$

$$\delta_{BC} = \varepsilon \left(\frac{L}{3}\right) = 1.71 \text{ mm}$$

$$\delta_C = \delta_{AB} + \delta_{BC} = 11.88 \text{ mm} \quad \leftarrow$$

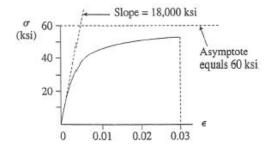
(Note that the displacement when both loads act simultaneously is *not* equal to the sum of the displacements when the loads act separately.)



$$L = 32 \text{ in.}$$
 $d = 0.75 \text{ in.}$
 $A = \frac{\pi d^2}{4} = 0.4418 \text{ in.}^2$

(a) STRESS-STRAIN DIAGRAM

$$\sigma = \frac{18,000\varepsilon}{1 + 300\varepsilon}$$
 $0 \le \varepsilon \le 0.03$ $(\sigma = ksi)$



(b) ALLOWABLE LOAD P

Maximum elongation $\delta_{max} = 0.25$ in.

Maximum stress $\sigma_{\text{max}} = 40 \text{ ksi}$

Based upon elongation:

$$\varepsilon_{\text{max}} = \frac{\delta_{\text{max}}}{L} = \frac{0.25 \text{ in.}}{32 \text{ in.}} = 0.007813$$

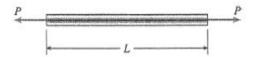
$$\sigma_{\text{max}} = \frac{18,000\varepsilon_{\text{max}}}{1 + 300\varepsilon_{\text{max}}} = 42.06 \text{ ksi}$$

BASED UPON STRESS:

$$\sigma_{\text{max}} = 40 \text{ ksi}$$

Stress governs.
$$P = \sigma_{\text{max}} A = (40 \text{ ksi})(0.4418 \text{ in.}^2)$$

= 17.7 k \leftarrow



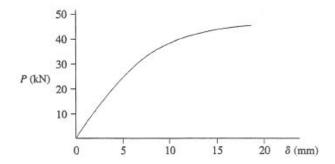
L = 2.0 m

 $A = 249 \text{ mm}^2$

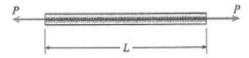
STRESS-STRAIN DIAGRAM (See the problem statement for the diagram)

LOAD-DISPLACEMENT DIAGRAM

P (kN)	$\sigma = P/A$ (MPa)	ε (from diagram)	$\delta = \varepsilon L$ (mm)
10	40	0.0009	1.8
20	80	0.0018	3.6
30	120	0.0031	6.2
40	161	0.0060	12.0
45	181	0.0081	16.2



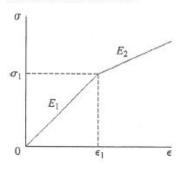
NOTE: The load-displacement curve has the same shape as the stress-strain curve.



$$L = 150 \text{ in.}$$

$$A = 2.0 \text{ in.}^2$$

STRESS-STRAIN DIAGRAM



$$E_1 = 10 \times 10^6 \, \text{psi}$$

$$E_2 = 2.4 \times 10^6 \text{ psi}$$

$$\sigma_1 = 12,000 \text{ psi}$$

$$\varepsilon_1 = \frac{\sigma_1}{E_1} = \frac{12,000 \text{ psi}}{10 \times 10^6 \text{ psi}}$$

$$= 0.0012$$

For $0 \le \sigma \le \sigma_1$:

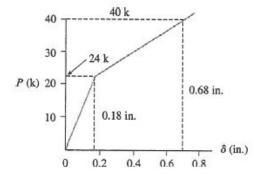
$$\varepsilon = \frac{\sigma}{E_1} = \frac{\sigma}{10 \times 10^6 \text{psi}} (\sigma = \text{psi})$$
 Eq. (1)

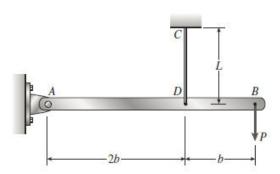
For $\sigma \ge \sigma_1$:

$$\begin{split} \varepsilon &= \varepsilon_1 + \frac{\sigma - \sigma_1}{E_2} = 0.0012 + \frac{\sigma - 12,000}{2.4 \times 10^6} \\ &= \frac{\sigma}{2.4 \times 10^6} - 0.0038 \quad (\sigma = \text{psi}) \quad \text{Eq. (2)} \end{split}$$

LOAD-DISPLACEMENT DIAGRAM

P (k)	$\sigma = P/A$ (psi)	ε (from Eq. 1 or Eq. 2)	$\delta = \varepsilon L$ (in.)
8	4,000	0.00040	0.060
16	8,000	0.00080	0.120
24	12,000	0.00120	0.180
32	16,000	0.00287	0.430
40	20,000	0.00453	0.680





Wire:
$$E = 210 \text{ GPa}$$

$$\sigma_Y = 820 \text{ MPa}$$

$$L = 1.0 \text{ m}$$

$$d = 3 \text{ mm}$$

$$A = \frac{\pi d^2}{4} = 7.0686 \text{ mm}^2$$

STRESS-STRAIN DIAGRAM

$$\sigma = E\varepsilon$$
 $(0 \le \sigma \le \sigma_Y)$ (1)

$$\sigma = \sigma_Y \left(\frac{E\varepsilon}{\sigma_Y}\right)^n \qquad (\sigma \ge \sigma_Y) \qquad (n = 0.2)$$
 (2)

(a) DISPLACEMENT δ_B AT END OF BAR

$$\delta = \text{elongation of wire } \delta_B = \frac{3}{2}\delta = \frac{3}{2}\varepsilon L$$
 (3)

Obtain & from stress-strain equations:

From Eq. (1):
$$\varepsilon = \frac{\sigma E}{(0 \le \sigma \le \sigma_{\gamma})}$$
 (4)

From Eq. (2):
$$\varepsilon = \frac{\sigma_{\gamma}}{E} \left(\frac{\sigma}{\sigma_{\gamma}}\right)^{1/n}$$
 (5)

Axial force in wire: $F = \frac{3P}{2}$

Stress in wire:
$$\sigma = \frac{F}{A} = \frac{3P}{2A}$$
 (6)

PROCEDURE: Assume a value of P

Calculate σ from Eq. (6) Calculate ε from Eq. (4) or (5) Calculate δ_B from Eq. (3)

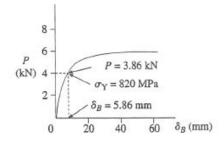
P (kN)	σ (MPa) Eq. (6)	ε Eq. (4) or (5)	δ_B (mm) Eq. (3)
2.4	509.3	0.002425	3.64
3.2	679.1	0.003234	4.85
4.0	848.8	0.004640	6.96
4.8	1018.6	0.01155	17.3
5.6	1188.4	0.02497	37.5

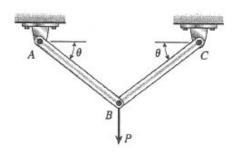
For $\sigma = \sigma_Y = 820$ MPa:

$$\varepsilon = 0.0039048$$
 $P = 3.864 \text{ kN}$

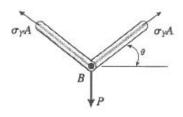
$$\delta_B = 5.86 \text{ mm}$$

(b) Load-displacement diagram





Structure is statically determinate. The yield load P_Y and the plastic lead P_P occur at the same time, namely, when both bars reach the yield stress.

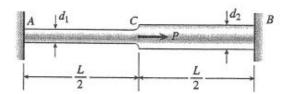


JOINT B

$$\Sigma F_{\text{vert}} = 0$$

$$(2\sigma_Y A)\sin\theta = P$$

$$P_Y = P_P = 2\sigma_Y A \sin \theta \leftarrow$$



$$d_1 = 20 \text{ mm}$$

$$d_1 = 20 \text{ mm}$$
 $d_2 = 25 \text{ mm}$

$$\sigma_Y = 250 \text{ MPa}$$

DETERMINE THE PLASTIC LOAD P_P :

At the plastic load, all parts of the bar are stressed to the yield stress.

Point C:

$$F_{AC}$$
 F_{CB}

$$F_{AC} = \sigma_Y A_1$$
 $F_{CB} = \sigma_Y A_2$
$$P = F_{AC} + F_{CB}$$

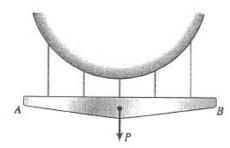
$$P_P = \sigma_Y A_1 + \sigma_Y A_2 = \sigma_Y (A_1 + A_2) \leftarrow$$

SUBSTITUTE NUMERICAL VALUES:

$$P_P = (250 \text{ MPa}) \left(\frac{\pi}{4}\right) (d_1^2 + d_2^2)$$

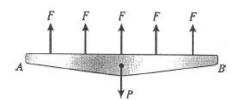
$$= (250 \text{ MPa}) \left(\frac{\pi}{4}\right) [(20 \text{ mm})^2 + (25 \text{ mm})^2]$$

$$= 201 \text{ kN} \leftarrow$$



(a) PLASTIC LOAD P_P At the plastic load, each wire is stressed to the yield stress. $\therefore P_P = 5\sigma_Y A$ \leftarrow

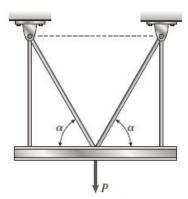
$$F = \sigma_{y}A$$



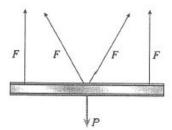
(b) BAR AB IS FLEXIBLE

At the plastic load, each wire is stressed to the yield stress, so the plastic load is not changed. ←

(c) Radius R is increased Again, the forces in the wires are not changed, so the plastic load is not changed. ←



At the plastic load, all four rods are stressed to the yield stress.

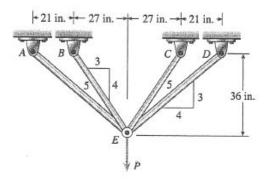


 $F = \sigma_{Y}A$

Sum forces in the vertical direction and solve for the load:

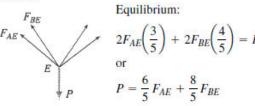
$$P_P = 2F + 2F \sin \alpha$$

$$P_P = 2\sigma_Y A (1 + \sin \alpha) \leftarrow$$



$$L_{AE} = 60 \text{ in.}$$
 $L_{BE} = 45 \text{ in.}$

JOINT E



PLASTIC LOAD P_P

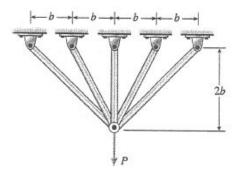
At the plastic load, all bars are stressed to the yield stress.

$$F_{AE} = \sigma_Y A_{AE} \qquad F_{BE} = \sigma_Y A_{BE}$$

$$P_P = \frac{6}{5} \sigma_Y A_{AE} + \frac{8}{5} \sigma_Y A_{BE} \quad \longleftarrow$$

SUBSTITUTE NUMERICAL VALUES:

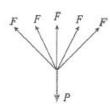
$$A_{AE} = 0.307 \text{ in.}^2$$
 $A_{BE} = 0.601 \text{ in.}^2$
 $\sigma_Y = 36 \text{ ksi}$
 $P_P = \frac{6}{5}(36 \text{ ksi}) (0.307 \text{ in.}^2) + \frac{8}{5}(36 \text{ ksi}) (0.601 \text{ in.}^2)$
 $= 13.26 \text{ k} + 34.62 \text{ k} = 47.9 \text{ k} \leftarrow$



$$d = 10 \text{ mm}$$

$$A = \frac{\pi d^2}{4} = 78.54 \text{ mm}^2$$

$$\sigma_Y = 250 \text{ MPa}$$



At the plastic load, all five bars are stressed to the yield stress

$$F = \sigma_{V} A$$

Sum forces in the vertical direction and solve for the load:

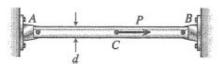
$$P_P = 2F\left(\frac{1}{\sqrt{2}}\right) + 2F\left(\frac{2}{\sqrt{5}}\right) + F$$
$$= \frac{\sigma_{\gamma}A}{5}(5\sqrt{2} + 4\sqrt{5} + 5)$$

$$= 4.2031\sigma_y A \leftarrow$$

Substitute numerical values:

$$P_P = (4.2031)(250 \text{ MPa})(78.54 \text{ mm}^2)$$

= 82.5 kN \leftarrow



$$d = 0.6 \text{ in.}$$

$$\sigma_Y = 36 \text{ ksi}$$

Initial tensile stress = 10 ksi

(a) PLASTIC LOAD P_P

The presence of the initial tensile stress does not affect the plastic load. Both parts of the bar must yield in order to reach the plastic load.

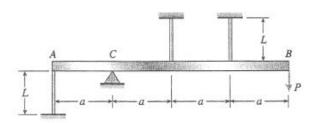
Point C:

$$\stackrel{\sigma_{Y}A}{\longleftarrow} C \stackrel{P}{\longrightarrow} \stackrel{\sigma_{Y}A}{\longleftarrow}$$

$$P_{P} = 2\sigma_{Y}A = (2) (36 \text{ ksi}) \left(\frac{\pi}{4}\right) (0.60 \text{ in.})^{2}$$

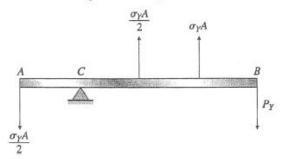
$$= 20.4 \text{ k} \qquad \leftarrow$$

(B) Initial tensile stress is doubled P_P is not changed. \leftarrow



(a) YIELD LOAD P_Y

Yielding occurs when the most highly stressed wire reaches the yield stress σ_Y



$$\Sigma M_C = 0$$

$$P_Y = \sigma_Y A \leftarrow$$

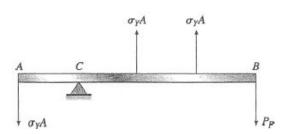
At point A:

$$\delta_A = \left(\frac{\sigma_{\gamma} A}{2}\right) \left(\frac{L}{EA}\right) = \frac{\sigma_{\gamma} L}{2E}$$

At point B:

$$\delta_B = 3\delta_A = \delta_Y = \frac{3\sigma_Y L}{2E} \leftarrow$$

(b) Plastic load P_P



At the plastic load, all wires reach the yield stress.

$$\sum M_C = 0$$

$$P_p = \frac{4\sigma_{\gamma}A}{3} \leftarrow$$

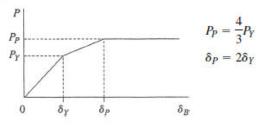
At point A:

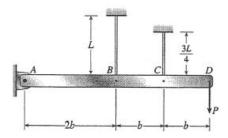
$$\delta_A = (\sigma_Y A) \left(\frac{L}{EA}\right) = \frac{\sigma_Y L}{E}$$

At point B:

$$\delta_B = 3\delta_A = \delta_P = \frac{3\sigma_\gamma L}{E} \leftarrow$$

(c) LOAD-DISPLACEMENT DIAGRAM



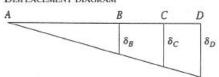


A = cross-sectional area

 σ_Y = yield stress

E = modulus of elasticity

DISPLACEMENT DIAGRAM

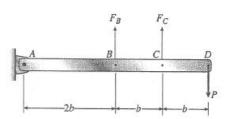


COMPATIBILITY:

$$\delta_C = \frac{3}{2} \delta_B \qquad (1)$$

$$\delta_D = 2\delta_B$$
 (2)

FREE-BODY DIAGRAM



EQUILIBRIUM:

FORCE-DISPLACEMENT RELATIONS

$$\delta_B = \frac{F_B L}{EA} \quad \delta_C = \frac{F_C \left(\frac{3}{4}L\right)}{EA} \tag{4,5}$$

Substitute into Eq. (1):

$$\frac{3F_CL}{4EA} = \frac{3F_BL}{2EA}$$

$$F_C = 2F_B \tag{6}$$

STRESSES

$$\sigma_B = \frac{F_B}{A} \quad \sigma_C = \frac{F_C}{A} \quad \sigma_C = 2\sigma_B \tag{7}$$

Wire C has the larger stress. Therefore, it will yield first.

(a) YIELD LOAD

$$\sigma_C = \sigma_Y$$
 $\sigma_B = \frac{\sigma_C}{2} = \frac{\sigma_Y}{2}$ (From Eq. 7)

$$F_C = \sigma_{Y}A$$
 $F_B = \frac{1}{2}\sigma_{Y}A$

From Eq. (3):

$$2\left(\frac{1}{2}\sigma_{Y}A\right) + 3(\sigma_{Y}A) = 4P$$

$$P = P_Y = \sigma_Y A \leftarrow$$

From Eq. (4):

$$\delta_B = \frac{F_B L}{EA} = \frac{\sigma_Y L}{2E}$$

From Eq. (2):

$$\delta_D = \delta_Y = 2\delta_B = \frac{\sigma_Y L}{E} \leftarrow$$

(b) PLASTIC LOAD

At the plastic load, both wires yield.

$$\sigma_B = \sigma_Y = \sigma_C$$
 $F_B = F_C = \sigma_Y A$

From Eq. (3):

$$2(\sigma_{V}A) + 3(\sigma_{V}A) = 4P$$

$$P = P_p = \frac{5}{4}\sigma_{\gamma}A \leftarrow$$

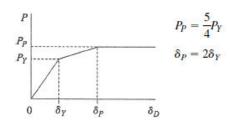
From Eq. (4):

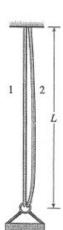
$$\delta_B = \frac{F_B L}{EA} = \frac{\sigma_Y L}{E}$$

From Eq. (2):

$$\delta_D = \delta_P = 2\delta_B = \frac{2\sigma_\gamma L}{E} \quad \longleftarrow \quad$$

(c) LOAD-DISPLACEMENT DIAGRAM





$$L = 40 \text{ m}$$
 $A = 48.0 \text{ mm}^2$
 $E = 160 \text{ GPa}$

d = difference in length = 100 mm

$$\sigma_Y = 500 \text{ MPa}$$

INITIAL STRETCHING OF CABLE 1

Initially, cable 1 supports all of the load. Let $W_1 = \text{load}$ required to stretch cable 1 to the same length as cable 2

$$W_1 = \frac{EA}{L}d = 19.2 \text{ kN}$$

 $\delta_1 = 100 \text{ mm} \text{ (elongation of cable 1)}$

$$\sigma_1 = \frac{W_1}{A} = \frac{Ed}{L} = 400 \text{ MPa} (\sigma_1 < \sigma_Y : ... > \text{OK})$$

(a) YIELD LOAD W_Y

Cable 1 yields first. $F_1 = \sigma_y A = 24 \text{ kN}$

 δ_{1Y} = total elongation of cable 1

 δ_{1y} = total elongation of cable 1

$$\delta_{1Y} = \frac{F_1 L}{EA} = \frac{\sigma_Y L}{E} = 0.125 \text{ m} = 125 \text{ mm}$$

$$\delta_Y = \delta_{1Y} = 125 \text{ mm} \leftarrow$$

$$\delta_{2Y}$$
 = elongation of cable 2

$$= \delta_{1y} - d = 25 \,\mathrm{mm}$$

$$F_2 = \frac{EA}{I} \delta_{2Y} = 4.8 \text{ kN}$$

$$W_Y = F_1 + F_2 = 24 \text{ kN} + 4.8 \text{ kN}$$

(b) PLASTIC LOAD W_P

$$F_1 = \sigma_Y A$$
 $F_2 = \sigma_Y A$

$$W_P = 2\sigma_V A = 48 \text{ kN} \leftarrow$$

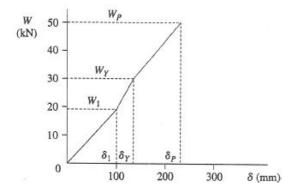
$$\delta_{2P}$$
 = elongation of cable 2

$$= F_2 \left(\frac{L}{EA}\right) = \frac{\sigma \gamma L}{E} = 0.125 \text{ mm} = 125 \text{ mm}$$

$$\delta_{1P} = \delta_{2P} + d = 225 \text{ mm}$$

$$\delta_P = \delta_{1P} = 225 \text{ mm} \quad \leftarrow$$

(c) Load-displacement diagram



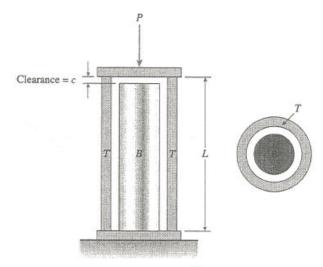
$$\frac{W_Y}{W_1} = 1.5 \quad \frac{\delta_Y}{\delta_1} = 1.25$$

$$\frac{W_P}{W_Y} = 1.667 \quad \frac{\delta_P}{\delta_Y} = 1.8$$

$$0 < W < W_1$$
: slope = 192,000 N/m

$$W_1 < W < W_Y$$
: slope = 384,000 N/m

$$W_Y < W < W_P$$
: slope = 192,000 N/m



$$L = 15 \text{ in.}$$

 $c = 0.010 \text{ in.}$
 $E = 29 \times 10^3 \text{ ksi}$

$$\sigma_Y = 36 \text{ ksi}$$

TUBE:

$$d_2 = 3.0 \text{ in.}$$

$$d_1 = 2.75$$
 in.

$$A_T = \frac{\pi}{4} (d_2^2 - d_1^2) = 1.1290 \text{ in.}^2$$

BAR

$$d = 1.5 \text{ in.}$$

$$A_B = \frac{\pi d^2}{4} = 1.7671 \text{ in.}^2$$

INITIAL SHORTENING OF TUBE T

Initially, the tube supports all of the load.

Let P_1 = load required to close the clearance

$$P_1 = \frac{EA_T}{L}c = 21,827 \text{ lb}$$

Let δ_1 = shortening of tube $\delta_1 = c = 0.010$ in.

$$\sigma_1 = \frac{P_1}{A_T} = 19,330 \text{ psi}$$
 $(\sigma_1 < \sigma_Y :: OK)$

(a) YIELD LOAD P_Y

Because the tube and bar are made of the same material, and because the strain in the tube is larger than the strain in the bar, the tube will yield first.

$$F_T = \sigma_Y A_T = 40,644 \text{ lb}$$

 δ_{TY} = shortening of tube at the yield stress

$$\sigma_{TY} = \frac{F_T L}{EA_T} = \frac{\sigma_Y L}{E} = 0.018621 \text{ in.}$$

$$\delta_V = \delta_{TY} = 0.018621$$
 in. \leftarrow

(b) PLASTIC LOAD P_P

$$F_T = \sigma_Y A_T$$
 $F_B = \sigma_Y A_B$

$$P_P = F_T + F_B = \sigma_Y (A_T + A_B)$$

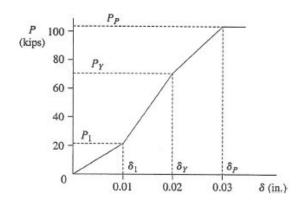
 δ_{BP} = shortening of bar

$$= F_B \left(\frac{L}{EA_B} \right) = \frac{\sigma_Y L}{E} = 0.018621 \text{ in.}$$

$$\delta_{TP} = \delta_{BP} + c = 0.028621$$
 in.

$$\delta_P = \delta_{TP} = 0.02862 \text{ in.} \leftarrow$$

(c) Load-displacement diagram



Part (c) continued

$$\delta_{BY} = \text{shortening of bar}$$
 $= \delta_{TY} - c = 0.008621 \text{ in.}$
 $F_B = \frac{EA_B}{L} \delta_{BY} = 29,453 \text{ lb}$
 $P_Y = F_T + F_B = 40,644 \text{ lb} + 29,453 \text{ lb}$
 $= 70,097 \text{ lb}$
 $P_Y = 70,100 \text{ lb} \leftarrow$

$$\frac{P_Y}{P_1} = 3.21$$
 $\frac{\delta_Y}{\delta_1} = 1.86$
 $\frac{P_P}{P_Y} = 1.49$ $\frac{\delta_P}{\delta_Y} = 1.54$
 $0 < P < P_1$: slope = 2180 k/in.
 $P_1 < P < P_Y$: slope = 5600 k/in.

 $P_Y < P < P_P$: slope = 3420 k/in.

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